GRAPHICAL PAGE DEVELOPMENT OF AN ELECTRONIC HORIZONTAL SITUATION INDICATOR

bу

CHARLES A. ROBERTSON

B.S., Kansas State University, 1985

A MASTER'S THESIS

submitted in partial fulfillment of the requirements for the degree

MASTER OF SCIENCE

Department of Electrical and Computer Engineering

KANSAS STATE UNIVERSITY
Manhattan, Kansas
1987

Approved by:

ajor Professor

LD 2618 .TY EECE 1987 R62 c.2

TABLE OF CONTENTS

CHAPTER O	NE INTRODUCTION All207 30453	17
1.1 1.2 1.3	What is an EHSI? Purpose of the EHSI Goals for the EHSI Development System. EHSI System Components Present Status of the EHSI Development System Previous Accomplishments EHSI Host-side Development: Gruenbacher's Accomplishments	1 2 3 7 13
	EHSI Host-side Development:	19
	EHSI Host-side Development:	22
1.5	EHSI Host-side Programming Model	23
CHAPTER T	WO MAIN ROUTINE	
2. 1 2. 2 2. 3	Functions Data Structures High-level Flowchart	26 29 33
CHAPTER T	HREE TOOLS FOR VECTOR GRAPHICS DEVELOPMENT OF THE EHSI	
3.1 3.2	General-Purpose Tools. Page-Specific Tools. DATA PAGE Tools. NAV PAGE Tools. ILS PAGE Tools.	36 42 42 44 47
CHAPTER F	OUR HP 1345A VECTOR GRAPHICS	49
CHAPTER F	IVE VECTOR MEMORY LAYOUT FOR THE EHSI	61
CHAPTER S	IX DATA PAGE	
6.1 6.2 6.3	Static Portion Dynamic Portion High-level Flowchart	69

TABLE OF CONTENTS CONTINUED

CHAPTER SI	EVEN NAV PAGE	
7.1 7.2 7.3 7.4	Static Portion Dynamic Portion Future Development High-level Flowchart	76 78 81 84
CHAPTER E	IGHT ILS PAGE	
8.1 8.2 8.3 8.4	Static Portion Dynamic Portion Future Development. High-level Flowchart.	89 90 92 97
CHAPTER N	INE CONCLUSION	
9.1 9.2 9.3	•	99 100 101
REFERENCES	5	103
APPENDIX A	CHANGES TO THE DACI	104
APPENDIX I	B EHSI USER'S MANUAL	116
APPENDIX (EHSI HOST PROGRAM MAINTENANCE	123
APPENDIX I	Main and Page Routines General-Purpose Routines Page-Specific Tools	139 139 184 223 276
GLOSSARY		290

LIST OF FIGURES

Figure		Page
1.1	Data Page	4
1.2	Navigational Page	5
1.3	ILS Page	6
1.4	Development System Components	8
1.5	Control Keyboard	11
1.6	EHSI Host Program Model	25
2.1	Main Routine Basic Functions	30
4.1	Set Condition Command	55
4.2	Plot Command	56
4.3	1345A Modified ASCII Character Set	57
4.4	Text Command	58
4.5	Boundary Limits for Character Plotting	59
4.6	Vector Memory Commands	60
5.1	Vector Memory Layout for the EHSI	64
6.1	Present Flight Data Page	72
7.1	Present Navigational Page	77
8.1	Present ILS Page	88
8.2	ILS Page Using Two Crosshairs	93
8.3	ILS Page with Approach Tunnel	94
A. 1	DACI to Host Parallel Port Addition of IRQIN and IRQOUT	107
A. 2	Control PIA and Alarm Horn Addition of IRQIN and IRQOUT	108
A.3	OUTSHAKE_IRQ Routine	109

LIST OF FIGURES CONTINUED

Figure		Page
A. 4	EEPROM's and Real-time Clock Addition of External Clock Switch	110
C. 1	Brief Quick Reference Commands	111
C. 2	Brief Quick Reference Commands	112

CHAPTER ONE

INTRODUCTION

1.1 What is an EHSI?

An electronic horizontal situation indicator (EHSI) is a digital avionics system which utilizes a cathode-ray tube (CRT) display. The EHSI provides a pilot with a pictorial representation of a plane's position relative to known navigational fixes. This representation contains distance and directional information to the fixes along with other pertinent data.

EHSI's have been developed for the larger commercial jets such as the Boeing 757 and the Airbus A310. Many of the military jets also contain EHSI's. However, no affordable EHSI system has been developed for general aviation aircraft in the lower and middle price ranges. This is surprising because most of these types of aircraft are flown by single pilots under instrument conditions. An EHSI would be ideal for such a pilot flying under instrument

flight rule (IFR) conditions.

The main goal of our EHSI is to provide an affordable EHSI for the general aviation public.

1.2 Purpose of the EHSI

The purpose of the Electronic Horizontal Situation Indicator (EHSI) is to ease the burden of assimilating all the data that a pilot has to make use of during the course of a flight. It will provide the pilot with three different pages of information which will be displayed on a vector graphics display and will have a control keyboard so that the pilot can enter flight parameters and control the various functions of the EHSI.

The functions which will be available on the EHSI will consist of the following:

- FLIGHT DATA PAGE: This page will contain such information as the plane's heading, airspeed, present altitude, assigned altitude, minimum descent altitude (MDA) or decision height (DH), navigational frequencies, communication frequencies, automatic direction finder(ADF) frequency, way points, current time, temperature, etc.
- 2) NAVIGATIONAL PAGE: This page will provide the pilot with information about the plane's heading in a compass type format, position information with regard to known navigational fixes such as VORTAC's and non-directional beacons (NDB's), present altitude, airspeed, etc.
- 3) INSTRUMENT LANDING SYSTEM (ILS) PAGE: This page will provide information about a plane's position on an ILS approach. It will contain the plane's position relative to the glideslope and localizer

and display it in some useful format to allow the pilot to do an instrument landing. Heading, altitude, MDA/DH, and marker status will also be displayed.

- 4) calculator functions: Basic calculator functions will be available to the pilot for making various calculations during the course of a flight. These functions will be implemented in an HP-style format.
- 5) miscellaneous functions: Other functions which will be incorporated will be the ability to enter various flight parameters such as the navigational, communication, and adf frequencies, assigned altitude, MDA/DH, and estimated wind. The EHSI will also contain a real-time clock which can be set with the control keyboard and a timer which will allow the pilot to know the time that he has been on an ILS approach.

All the pages discussed above will be dealt with in detail in subsequent chapters. The pages are also discussed in [1] and [2]. The majority of the information for the page layouts was derived from these papers. Examples, of these pages can be seen in Fig. 1.1 through 1.3.

1.3 Development System Basics

1.3.1 Goals for the EHSI Development System

The EHSI Development System has six basic goals which it needs to accomplish. These goals are the following:

- provide access to pertinent flight data from flight simulator
- retrieve control keyboard inputs, execute operation or send input to host

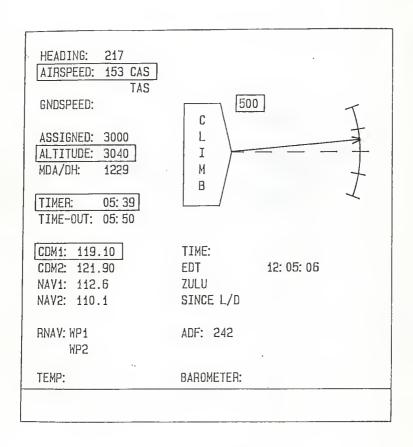


Figure 1.1 DATA PAGE

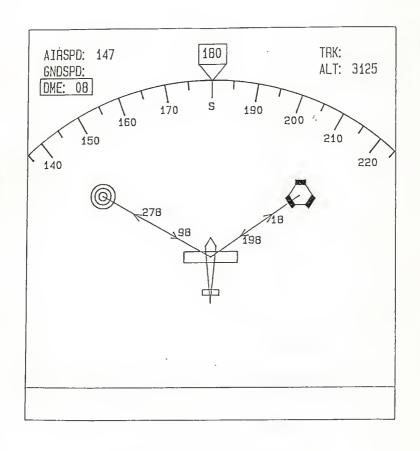


Figure 1.2 NAVIGATIONAL PAGE

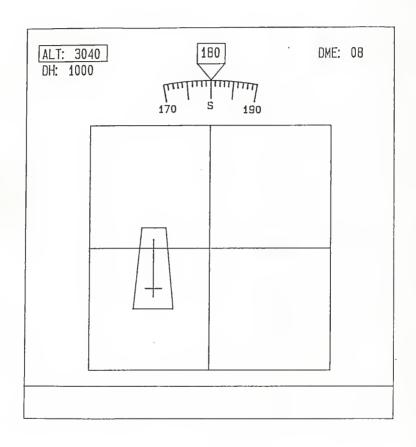


Figure 1.3 INSTRUMENT LANDING SYSTEM (ILS) PAGE

- 3) provide alarm capability
- 4) provide screen data transfer capability
- 5) have a simple communications protocol between interface and host
- 6) graphics development and data manipulation are to be done on host computer

1.3.2 EHSI System Components

The EHSI Development System consists of five basic components which can be seen in Fig. 1.4. The five components are the following:

- 1) Data Acquisition and Communications Interface (DACI)
- 2) Control keyboard with system on/off switch
- 3) HP 1345A Vector Graphics Display (VGD)
- 4) ATC-610 flight simulator
- 5) Host computer : Zenith 158-A PC

The DACI is a Motorola MC68000-based system which makes use of a Motorola MC68000 Educational Board as the controller. The DACI provides the means by which all the other system components talk with each other. This interface was necessary because those components are stand-alone devices and have no otherwise effective means of communicating with one another. The DACI acquires all the analog and binary signals from the flight simulator. The DACI also conditions all of these signals, converts them to digital values, and

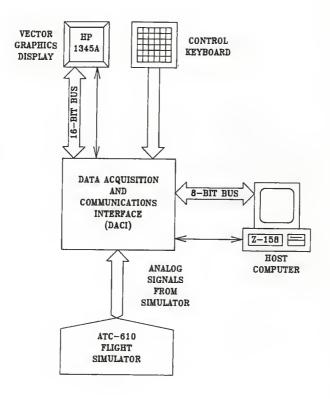


Figure 1.4 DEVELOPMENT SYSTEM COMPONENTS

stores them away so they are available to be sent to the host computer when requested. The host computer will then process these digital values. The DACI has a real-time clock and calendar on it. The real-time clock value is also sent with the converted flight simulator signals when requested by the host. The DACI processes all key presses, including the state of the system on/off switch, and sends them to the host for evaluation. The DACI sends display commands to the HP 1345A VGD and allows the host to transfer display commands to or from the VGD. The DACI contains a piezo-electric alarm which can be toggled by the host. The DACI also performs all the necessary initializations for the display. This initial system was designed by Lagerberg and more details about the DACI can be found in [2].

The commands accepted by the DACI are the following:

- 1) send data package
- 2) send screen data to the VGD
- 3) receive screen data from the VGD
- 4) toggle alarm

All command requests are sent by the host to the DACI via a parallel port. The commands allow the host to get the flight data from the simulator, process it, generate displays, and send them to the VGD. All processing for the displays is done by the host; the DACI just acts as an intermediary between the host and the VGD. The alarm is

used to alert the pilot of impending conditions which may be threatening.

The control keyboard, shown in Fig. 1.5, is a six-bysix matrix keyboard with system on/off switch. This keyboard is used to determine system status (on or off) and to allow entry of various commands or parameters. These parameters include such things as navigational frequencies, communication frequencies, ADF frequency, timer, assigned altitude, MDA/DH, and estimated wind. These are entered by the pilot either during preflight or during the course of a flight. The commands which can be entered by the keyboard are "start timer," "reset timer," basic Hewlett-Packard(HP)style calculator functions, "clear," "enter." "toggle alarm, " "set clock, " etc. The three different page keys are used to switch among the various pages displayed by the EHSI. It should be noted that when the system switch is turned off, the system enters a wait state. When the system switch is turned back on, the whole system is reinitialized and is set up in its original state, ready for commands from the host and for retrieval of data from the flight simulator.

The HP 1345A is a high-resolution vector graphics display. It is used to display all the graphics generated by the interface and host. It has a 2048-by-2048 addressable area for plotting. It is especially suited for this type of application since it is capable of operating up to 15,000

EHSI SYSTEM SWITCH



START	BRG/HLD INBND		TIMER	RESET	SET
ADF	FLIGHT DATA PAGE	MDA/DH	ASGN ALT	EST	SET/RST ALARM
VOR1	RNAV		7	8	9
VOR2	ILS PAGE	+	4	5	6
COM1	CLEAR	X	1	2	3
сом2	ENTER	•	0	•	

Figure 1.5 CONTROL KEYBOARD

feet pressure altitude. It accepts four basic commands and has a built in character set. The display and its various commands and attributes will be discussed in more detail in Chapter 4.

The ATC-610 flight simulator is used to obtain all the pertinent flight data. It is an FAA approved instrument flight simulator. All of the analog signals from the simulator are conditioned by the DACI and converted into digital values which can by requested by the host.

The host computer that was chosen for this system was the Zenith 158-A PC. It is responsible for manipulating the data retrieved by the DACI and generating all the graphics for the different pages of the EHSI. The C programming language was chosen; it has many high-level language features, can still do bit manipulations very easily, and is a fast-running language. The Microsoft C Compiler, Version 4.0, was used.

The Z-158 used has a standard RS-232C serial asynchronous communications port, a Centronix standard parallel printer port, a CPU clock speed of either 4.77 Mhz or 8 Mhz, an 8087 numeric co-processor, and additional backplane expansion slots. The computer is configured with 640k bytes of RAM. It has one floppy disk drive and a 20-megabyte Winchester hard disk system. It should be noted that all the routines are run with the Z-158 set at the 8 MHz clock rate

to keep processing time to a minimum.

The Centronix standard parallel port is used as the general purpose I/O port. It should be noted that hardware modifications were made to this port and are discussed by Lagerberg in [2]. Since the standard parallel port drivers have a limited transfer rate of 1000 characters per second, parallel port drivers were written in 8088 assembly language to communicate at 10,000 characters per second with the DACI. This transfer rate was need for the development system. These routines were written by Gruenbacher and are discussed in detail in [3]. These drivers are accessed by the C routines used to generate the displays.

The Z-158 operates in an interrupt environment and makes use of the parallel port autovectored interrupt number seven. This interrupt is actually installed as a background process and is discussed in detail in [3]. The background interrupt process retrieves all interrupts sent by the DACI and stores them in a 20-byte stack. The stack is needed so that no interrupts will be missed.

1.4 Present Status of the EHSI Development System

1.4.1 Previous Accomplishments

Lagerberg designed the initial development system which satisfied the six basic goals stated previously. The initial system accomplishes the following:

- provides access to pertinent flight data from flight simulator
- retrieves control keyboard inputs, execute operation or send input to host
- 3) has alarm capability
- 4) provides screen data transfer capability
- 5) has a simple communications protocol between interface and host
- 6) allows graphics development and data manipulation to be done on host computer

This system involves a combination of hardware and software to accomplish its duties. Details about this portion of the development system can be found in [2].

1.4.2 EHSI Host-side Development: Gruenbacher's Accomplishments

Gruenbacher developed five separate areas of the hostside of the development system:

- background interrupt process to accept interrupts from the DACI
- interface communications routines in 8088 assembly language
- interface of the assembly routines with the C programs
- 4) portions of the main routine of the EHSI program
- 5) key routines of the EHSI program

The background interrupt process retrieves all interrupts sent by the DACI and stores them in a 20-byte stack. The stack structure was implemented so no interrupts would be missed while display data is being generated in the EHSI routines. This interrupt process is installed when the EHSI system switch is turned "on". Note, the EHSI program must running before the system switch is turned "on".

The interface communications routines allow the host computer to talk with the DACI. They are written in 8088 assembly language and handle all the communications with the DACI. They also allow the EHSI program to use the four basic DACI commands which were discussed previously.

The 8088 assembly routines had to be interfaced with the EHSI program which is written in C. This involved using several public data structures which could be accessed both the assembly and C routines. These public structures were necessary so that both the assembly routines and the C routines could access them. There are two main public structures which were used. The first one contains the flight simulator data collected by the DACI. It is called the data pkg. The second contains the screen data to sent to the VGD. It is called SCREEN and is basically a single dimensional array of display commands in word form.

All the assembly communications routines can be accessed by the C routines. These include the routines which send

commands to the DACI. The DACI command routines are which can be accessed are the following:

- 1) GET_DATA_PACKAGE()
- 2) SEND_SCREEN()
- 3) TOGGLE_ALARM_SWITCH()
- 4) RECEIVE SCREEN()

The GET_DATA_PACKAGE routine requests the DACI to send the twenty-four flight data values retrieved by the DACI. As the data values are transferred from the DACI to the host they are stored in the data_pkg structure which was mentioned previously. Once this has been accomplished any C routine can access the flight simulator data.

The SEND_SCREEN routine requests the DACI to accept the vector graphics display data in the SCREEN structure and transfer the data to the VGD. This display data which is generated by the C routines consists of the necessary commands to generate the various graphics needed for the different pages of the EHSI. The routine sends the data a word at a time until an hexadecimal value of FF is encountered in the upper byte of a screen data word. The routine then sends the word with FF in the upper byte to the DACI to indicate to the DACI that this is the end of the screen data being sent. The SEND_SCREEN routine then terminates.

The TOGGLE_ALARM_SWITCH routine requests that the DACI toggle the bit which activates or deactivates the piezo-

electric alarm on the DACI. This routine is used when enabled alarm conditions are impending. This alarm will indicate to the pilot that an alarm condition exists. It will be coupled with a warning message to indicating the cause of the alarm.

The RECEIVE_SCREEN routine requests the DACI to send a portion of the display memory. The portion of the display memory to be read is sent by the routine to the DACI. It consists of a beginning address and ending address of the block of the block to be read. As the block of memory is transferred by the DACI a word at a time it is stored in the SCREEN data structure. This data can then be accessed by the C routines for purposes of analysis.

The main routine is the controlling routine for all the vector graphics development of the EHSI. It is written in C and performs several basic functions. First, the background interrupt process which was discussed previously is installed. Second, initialization of data structure values passed to other routines is accomplished. Third, the static data for the various pages of the EHSI is sent to the VGD. Fourth, the data page is put as the first page to be displayed. Fifth, the routine watches for interrupts received from the DACI by the background interrupt process. Sixth, it evaluates the interrupts received from the DACI. If they are valid interrupts, it takes appropriate action to execute

the DACI interrupt. Note, the main routine and its functions will be discussed in more detail in Chapter 2.

The key routines implement the DACI interrupts generated by the control keyboard. These routines are written in C and consist of the following:

- update_key_buffer: This routine converts the DACI interrupts for the numbers 0 - 9 and the decimal point into characters and stores them in a buffer. It also displays the buffer on the VGD.
- 2) roll_stack : This routine pushes the present value in the key buffer onto the two number HP-style stack used for the calculator functions.
- clear_stack: This routine clears out both locations of the HP-style stack used for calculator functions.
- 4) insert_new_freq : This routine enters the communication frequency, navigational frequency, or the ADF frequency which is in the key buffer and stores it in the proper frequency variable. If the frequency entered is out of range, an error message is displayed on the VGD.
- 5) set_timer: This routine enters the timer value in the key buffer and stores it in the timer variable. Note, the timer value is entered in minutes and seconds with a decimal point between them. The routine separates the number into minutes and seconds.
- 6) reset_alarm : This routine is used to toggle the alarm on the DACI. It is just like an on/off switch for the alarm.
- 7) do_math: This routine implements all the math functions that are available on the control keyboard. The number which is presently on the bottom of the stack is the first operand and the number in the key buffer is the second operand. The operation to be performed is implemented on the two operands and the result is displayed on the VGD. Note, if the denominator is equal to zero when a divide operation is being implemented, an appro-

priate "zero divide error" is displayed on the VGD.

8) display_data_page, display_nav_page,
display_ils_page:
These routines change the internal jump in the
first word of the vector memory of the VGD.
The jump is changed to point to the appropriate
section in vector memory which contains the information used to generate the graphics for the page
control key that has been pressed. Thus, this
routine switches among the various pages displayed
by the EHSI, depending on what page control key
has been pressed.

These routines along with the other areas of the hostside developed by Gruenbacher will be discussed in more detail in [3].

1.4.3 EHSI Host-side Development: Robertson's Accomplishment's

Robertson's portion of the host side of the EHSI development system consisted of developing all the necessary code to generate the various pages of the EHSI. This involved doing the following:

- Data structures passed to the routines that generate the various pages of the EHSI were developed.
- Basic tools used to generate the vector graphics displays for each page were developed.
- The static and dynamic portions of the FLIGHT DATA PAGE were developed.
- The static and dynamic portions of the NAVIGATIONAL PAGE were developed.

- The static and dynamic portions of the ILS PAGE were developed.
- 6) Command line and warning areas were developed for each of the pages for displaying appropriate warning messages, entering pertinent flight data, and using the calculator functions of the control keyboard.

All of the above areas will be discussed in detail in subsequent chapters, but a brief overview of each of the areas follows.

Data structures were needed to pass information between the various routines of the EHSI program. There are two structures at the present time which are being used. The first structure is is the clock package. It contains the information which is used by the timer function of the EHSI and the frequencies, altitudes, and estimated wind values which can be entered by the pilot via the control keyboard. The second structure is the alarm package. It contains the various alarm flags which are used to indicate the status of the various warning and alarm conditions which are entered by the pilot.

To generate the various pages of the EHSI some basic tools had to be developed. These tools do some of the general things which are needed on all of the pages. For example, a routine was written to insert data which has been generated into the screen data structure which was discussed

earlier. Another routine generates a line given two separate coordinates. There are many other tools which were developed and they are discussed in a Chapter 3.

The FLIGHT DATA PAGE was developed in two different parts. First, the static portion of the page was developed. The static portion contains those portions of the DATA PAGE which will not change. This includes all the various titles for the information being displayed and certain portions of the climb rate indicator. The routine which generates this data is run only once when the EHSI system is powered up. The data generated is stored in the vector memory of the VGD at a certain location. Next, the dynamic portion of the data page was developed. This portion contains all various information which can change during the course of a flight. For example, the air speed and altitude are generated by the dynamic routine and displayed next to their appropriate titles which were generated by the routine which generates the static information. It should be noted that the dynamic routine has to manipulate the data which collected by the DACI and convert it into value which usable.

The NAVIGATIONAL PAGE and the ILS PAGE are also divided into static and dynamic portions. Once again the routines used to generate the static data are only run once upon power up of the EHSI system. This static data is stored in

its proper location in the vector memory of the VGD. The dynamic portions of these two pages are much more complicated than that of the FLIGHT DATA PAGE. The algorithms and procedures for these portions will be discussed in subsequent chapters of the thesis. These pages involve converting data received from the DACI, analyzing the converted data, and generating the appropriate vector display data to produce the graphics for each of the pages.

Command line and warning areas are used to display pertinent information to the pilot. These areas are slightly different for each page but contain the same basic information. This information includes appropriate warning messages when conditions occur which warrant the pilot's attention. Also, flight parameters entered by the pilot on the control keyboard are displayed along with entries which are to be operated on by the calculator functions of the EHSI.

1.4.4 EHSI Host-side Development: Robertson and Gruenbacher

There were a few things which were added to the EHSI through a collaborative effort by Gruenbacher and myself. First, dedicated interrupt lines were added to the system. This involved both hardware and software modifications to the DACI and software modifications to the host. The changes to the DACI appear in Appendix A. Second, the clock interrupt routine of the DACI was modified to analyze what

would need to be done to speed up the EHSI system. involved changing the main routine of the DACI to retrieve the clock data every time in its main loop rather than on an interrupt from the real time clock, disabling the real time clock interrupt, and removing the interrupt handshaking from the clock interrupt routine. The results of this test showed that a faster processor is needed on the host-side and that a key decoder with latched outputs is needed for the control keyboard. These changes were then removed and the DACI routines were returned to their original state. Third, a switch was added to the DACI to allow the input of an external clock to the clock interrupt line of the DACI. This was also added to analyze the speed of the EHSI system. It also served to demonstrate the operation the interrupt stack on the host side. Finally, as a temporary cure for noise problems, the flat cables of the EHSI system were twisted and wrapped in a foil shield, which was grounded to the chassis of the host computer.

1.5 EHSI Host-side Programming Model

The functional programming model for the EHSI consists of five basic levels:

- 1) BACKGROUND INTERRUPT PROCESS
- 2) INTERRUPT STACK
- 3) MAIN ROUTINE

- 4) EHSI FUNCTION MODULES
- 5) COMMUNICATION ROUTINES

These levels can be seen in Fig. 1.6. The paths between these different levels can also be seen in the figure.

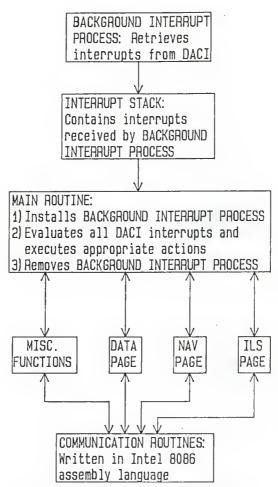


Figure 1.6 EHSI HOST PROGRAM MODEL

CHAPTER TWO

MAIN ROUTINE

The main routine is the controlling routine for all the vector graphics development of the EHSI. It is written in the C programming language and has to perform several basic functions. This routine continually loops once it is started and terminates when the EHSI system switch is turned "off."

2.1 Functions

The first function performed by the main routine is the installation of the background interrupt process. The background interrupt process retrieves all the interrupts sent by the DACI and stores them in a twenty-byte stack. This stack decreases the possibility of missing a DACI interrupt while display data is being generated by the functions called by the main routine.

Second, all the data structure values passed to other routines are initialized. This includes clock package val-

ues such as the timer minutes and seconds, time-out minutes and seconds, ADF frequency, communication frequencies, navigation frequencies, timer status, and timer operation. Alarm conditions in the alarm package are also initialized to the off state.

Note, at this point, the main routine enters a continuous loop and waits for the "system on" interrupt to be sent by the DACI. The main routine is executing the clock interrupt at this time, but the static information for the different display pages has yet to be installed. Therefore, nothing is displayed on the vector graphics display.

Third, when the "system on" interrupt is sent by the DACI, a system on message is displayed and the static data for the different pages to be displayed is installed in the vector graphics memory. The term static is used here to indicate display data that will not change for the various pages. This includes such things as headings for the various flight data that are to be displayed. The DATA PAGE of the EHSI is now displayed on the VGD. This page will always be the first page shown when the "system on" interrupt is received from the DACI.

Fourth, the main routine continues looping and checks for interrupts from the DACI. When an interrupt is received it is then evaluated and checked for validity. There are three types of valid interrupts and they consist of the

following:

- 1) key interrupts
 - A) function keys
 - B) number keys
 - C) page keys
- 2) clock interrupts
- 3) system status
 - A) system on
 - B) system off

interrupts are generated when a key is pressed on the control keyboard of the EHSI. The function key interrupts are used to execute the various functions which are available. These functions include entering the ADF frequency, communication frequencies, navigational frequencies, timer functions, calculator functions, etc. When an interrupt is evaluated to be a key interrupt, the appropriate function used to execute the key interrupt is called by the main routine. Upon completion of the key interrupt routine, control is returned to the main routine. Number key interrupts are used to enter numerical values into a buffer. These numerical values in the buffer are then used when the various function keys of the EHSI are pressed. Page key interrupts are used to determine which page is to be displayed by the EHSI. Remember, the system first displays the However, after the "system on" interrupt is DATA PAGE. received, any of the three pages available on the EHSI may be selected by pressing the appropriate page key on the control keyboard.

Clock interrupts are interrupts generated by the DACI's real-time clock. Presently, the interrupt occurs every half second. This corresponds to a frequency of two hertz. When a clock interrupt is received, the routine which generates the dynamic data for the present page being displayed is called. The term dynamic here is used to refer to the vector graphics commands which are subject to change for the various displays.

System status interrupts are generated when the system switch of the EHSI is turned "on" or "off". As was stated earlier, the "system on" interrupt results in the installation of all the static data for the various pages into the VGD memory. When the "system off" interrupt is received from the DACI, the main routine removes the background interrupt process which was installed, displays a system off message, and exits to DOS.

The basic functions of the main routine can be seen in Fig. 2.1.

2.2 Data Structures

The main routine has four different data structures which it accesses. Two of these structures are public structures and two are local to the main routine. The reason for the use of the public structures was to allow the interface of the C routines and the assembly routines of the

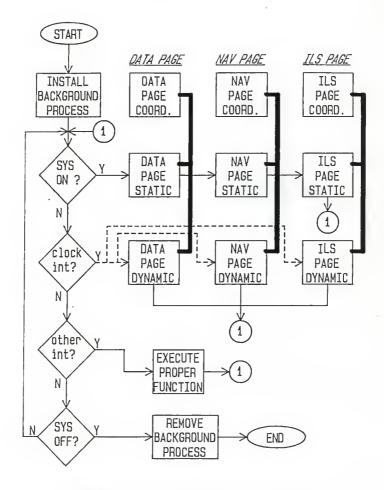


Figure 2.1 MAIN ROUTINE BASIC FUNCTIONS

host-side.

The first of the public data structures is the data package. This structure contains all the raw digital flight values obtained by the DACI from the flight simulator. There are a total of twenty-four different values. These values are manipulated and converted by various routines to generate the various pages of the EHSI. They are retrieved by calling the assembly routine, GET_DATA_PACKAGE(), which was mentioned in the introduction.

The second of the public data structures is the screen data. This structure contains all the VGD commands which are to be sent to the VGD. It is one-dimensional array consisting of up to one thousand different display commands. These commands are sent by calling the assembly routine, SEND_SCREEN(), which was mentioned in the introduction.

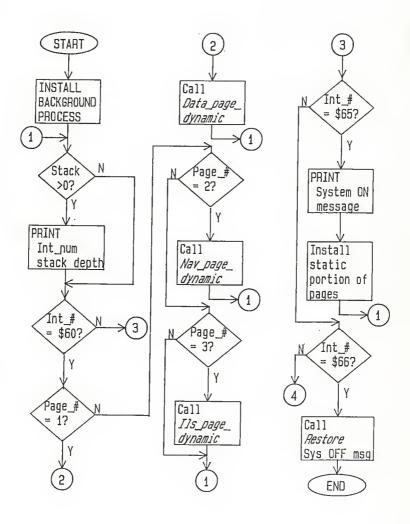
The third data structure is the clock package. This structure is local to the main routine and contains a variety of values which are passed by the main routine to various functions which it calls. This structure contains the following:

- 1) time values
 - A) timer minutes and seconds
 - B) time-out minutes and seconds
 - C) timer operation flag
 - D) timer status flag
- 2) frequency values
 - A) ADF frequency
 - B) communication frequencies
 - C) navigation frequencies

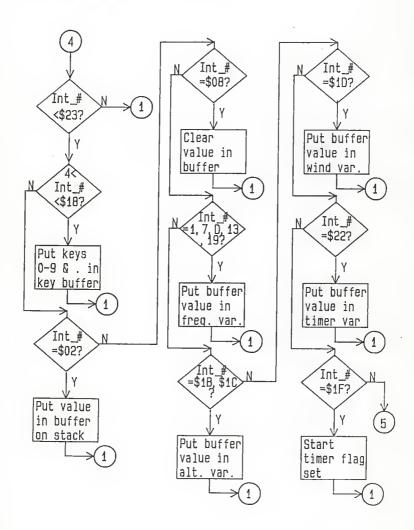
- 3) altitude values
 - A) assigned altitude
 - B) MDA/DH
- 4) estimated wind direction
- 5) math operation flag

The fourth structure is the alarm package. This structure is also local to the main routine and is passed by the main routine to other routines. This structure contains the alarm flags for the EHSI. Presently there are alarm flags for the airspeed, assigned altitude, MDA/DH, and the timeout. The structure also contains alarm enable flags for the MDA/DH and assigned altitude and a system alarm status flag.

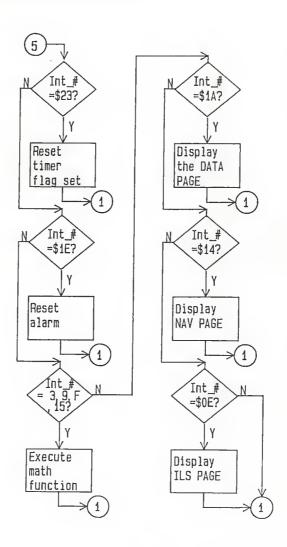
2.3 High Level Flowchart



2.3 High Level Flowchart (cont.)



2.3 High Level Flowchart (cont.)



CHAPTER THREE

TOOLS FOR VECTOR GRAPHICS DEVELOPMENT OF THE EHSI

The first step in developing the vector graphics displays for the EHSI was the development of some basic graphics tools. These tools were needed to facilitate the task at hand. These tools consist of functions which were written in the C programming language. They accomplish a variety of separate tasks and can be divided into two different The first category consists of those functions categories. which have a general-purpose nature. These functions or tools are used in the development of the graphics for all the pages of the EHSI. The second category consists of those tools which are page specific. These page-specific tools accomplish tasks which are unique to the vector graphics development of specific pages of the EHSI.

3.1 General-Purpose Tools

There are many different general purpose tools which

were developed to help generate the different pages of the EHSI. For the most part, these tools were the first functions that were written to ease the burden of the vector graphics development for the EHSI. Presently, these tools consist of the following:

- 1) insert
- 2) line
- 3) arc_circ
- 4) string gen
- 5) box
- 6) zero pad
- 7) altitude
- 8) dime
- 9) airspeed
- 10) arrow
- 11) heading
- 12) timer

The insert function inserts a conversion array into the screen data structure which was discussed earlier. It inserts this array based on an index which is passed to the routine. The conversion array contains the VGD commands which have been generated by other functions. The routine only inserts a certain number of commands in the conversion array. The number of commands inserted is determined by a length passed to the routine. This routine is used more

than any of the other general routines.

The line function generates the necessary VGD commands to plot a line between two points on the vector graphics display. The coordinates of the end-points are passed to the routine and the VGD commands which are generated are stored in a conversion array.

The arc_circ function generates the necessary VGD commands to generate an arc between zero and three hundred sixty degrees. The plot origin for the arc is at the center of the arc. The arc is drawn at a certain radius. The routine generates this arc by drawing many separate line segments. The angle increment for drawing the line segments is determined by the radius, with a larger radius having a smaller angle increment. This is necessary to make the arc appear to have a smooth curvature. The VGD commands generated are stored in a conversion array. The plot coordinates, start angle, end angle, and radius are all passed to the routine.

String gen is a function which generates the necessary VGD commands to place a string at a given plot coordinate on the VGD. It can generate characters of four different sizes. The character size, plot coordinate, and string are all passed to the routine. The commands generated are stored in a conversion array. This routine is used extensively.

Box is a function which used to generate a box around a given string being plotted on the display. The size of the box generated is based on the number of characters in the string and the size of the characters. The location of the box is based on the plot location of the string on the display. The plot coordinates, character string, and character size are all passed to the routine. All the VGD commands generated are stored in a conversion array.

The zero_pad function is used to zero pad values which are to be displayed. Zero padding is just the addition of zeros to a value to maintain the number of digits shown at a constant. The zeros are added to the front of a number. For example, since a compass can have a heading of up to 360 degrees, it might be desirable to have the heading displayed with a three digit field. In the case of a heading of 185 degrees, no zero padding would be necessary. However, a heading of 5 degrees would be zero padded to 005 degrees. This function takes a given number to be padded , zero pads the number based on a given pad limit, and converts the padded value to a string. The value to be padded, the pad limit, the precision, and the type of number to be padded are all passed to the function. The padded value is stored in a conversion string.

Altitude is a function which retrieves the digital altitude value from the data package structure and converts

this value using a linear conversion into feet. This converted value is zero padded and is plotted as string at a given location on the VGD. This routine also monitors the alarm conditions for the assigned altitude and MDA/DH. It will control the alarm horn of the EHSI and will display appropriate warning messages when necessary. The clock package, alarm package, and plot coordinates are passed to the routine. The VGD commands generated are stored in a conversion array.

Dime is a function which retrieves the digital DME value from the data package structure and converts this value using a linear conversion into nautical miles. This converted value is zero padded and is plotted as string at a given location on the VGD. The clock package, alarm package, and plot coordinates are passed to the routine. The VGD commands generated are stored in a conversion array.

Airspeed is a function which retrieves the digital airspeed value from the data package structure and converts this value using a linear conversion into knots. This converted value is zero padded and is plotted as string at a given location on the vector graphics display. This routine also monitors the alarm conditions for the stall airspeed of the aircraft. Later, a structural-limit airspeed may be added. It will control the alarm horn of the EHSI and will display appropriate warning messages when necessary. The

clock package, alarm package, and plot coordinates are passed to the routine. The VGD commands generated are stored in a conversion array.

Arrow generates the necessary VGD commands to plot an arrow head on the display. The arrow is plotted for a given radius, location, angle, and direction. The direction determines whether or not the arrow head points "to" or "from" the given plot location. The commands generated are stored in a conversion array.

Heading retrieves the digital value for the plane heading from the data package and converts it into an appropriate plane heading in degrees. It accomplishes this conversion by using a lookup table which is in the form of a structure. If the value retrieved does not equal a value in the lookup table, interpolation is used to get the appropriate plane heading. The plane heading found is passed to the calling routine.

Timer implements the TIMER and TIME-OUT function available on the EHSI. The routine keeps track of the timer value and compares it to the time-out value which is entered by the operator via the control keyboard. The routine reacts to all key presses dealing with the timer. These include the "TIMER" key, "RESET TIMER" key, and "START TIMER" key of the control keyboard. When the timer value is greater than or equal to the time-out value which has been

entered, the alarm horn of the EHSI is sounded and an appropriate message is displayed. The only page which displays the timer and time-out value is the DATA PAGE. The routine stores all generated vector graphics commands in a conversion array. The clock package and the alarm package are required by the routine.

3.2 Page-Specific Tools

Each of the pages of the EHSI has certain portions which are unique to that particular page. To generate these unique graphics, functions had to be written to accomplish these tasks. These functions are page-specific, which means that they are only used in the development of a graphical portion of one specific page of the EHSI. The page-specific tools can thus be broken down into categories defined by the EHSI pages.

3.2.1 DATA PAGE Tools

The DATA PAGE tools were the first page-specific tools which were developed. For the most part, they are very simple routines because the graphics generated for the DATA PAGE are not that complicated. The page-specific tools for the DATA PAGE consist of the following:

- 1) time_string_gen
- 2) climb_box

- 3) clim_hsh
- 4) clim_arrow
- 5) climb_rate

The Time_string_gen routine is used to generate a string depicting the time in hours, minutes, and seconds. The routine retrieves the values of the real-time clock on the DACI from the data package. These values, which consist of the hours, minutes, and seconds, are converted into characters and are zero padded when appropriate. These zero-padded values are put together as a string with colons between them.

Climb_box is a routine which generates the vector VGD commands for the climb indicator box of the DATA PAGE. It is a very straight forward routine which consists of generating the five necessary connected line segments which form the climb indicator box. The only values needed by the routine are the plot coordinates for the box. The generated commands are stored a conversion array.

The clim_hsh routine is also a very straight forward routine. It generates the hash marks for the climb indicator of the DATA PAGE. These hash marks are just line segments plotted at certain intervals along the climb indicator arc. Four hash marks are generated by this routine at evenly spaced intervals. The intervals are determined by dividing a given arc into four parts. Note that the routine

does not put a hash mark in the middle of the arc. The information required by the routine consists of the plot coordinates of the climb indicator arc, the start and end angle of the climb indicator arc, and the radius of the climb indicator arc. The VGD commands generated by the routine are stored in a conversion array.

Clim_arrow is a routine which generates the VGD commands to plot the climb indicator arrow of the DATA PAGE.

The routine generates this arrow given a plot location, angle, and radius. This arrow consists of three connected line segments. One corresponding to the shaft of the arrow and the other two making up the head of the arrow. The VGD commands for the arrow are stored in a conversion array.

Climb_rate is a function which retrieves the digital vertical speed value from the data package structure and converts this value using a linear conversion into feet per minute. This converted value is zero padded and is plotted as string at a given location on the vector graphics display. The clock package, alarm package, and plot coordinates are passed to the routine. The VGD commands generated are stored in a conversion array.

3.2.2 NAV PAGE Tools

The tools needed for the vector graphics development of the NAV PAGE vary greatly in their complexity. Some are

very straight forward and simple while others are rather complex and involve a large number of calculations and decisions to accomplish their tasks. The simple tools of this page consist of the following:

- 1) plane
- 2) waypoint
- 3) vortac
- 4) ndb

All of these simple routines basically the same. The first three involve connecting numerous line segments to form the appropriate graphic to be displayed. The ndb routine makes three calls of the aro_circ function which was discussed in the general tools section. The only information required by these routines is the plot coordinates for the graphic. All the VGD commands are stored in a conversion array.

The more complicated tools of the NAV PAGE consist of the following:

- 1) compass
- 2) heading and bearing
- 3) get_ndb_plot_engle

Compass is the routine which generates the vector graphics commands to plot the hash marks and appropriate markings for the compass portion of the NAV PAGE. A ninety degree portion of the compass is generated showing forty-

five degrees to either side of the plane's heading. The information which is required by this routine consists of the plot coordinates for center of the compass arc, the radius of the compass arc, and the heading of the plane. Hash marks are generated in intervals of five degrees. The hash marks which are at a compass headings which are divisible by ten are twice as long as those that are not divisible by ten. Markings along the compass are done at compass headings which are divisible by ten. These markings consist of headings between 0 and 360 degrees with markings of N, E, W, and S for respective headings of 0 degrees, 90 degrees, 180 degrees, and 270 degrees. The VGD commands generated by this routine are stored in a conversion array.

Heading_and_bearing calculates the heading and bearing of an NDB or VORTAC. It calculates these, given information consisting of the plane's heading in degrees and the plot angle of the VORTAC or the NDB. After calculating the heading and bearing the routine generates the appropriate VGD commands to plot the heading and bearing as strings on the VGD. The heading is plotted at distance two-thirds of the way between the NDB or VORTAC. The bearing is plotted at a distance one-third of the way between the The distance of the NDB or VORTAC is passed to the Arrows are also drawn by the routine, with an arrow pointing "to" the NDB or VORTAC at the heading location and an arrow pointing "from" the NDB or VORTAC at the bearing location. The VGD commands are stored in conversion array.

Get_ndb_plot_angle retrieves the digital value for the ADF from the data package and converts it into an appropriate NDB plot angle. It accomplishes this conversion by using a lookup table which is in the form of a structure. If the value retrieved does not equal a value in the lookup table, interpolation is used to get the appropriate plot angle for the NDB. The angle found is passed to the calling routine. The converted NDB plot angle is in degrees.

3.2.3 ILS PAGE Tools

Presently, there are only two page-specific tools on the ILS PAGE. The first tool is the runway function. Runway generates the necessary VGD commands to plot a runway on the display. The runway is plotted for a given location, scale factor, and angle. The plot coordinate is centered at the intersection to the runway centerline and touchdown line. The runway is drawn using vanishing-point techniques. The commands generated are stored in a conversion array.

Ils_compass is the second tool. This routine generates the vector graphics commands to plot the hash marks and appropriate markings for the compass portion of the ILS PAGE. A twenty-degree portion of the compass is generated

showing ten degrees to either side of the plane's heading. The information which is required by this routine consists of the plot coordinates for center of the compass arc, the radius of the compass arc, and the heading of the plane. Hash marks are generated in intervals of one degree. The hash marks which are at compass headings divisible by five are twice as long as the other hash marks generated. Markings along the compass are done at compass headings which are divisible by ten. These markings consist of headings between 0 and 360 degrees with markings of N, E, W, and S for respective headings of 0 degrees, 90 degrees, 180 degrees, and 270 degrees. The VGD commands generated by this routine are stored in a conversion array.

CHAPTER FOUR

HP 1345A VECTOR GRAPHICS

The HP 1345A is a high-resolution vector graphics display. It was chosen for the EHSI development system because it is especially suited for this type of application. The display is capable of operating up to 15,000 feet pressure altitude. The plotting area is 9.5 cm high and 12.5 cm wide and has 2048 addressable points in either direction. The display also contains the vector memory option which allows storage of up to 4096 command words and provides automatic refresh capability for the CRT. The display also has a built-in character set.

The HP 1345A accepts four basic commands:

- 1) SET CONDITION
- 2) PLOT
- 3) TEXT
- 4) GRAPH

These commands allow complete text and vector genera-

tion, while keeping programming overhead to a minimum.

Note, each command word is 16 bits long. The GRAPH command was the only command which was not used in the development of the EHSI.

The SET CONDITION command is used to establish vector attributes. The attributes which it affects are line type, speed, and intensity. Using the line intensity and speed parameters, the user can generate up to 12 different intensity levels. This allows the user to intensify certain areas of the display and generate background areas. The vector generated will be the brightest when the line intensity is set at full brightness and the slowest writing speed. Once the SET CONDITION command has been executed, it stays in effect until another SET CONDITION command is executed. This command can be executed at any time. The bit patterns required for the SET CONDITION can be found in the found in Fig. 4.1(4).

The PLOT command is used to draw vectors on the display. The command moves a beam to a specific location in the Cartesian plane of the display each time an X-Y coordinate pair is received. The origin of the Cartesian plane is in the lower left hand corner of the display. The X-Y value at the origin is equal to (0,0). The plotting range is from 0 to 2047 in both the X and Y the directions. The beam can also be turned on and off by the command. A vector

is drawn from the previous beam coordinates to the coordinates specified by a new PLOT command. Note, all vectors are drawn in accordance with last SET CONDITION command received by the display. The bit patterns for the PLOT command can be found in Fig. 4.2[4].

An example will demonstrate how easy it is to use the PLOT command of the HP 1345A. Sending the following command words will draw a vector on the display. The command words are 16 bits wide and are shown in hexadecimal format.

CO	MMAND	SENT		COMMAND	WORD
1)	SET (CONDITIO	1	\$78	318
2)	PLOT	XO(beam	off)	\$00	000
3)	PLOT	YO(beam	off)	\$08	300
4)	PLOT	X1(beam	off)	\$00	FF
5)	PLOT	Y1(beam	on)	\$18	BFF

Executing these commands will draw a line from the origin on the display to a point located at (255,255) on the display. The first command sets the line type to full solid, full brightness, and the slowest writing speed. The next two commands move the beam to the origin of the display with the beam off. The last two commands move the beam to (255,255) with beam on.

If a vector is to be drawn vertically, the Y value is the only command which is needed to establish the vector endpoint. This is possible because the display has a "last

X" register which stores the value of the last X location entered. This allows drawing of vertical vectors with one less command word.

There are a few limitations which should be noted when using the PLOT command of the HP 1345A. It was found that when drawing a vector, only one of the commands for vector end location should be executed with the beam When both the commands for the end location were executed with the beam on, the display reacted unpredictably. The user should also take into account that the display width and height are different. But, there are 2048 addressable points in either direction. If this difference is not taken into account, a box will appear as a rectangle and a circle as an ellipse. To correct for this difference, a scaling factor must be used when calculating vector endpoints. screen is 9.5 cm high and 12.5 cm wide; a scaling factor can be derived from this information. The user should also make sure that all vectors are plotted within the usable plotting area of the display. If vectors are drawn outside the vector drawing area, the vectors will be drawn to the opposite side of the display. This phenomenon is known as "wrap around". This can result in a very distorted image and should be avoided at all times.

The TEXT command is used to plot characters on the HP 1345A. The display has a complete built-in character set.

It is a modified ASCII set for graphics use. A portion of the set can be seen in Fig. 4.3[4]. The characters drawn on the display are drawn in accordance with the last SET CONDITION command received. The characters are always drawn at the slowest writing speed. Note, line type has no affect except on the largest character size. Characters are drawn at the last X-Y coordinates received.

Automatic spacing is provided between the characters when they are generated. Characters can be rotated up to four different ways and can be drawn in four different sizes. The starting position for each character is in the lower left hand corner of the defined character cell. Bit patterns for text generation can be found in Fig. 4.4[4].

The "wrap around" phenomenon which was discussed earlier applies to text generation as well. Characters should not be plotted to close to the plotting boundaries or "wrap around" will occur. If the boundary specifications shown in Fig. 4.5[4] are followed, no problems will be encountered when writing text at the screen boundaries.

The vector memory option of the HP 1345A makes the job of programming the display much easier. It allows the user to store up to 4096 command words and provides automatic refresh for the CRT at a 60 Hz rate or a rate defined by an external clock. This relieves the user processor of data storage and refresh requirements.

The vector memory recognizes two commands for programming. These commands are for memory address pointer manipulation and data transfer. Address-pointer operations are used for positioning data in the vector memory. Data transfer can be either a read or a write of data in the vector memory.

There are two pointers used to control access of data to and from the vector memory. The first pointer is not a user accessable pointer and is used for refresh purposes. The second pointer is the vector memory address pointer. This is the pointer which is used to control the part of the vector memory which is being accessed by the user. The bit patterns for the available vector memory commands can be seen in Fig. 4.6[4].

MSB
D14 D13 D12 D11 D10 D09 D08 D07 D06 D05 D04 D03 D02 D01 D00
1 1 II IO X X L1 L0 O X W1 W0 X X X

Note: Bit 6 (DO6) must be zero.

Command Modifiers:

a. To Set Line Intensity:

I1	IO	Intensity
0	0	Blank
0	1	Dim
1	0	Half Brightness
1	1	Full Prightness

b. To Set Line Type:

L1	LO	Type	
O	0	Solid Line	
0	1	Intensified End	Points
1	0	Long Dashes	
1	1	Short Dashes	

c. To Set Writing Speed:

W1	WO	Speed		
1	1	0.05 in.	per	microsec
1	0	0.10 in.	per	microsec
0	1	0.15 in.	per	microsec
0	0	0.20 in.	per	microsec

Figure 4.1 SET CONDITION COMMAND [4]

Command Modifiers:

- a. XY Information (D12)
 - 0 = X coordinate (0-2047), specified by D00-D10 1 = Y coordinate (0-2047), specified by D00-D10
- b. PC Beam Control Information (D11)
 - O = Beam OFF (move)
 - 1 = Beam ON (draw)

Figure 4.2 PLOT COMMAND [4]

		MOST	SI	GNIFICAN	Т	CHARACTE	ER
		2	3	4	5	6	7
LEAST SIGNIFICANT CHARACTER	0 1 2 3 4 5 6 7 8 9 A B C D E F	SP # \$ % & () * + , /	0123456789:;<=>?	I	PQRSTUVWXYZ[\]^	ab c d e f g h i Jk i m n O	pqrstuvwxyx{ }
					_		

Figure 4.3 Partial 1345A MODIFIED ASCII CHARACTER SET [4]

MSB														LSB
D14	D13	D12	D11	D10	D09	D08	D07	D06	D05	D04	DO3	D02	DO1	DOO
1	0	S1	SO	R1	RO	ES	D07	D06	D05	D04	DO3	D02	DO1	DOO
	CHARACTER>									>				

Command Modifiers:

- a. ES Establish Size of Character
 - O = Use previous size and rotation
 - 1 = Establish new size and rotation according to S1, S0, R1, and R0
- b. Rotate Character CCW
 - R1 R0 Rotation O O degrees O 1 90 degrees 1 O 180 degrees 1 270 degrees
- c. Character Size

51	so	Size	W	Х	Н
0	0	1 X	24	Х	36
0	1	1.5X	36	Х	54
1	0	2. OX	48	Х	72
1	1	2.5X	60	Х	90

Figure 4.4 TEXT COMMAND [4]

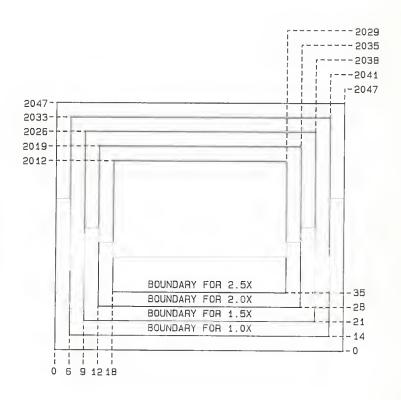


Figure 4.5 BOUNDARY LIMITS FOR CHARACTER PLOTTING [4]

VECTOR MEMORY WORD

MSB

LSB
M15 M14 M13 M12 M14 M10 M9 M8 M7 M6 M5 M4 M3 M2 M1 M0

0 B14 B13 B12 B11 B10 B9 B8 B7 B6 B5 B4 B3 B2 B1 B0

Note: See data bit definitions for 1345A commands

INTERNAL JUMP

MSB
M15 M14 M13 M12 M11 M10 M9 M8 M7 M6 M5 M4 M3 M2 M1 M0
1 O X X A11 A10 A9 A8 A7 A6 A5 A4 A3 A2 A1 A0

Note: All-AO specify jump to vector memory address during refresh

ADDRESS POINTER

MSB
M15 M14 M13 M12 M11 M10 M9 M8 M7 M6 M5 M4 M3 M2 M1 M0
1 1 X X A11 A10 A9 A8 A7 A6 A5 A4 A3 A2 A1 A0

Note: Sets address pointer to address specified by A11-A0

Figure 4.6 VECTOR MEMORY COMMANDS [4]

CHAPTER FIVE

VECTOR MEMORY LAYOUT FOR THE EHSI

Since the HP 1345A has the vector memory option, a memory layout for the EHSI was designed. The vector memory was broken up into seven separate areas. Three of these areas contain the static data for the different pages of the EHSI. The four other areas of the memory are used for the dynamic portions of the EHSI. The fourth dynamic area is used for the command line and warning messages which are to be displayed.

The vector memory layout can be seen in Fig. 5.1. The first word of the vector memory contains the jump vector which determines which page of the EHSI is being displayed. The jump vector is changed whenever one of the page keys is pressed on the control keyboard. The first section of memory contains the static information for the DATA PAGE. This is located in the vector memory between \$0001 and \$02FF. Note, the last command of the static data for the

DATA PAGE contains a jump to the location in the vector memory which contains the dynamic portion of the DATA PAGE. The second section of memory contains the static information for the NAV PAGE. This is located in the vector memory between \$0300 and \$04FF. Note, the last command of the static data for the NAV PAGE contains a jump to the location in the vector memory which contains the dynamic portion the NAV PAGE. The third section of memory contains the static information for the ILS PAGE. This is located in the vector memory between \$0500 and \$06FF. Note, the last command of the static data for the ILS PAGE contains a to the location in the vector memory which contains the dynamic portion of the ILS PAGE. The fourth section memory contains the dynamic information for the DATA PAGE. This is located in the vector memory between \$0700 and \$08FF. The fifth section of memory contains the dynamic information for the NAV PAGE. This is located in the vector memory between \$0900 and \$0BFF. The sixth section of memory contains the dynamic information for the ILS PAGE. located in the vector memory between \$0000 and \$0DFF. Note, last command of the dynamic data for all the pages of EHSI contains a jump to the location in the vector memory which contains the command line and warning area for the EHSI. The final section of memory contains the command line and warning area for the EHSI. This is located in

vector memory between \$0E00 and \$0FFE. The final word of the vector memory contains a NO-OP. This NO-OP is necessary because a jump may not be made to another internal jump.

This memory layout can be changed at any time with the programming capability of the vector memory. This will make it possible to change the memory layout easily, when changes are made to the host-side of the EHSI program.

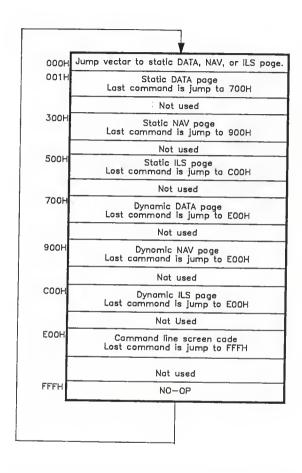


Figure 5.1 VECTOR MEMORY LAYOUT FOR THE EHSI

CHAPTER SIX

DATA PAGE

The DATA PAGE provides the pilot with basic flight information. It shows flight parameters which are inputed by the pilot prior to take-off or during the course of a flight and basic flight data showing the plane's status. Enabled alarm conditions are also shown on this page to alert the pilot as to which of the available alarms are set. For the most part, this page is in a digital type format. This means that most of the flight parameters are shown as digital values. The one portion of the DATA PAGE which is analog in appearance is the rate of climb indicator. This indicator will show the plane's rate of climb in feet per minute. An example of the DATA PAGE can be seen in Fig. 6.1.

The DATA PAGE is broken down into two basic parts. The first part consists of the static information for the page. This is information which will not change during the course

of the EHSI operation. This static information is loaded into the vector memory only once. This loading occurs when the EHSI system switch is turned "on". The second part consists of the dynamic information for the page which is generated during the course of the flight. When the DATA PAGE is the present page being displayed, this information gets loaded into the vector memory every time a clock interrupt is received from the DACI.

6.1 Static Portion

The static portion of the DATA PAGE can be broken down into three areas. The first area consists of text strings. These strings provide the headings for the information being displayed on the DATA PAGE. The strings are all generated using the (1.5) character size of the HP 1345A's built-in character set. All the plot locations for the strings are in an include file entitled "datpg_xy.h". This was done to allow the developer to easily change the location of heading strings. This makes it possible to try new configurations on the DATA PAGE with a minimal amount of Presently, some of the text strings are grouped together. Some of this grouping is random, while groups of the strings share information which is similar in nature. The VGD commands used to plot the strings were generated using the string gen routine which was discussed

in the tools chapter. The text strings which are presently shown on the DATA PAGE are the following:

- 1) HEADING: plane's present heading
- 2) AIRSPEED
 - A) CAS: calculated airspeed
 - B) TAS: true airspeed
- 3) GROUNDSPEED: plane's speed relative to the ground
- 4) ALTITUDES
 - A) ASSIGNED: plane's assigned flight altitude
 - B) PRESENT: plane's present altitude
 - C) MDA/DH: minimum descent altitude/decision height
- 5) TIMER
- 6) TIME-OUT
- 7) FREQUENCIES
 - A) COMMUNICATION: radio frequencies
 - B) NAVIGATION: VORTAC frequencies
 - C) ADF: automatic direction finder frequency
- 8) WAYPOINTS: intermediate navigation points on flight path
- 9) TEMPERATURE: outside temperature
- 10) BAROMETER: outside barametric pressure
- 11) REAL TIME VALUES
 - A) MILITARY: military standard time
 - B) EDT:
 - C) ZULU: Greenwich Mean Time (GMT)

12) TIME SINCE LIFT-OFF

The second area of the static portion of the DATA PAGE is the climb indicator. The plot location for the indicator is also contained in the include file, "datpg_xy.h". Although the indicator is comprised of several different parts, the whole indicator can be moved by changing just one plot coordinate in the include file. This indicator is in the upper right corner of the display and consists of the following:

- LETTERING: The letters are plotted vertically and spell out the word "CLIMB". The characters used are size (1.5) of the built-in character set. String gen was used here.
- 2) INDICATOR ARC: The arc was plotted using arc_circ. It provides the boundary for the meter.
- INDICATOR CENTERLINES: Line was used to plot these. They provide the zero climb rate reference for the meter.
- 4) CLIMB BOX: The climb_box routine was used here. The climb box provides a starting point for the climb indicator arrow which is in the dynamic part of the page. It also boxes in the indicator's lettering.
- 5) INDICATOR HASH MARKS: These marks provide reference points on the indicator at certain climb rates. These reference points are at rates of -2000, -1000, +1000, and +2000 feet per second. They are plotted using clim h

The third area contains the command line. Below this line is where all messages to the pilot will be displayed. These include both warning and error messages. This area

will show numeric entries from the control keyboard and results of operations performed with the HP-style calculator functions available on the EHSI. This line is plotted using the line routine.

6.2 Dynamic Portion

The dynamic portion of the DATA PAGE is responsible for plotting the flight status information, flight parameters entered by the pilot, and the climb indicator arrow of the climb indicator. This portion also uses the "datpg_xy.h" coordinate file when plotting the vector graphics which it generates. Note, all the VGD commands used to plot strings in this portion of the page are generated using string_gen. Also, all zero padding of values is done with the zero_pad routine. The dynamic section of this page consists of the following:

- HEADING: The heading of the plane is calculated using the get_heading routine. It is then zero padded and plotted.
- AIRSPEED: The airspeed of the plane is calculated using the airspeed routine. It is then zero padded and plotted. Alarm conditions are also monitored by this routine for the stall airspeed.
- GROUNDSPEED: The groundspeed has yet to be implemented.

4) ALTITUDES

A) ASSIGNED ALTITUDE: The assigned altitude is retrieved from the clock package, zero padded, and plotted. This is entered by the pilot.

- B) PRESENT ALTITUDE: The present altitude is calculated using the altitude routine. It is then zero padded and plotted. Alarm conditions are monitored for both assigned altitude and MDA/DH.
- C) MDA/DH: The MDA/DH is retrieved from the clock package, zero padded, and plotted. This is inputed by the pilot.
- 5) TIMER and TIME-OUT: The timer routine is called for both these values. The timer routine calculates the TIMER value and monitors the alarm condition for the TIME-OUT value. The TIME-OUT value is entered by the pilot. Both values are zero padded and plotted.

6) FREQUENCIES

- A) COMMUNICATION: The communication frequencies are retrieved from the clock package, zero padded, and plotted. These are entered by the pilot.
- B) NAVIGATION: The navigation frequencies are retrieved from the clock package, zero padded, and plotted. These are entered by the pilot.
- C) ADF: The ADF frequency is retrieved from the clock package, zero padded, and plotted. It is entered the pilot.
- WAYPOINTS: Waypoints have yet to be implemented.
 They will be entered by the pilot.
- 8) TEMPERATURE: Temperature has yet to be implemented. It will be entered by the pilot and will be displayed in both fahrenheit and celsius.
- 9) BAROMETER: The barometer has yet to be implemented. This will be entered by the pilot.

10) REAL-TIME CLOCK:

- A) MILITARY: The time_string_gen routine is called which retrieves the clock values, zero pads them, and generates the time string. The time string is then plotted.
- B) EDT: EDT time has yet to be implemented.

- C) ZULU: ZULU time has yet to be implemented.
- 11) TIME SINCE LIFT-OFF: This has yet to be implemented.

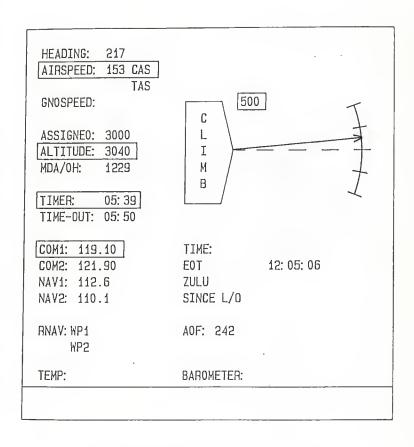
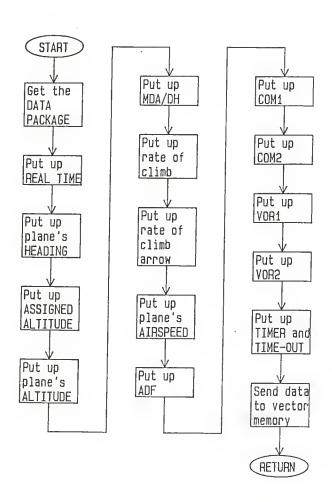


Figure 6.1 PRESENT FLIGHT DATA PAGE

6.3 High Level Flowchart



CHAPTER SEVEN

NAV PAGE

The NAV PAGE is probably the page which is of most interest to the small plane pilot. It provides a large of amount of information to the pilot. This information is assimilated from various flight instruments and is displayed in a format which will show the position of the plane relative to known navigational fixes. Unlike the DATA PAGE, most of the information on this page is provided in an analog type format. An example of the NAV PAGE can be seen in Fig. 7.1. The information supplied on this page consists of the following:

1) HEADING:

- A) The heading is displayed in an analog type compass format.
- B) A digital value for the heading is displayed at the compass heading.

2) VORTAC:

A) The heading "to" and bearing "from" the VORTAC are displayed along with appropriate arrows.

- B) DME information to the VORTAC is displayed.
- C) The VORTAC is displayed in its position relative to the plane.

3) NDB:

- A) The heading "to" and bearing "from" the NDB are displayed along with appropriate arrows.
- B) Later, possibly, distance information to the NDB will be implemented.
- C) The NDB is displayed in its position relative to the plane.
- MISCELLANEOUS INFORMATION: This information is displayed in a digital format.
 - A) AIRSPEED
 - B) GROUNDSPEED
 - C) TRACK
 - D) ALTITUDE
 - E) DME
- 5) WARNING MESSAGE AREA: Warning messages to the pilot are displayed in this area. These messages are generated by various routines.
- 6) COMMAND LINE AREA: This area is for displaying numeric values entered by the pilot via the control keyboard.

This page will make it much easier for the pilot to navigate an established course between two points. Rather than having to draw a mental picture of the plane's position relative to known navigational fixes, this position will be shown graphically with all relative navigational fixes displayed.

The programming for the NAV PAGE is broken down into two basic parts as in the DATA PAGE. The first part consists of the static information for the page. The second part consists of the dynamic information. Remember, the static information is loaded into the vector memory only once when the EHSI system switch is turned "on". When the NAV PAGE is the present page being displayed, the dynamic information is generated and loaded into the vector memory every time a clock interrupt is received from the DACI.

7.1 Static Portion

The static portion of the NAV PAGE can be broken down into four areas. The first area consists of text strings. These strings provide the headings for the information being displayed on the NAV PAGE. All the plot locations for the strings are in an include file entitled "navpg_xy.h". This is the same type of include file that was used in the DATA PAGE for its plot locations. All the strings are generated using the (1.5) character size of the built-in character set. String_gen was used to generate the VGD commands to plot the strings. The text strings which are presently shown on the NAV PAGE are the following:

- 1) AIRSPEED: calculated airspeed
- 2) GROUNDSPEED: speed relative to ground
- 3) TRACK: ground track of airplane

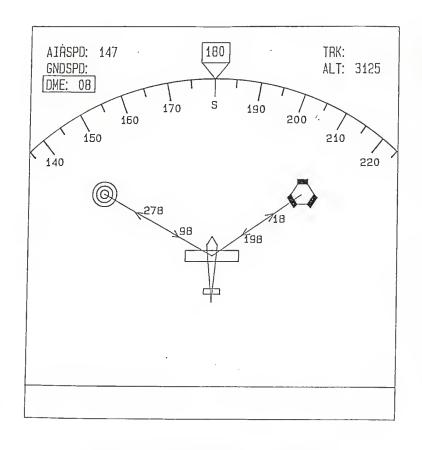


Figure 7.1 PRESENT NAVIGATIONAL PAGE

- 4) ALTITUDE: present altitude
- 5) DME: distance information to VORTAC or localizer

The second area of the static portion is the plane graphic. The plot location for the plane is also contained in the "navpg_xy.h" file. It is plotted in the center of the NAV PAGE and is generated using the plane routine. All VOR's and NDB's are plotted relative to this plane graphic. Therefore, this graphic represents the position of the plane relative to the navigational fixes being plotted.

The arc of the analog compass generated by the NAV PAGE is the third area of this portion. This arc has an angle of ninety degrees and is plotted at the top of the NAV PAGE. Its plot location is contained in the plot coordinate include file. All the static text strings generated are displayed above this arc.

The fourth area of the static portion of the NAV PAGE is the heading box and pointer. This area will contain the digital heading of the airplane and will provide a pointer to the plane's heading on the compass. It is plotted above the compass arc in the middle of the display. Its plot coordinates are in the plot coordinate include file.

7.2 Dynamic Portion

The dynamic portion of the NAV PAGE is much more complicated than that of the DATA PAGE. It is responsible for

calculating and plotting the flight status information mentioned and generating the graphical representation of the plane's position relative to the known navigational fixes. The latter of these is the more complicated part of this portion. It involves assimilating all the flight information about the known navigational fixes collected by the DACI and generating the VGD commands to plot this assimilated data in a usable graphical representation. All plot coordinates for the vector graphics generated in this portion are also contained in the "navpg_xy.h" include file. All strings are plotted using string_gen and are zero padded using zero_pad. The dynamic portion of the NAV PAGE consists of the following:

- AIRSPEED: The airspeed of the plane is calculated using the airspeed routine. It is then zero padded and plotted. Alarm conditions are also monitored by this routine for the stall airspeed.
- GROUNDSPEED: The groundspeed has yet to be implemented.
- 3) TRACK: The track has yet to be implemented.
- ALTITUDE: The present altitude is calculated using the altitude routine. It is then zero padded and plotted. Alarm conditions are monitored for both assigned altitude and MDA/DH.
- DME: The DME value is calculated using the dme routine. It is then zero padded and plotted.

6) HEADING

- A) The heading is calculated using the get_heading routine. It is then zero padded and plotted in the heading box.
- B) The compass routine is then used to generate

the dynamic portion of the compass. The heading is passed to the routine. This routine then generates the vector graphic commands to put the hash marks and markings for the heading sent. The heading of the plane is centered at the base of the heading pointer below the heading box.

7) VORTAC

- A) A VORTAC is displayed if it is within 20 nautical miles.
- B) Position of VORTAC relative to plane is calculated.
- C) VORTAC's angle relative to plane is calculated.
- D) Heading information is added into this angle and the angle range is checked.
- E) Distance to VORTAC is scaled.
- F) VORTAC is plotted using the vortac routine.
- G) If the plot distance of the VORTAC is greater than a certain limit the heading "to" and the bearing "from" the VORTAC are generated.
 - Heading_and_bearing is called to calculate the heading and bearing.
 - The heading and bearing are plotted with appropriate arrows.

8) NDR

- A) The angle of the NDB relative to the plane is calculated using the get_ndb_plot_angle routine.
- B) The NDB is then plotted at a constant distance from the plane graphic using the ndb routine.
- C) Heading_and_bearing is called to calculate the heading and bearing.
- D) The heading and bearing are plotted with arrows.

7.3 Future Development

The present status of the NAV PAGE leaves allot of room for improvement. There are many things which could be added to the page and several problems which must be overcome.

The first thing which must be added to the page is the use of waypoints. These waypoints will provide the pilot with intermediate navigational fixes. A routine must be written which allow the pilot to enter the waypoints. The dynamic portion of the NAV PAGE will also need to be modified to incorporate the calculation of the waypoints position relative to the airplane.

Second, the ground track will need to be implemented. This will involve writing a routine to calculate the ground track. Another routine will also be required to plot the ground track arrow above the compass arc.

Third, a wind speed vector will need to be added. This will involve using basic vector mechanics and the available flight information provided by the DACI after it has been converted.

Fourth, an additional vortac will have to be implemented. This will be necessary because most small planes can have two VORTAC's tuned in at once. This addition will involve slight modifications of the dynamic portion of the NAV PAGE to incorporate the navigational frequencies entered by the pilot. These frequencies will be used by the

routine, along with the information being used presently, to calculate the position of each of the VORTAC's relative to the plane.

Fifth, a data base should be added. This data base could contain the navigational coordinates for VORTACS's, NDB's, area airports, cities, towns, and other known landmarks. Such information would be invaluable when calculating the position of waypoints. It would also make it possible to show the position of the plane relative to navigational fixes in the data base.

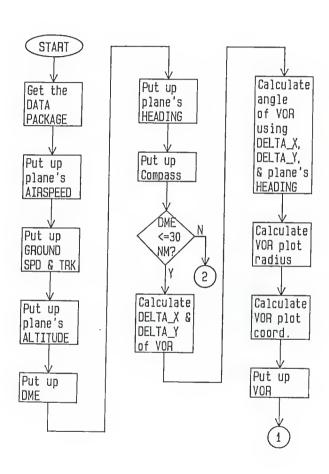
Sixth, a way of calculating the distance to an NDB would be convenient. If a data base existed this calculation would be easy. Without a data base, however, another method must be found to calculate this distance. A possible solution is found in Lagerberg's thesis[2].

The first problem which must be overcome is that of processing speed. Presently, the dynamic portion of the NAV PAGE takes the most time of all the routines. However, this may not prove to be true when the ILS page is developed further. A way must be found to speed up the VGD update as close to real time as possible. This will probably involve moving to a faster processor and coprocessor along with a decoded and latched keyboard.

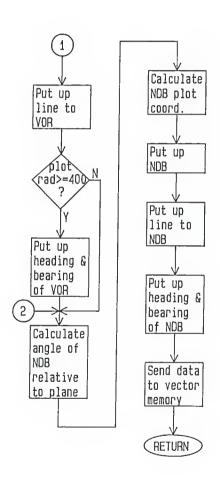
Second, there is a problem the positional information for the VORTAC's obtained from the flight simulator. The

positional information retrieved from the flight simulator is the form of rectangular coordinates. Thus, the position of the plane can viewed as the origin of a cartesian coordinate system with the VORTAC located at some specific coordinate in that system. Positional information to a VORTAC in a real plane is in the form of radial information along with DME information. Therefore, minor changes will have to be made in the dynamic portion of the software to calculate the position of VORTAC's relative to the plane.

7.4 High Level Flowchart



7.4 High Level Flowchart (cont.)



CHAPTER EIGHT

ILS PAGE

The ILS PAGE provides a graphical representation of an ILS approach to an airfield. This is something which will be of great interest to the small plane pilot who has an IFR rating or who is in the process of obtaining one. A large amount of information is assimilated from the flight instruments on this page. This assimilated information is presented in a format which will show the plane's position relative to the approach path. This will make it easier for a pilot to stay on the approach path. The pilot will no longer have to rely on a mental picture of where the plane is located relative to the approach path. At present, a very simple graphical representation of an ILS approach is being generated. However, future development of this page will result in a very complex graphical representation which will make it even easier for the pilot to fly an approach. The algorithm for the future ILS PAGE will

much more complicated and involve a larger amount of calculations than the present algorithm. The ILS PAGE will then be the most complicated of the three pages of the EHSI.

Most of the information on this page is in an analog type format. An example of the present version of the ILS PAGE can be seen in Fig. 8.1. The information presented on this page is the following:

1) HEADING:

- A) The heading is displayed in analog type compass format showing a 10 degree view either side of the heading.
- B) A digital value for the heading is displayed at the compass heading.

2) ALTITUDE:

- A) PRESENT: The present altitude of the plane is displayed in a digital format.
- B) MDA/DH: The MDA/DH is displayed in a digital format.
- 3) DME: The DME information to a localizer or VORTAC is displayed in a digital format.
- 4) APPROACH STATUS: This is in an analog format.
 - A) Plane's position relative to glideslope is displayed.
 - B) Plane's position relative to localizer is displayed.
 - C) Scaled runway is shown.
 - D) Heading of runway relative to the plane's heading is shown.

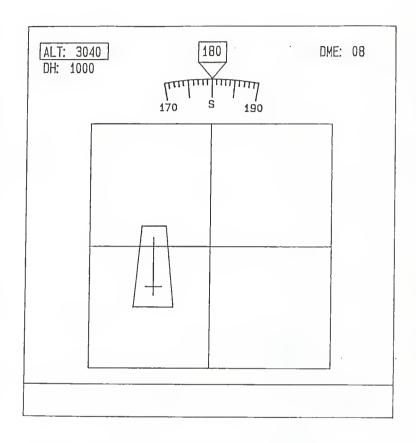


Figure 8.1 PRESENT INSTRUMENT LANDING SYSTEM (ILS) PAGE

- 5) WARNING MESSAGE AREA: Warning messages to the pilot are displayed in this area. These messages are generated by various routines.
- 6) COMMAND LINE AREA: This area is for displaying numeric values entered by the pilot via the control keyboard.

The programming for the ILS PAGE is also broken up into two basic parts just like the DATA and NAV pages. The first part generates the static information for the page. The second part generates the dynamic information for the page. Once again, the static information is loaded into the vector memory only one time. When the ILS PAGE is the present page being displayed, the dynamic information is generated and loaded into the vector memory every time a clock interrupt is received from the DACI.

8.1 Static Portion

The static portion of the ILS PAGE consists of three separate areas. The first area is comprised of text These strings provide headings for the flight status information being displayed on the page. This page also has an include file of the same type used in the DATA and NAV pages which contains the plot coordinates for the This file is entitled "ilspg_xy.h". strings. the strings are generated using the (1.5) character size of HP 1345A internal character set. String gen is used to plot the strings. The text strings presently being displayed on

the ILS PAGE are the following:

- 1) ALTITUDE: present altitude
- 2) MDA/DH: minimum decision altitude or decision height
- 3) DME: distance information to VORTAC or localizer

The second area of this portion of the page consists of the arc of the analog compass. This is a twenty degree arc which is plotted at the middle of the top part of the page. It is generated with the arc_circ routine and its plot coordinates are contained in the coordinate include file.

The third area of the static portion of the ILS PAGE consists of the heading box and pointer. This area will contain the digital heading of the airplane and will provide a pointer to the plane's heading on the twenty degree portion of the analog compass. Its plot coordinates are also in the coordinate include file. It is plotted above the compass arc in the middle of the display.

8.2 Dynamic Portion

The dynamic portion of the ILS PAGE generates the graphical representation of the plane's position relative to the approach path. It also plots the flight status information which was mentioned above. To generate the representation of the approach, data is assimilated from the localizer, glidepath, heading, and DME. As was stated pre-

viously, the present version is a very simple. This is the second most complicated page of the EHSI. All plot coordinates for the vector graphics generated by this portion are in the "ilspg_xy.h" include file. All strings are plotted using string_gen and are zero padded using the zero_pad routine. The dynamic portion of the ILS PAGE consists of the following:

- ALTITUDE: The present altitude is calculated using the altitude routine. It is then zero padded and plotted. Alarm conditions are monitored for both assigned altitude and MDA/DH.
- 2) MDA/DH: The MDA/DH is retrieved from the clock package, zero padded, and plotted. This is inputed by the pilot.

3) HEADING

- A) The heading is calculated using the get_heading routine. It is then zero padded and plotted inside the heading box.
- B) The ils_cmps routine is then used to generate the dynamic portion of the compass. Remember, this compass only shows a twenty degree portion around the heading. The heading of the plane is centered at the base of the heading pointer below the heading box.
- DME: The DME value is calculated using the dme routine. It is then zero padded and plotted.

5) APPROACH STATUS

- A) If plane is within 20 nautical miles, the following procedure is executed.
 - 1. The runway scale is calculated.
 - The plot boundaries are calculated based on the runway scale.

- 3. The crosshairs are plotted.
- If the plane is within 15 nautical miles, the following procedure is executed.
 - a. The approach sensitivity is calculated based on the DME value.
 - The glideslope is calculated using the approach sensitivity and boundary conditions.
 - c. The course deviation from the localizer is calculated using the approach sensitivity and boundary conditions.
 - d. The crab angle is calculated using the plane heading and the runway heading.
 - e. The runway is then plotted based on the glideslope, course deviation, and runway scale.

8.3 Future Developments

There are many improvements which could be made to the present version of the ILS PAGE. Eventually, these improvements will probably result in a graphical representation of an ILS approach which is only vaguely similar to the present version. An example of what the future ILS PAGE might look like appears in Fig. 8.2 and Fig. 8.3[2].

The later versions of software which generate the ILS PAGE will assimilate a larger amount of data from the flight instruments than the present version of the page. In the final version of the EHSI, the ILS PAGE will become the most complicated of the three pages available on the system.

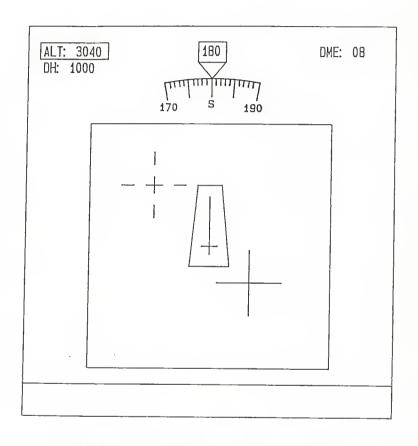


Figure 8.2 ILS PAGE USING TWO CROSSHAIRS

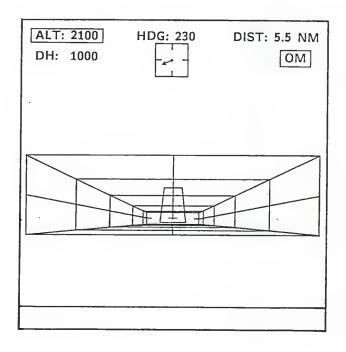


Figure 8.3 ILS PAGE WITH APPROACH TUNNEL [2]

The first thing which might be added to the page is the addition of tendency information. This information will be very important to the pilot. It will give immediate feedback about whether or not a maneuver being made is going to get the plane back on the approach path. Digital will have to be written to accomplish this task. Gruenbacher discusses the development of such filters in thesis [3]. An example of tendency representation can be seen in the box with the arrow at the top of the display in Fig. 8.3[2]. The arrow represents the plane's tendency to move off the glideslope and localizer. If the arrow is pointing up and to the right, the plane's tendency is to move up and away from the approach path. Another example of representing the tendency information can be seen in Fig. The dotted crosshairs and solid crosshairs are used to indicate the plane's tendency. When both the crosshairs are on top of one another the plane is on the approach path.

Second, an ILS approach tunnel might be added. This tunnel would display the boundaries of the approach path. The outer and middle marker information would be represented by the beginning and end of the tunnel. The plane's position on the approach path would be indicated by the graphical representation of the tunnel on the VGD. An example of the ILS tunnel can be seen in Fig. 8.3[2].

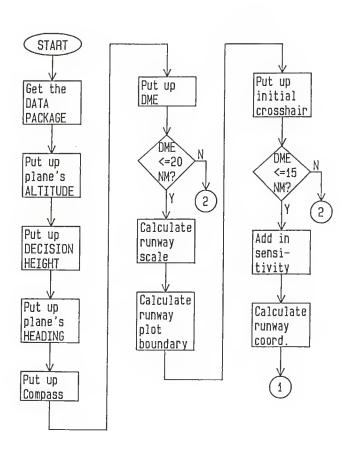
Third, outer and middle marker information needs to be

incorporated into the ILS page. This was mentioned in the previous paragraph. This information is important because it lets the pilot know how far he is into an approach. There are many possible ways the marker information might be presented.

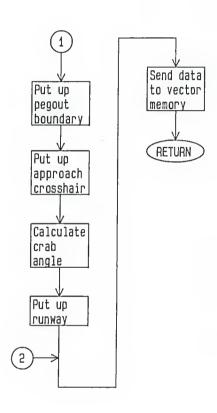
Fourth, the addition of a data base like the one discussed in the future developments of the NAV PAGE would be of great importance. This data base could contain information for each airport such as the runway heading, outer marker distance, middle marker distance, MDA/DH, and field altitude. This information would be invaluable in generating any version of the ILS PAGE.

Fifth, reverse course information would be a convenient addition. This would allow the pilot to fly outbound on an ILS approach path without having to change his reactions to the graphical representation of the approach. This would be especially nice on the present version of the ILS PAGE. If an ILS tunnel was being used, this information would not would not be necessary.

8.4 High-level Flowchart



8.4 High-level Flowchart (cont.)



CHAPTER NINE

CONCLUSION

9.1 Results

The present version of the EHSI is still basically in its infancy. However, a large amount of progress has been made since the initial proposal by Dyer in 1982[1]. An EHSI Development System now exists, so that the different pages of the EHSI can be developed. This development system discussed briefly in Chapter 1 and in detail in [2]. Usable communication routines have been developed to allow the host computer to communicate with the DACI of the development These routines are discussed in Chapter 1 and in system. detail in [3]. The initial software to generate the graphics for the three pages of the EHSI has been developed. This software is written in the C programming language and is discussed Chapters 2, 6, 7, and 8. The code for the necessary routines can be seen in Appendix E. The DATA PAGE and the NAVIGATIONAL PAGE are the pages which have seen the

most progress. The ILS PAGE is going to require much more development.

The present working version of the EHSI gives confidence that the goal of providing an affordable EHSI for the general-aviation public can be reached. However, the final version of the EHSI will take several more personyears of development.

9.2 Future Development

Along with the future development of the NAVIGATIONAL PAGE and the ILS PAGE discussed in Chapters 7 and 8, there are several other areas which must be developed. First, EEPROM's must be programmed with the DACI control program. This will include the code in [2] and the additions discussed in Appendix A.

Second, the possibility of developing a dedicated graphics processor should be considered. This would take the burden of generating the VGD commands to produce the various graphics of the EHSI pages off the host. The host could then be used for evaluating DACI interrupts, making decisions, and processing data retrieved by the DACI.

Third, a faster processor and coprocessor are needed on the host side of the development system. This could be accomplished by upgrading the present host to a faster PC with a faster numerical coprocessor. Or, a dedicated host processor could be designed to handle the host-side responsibilities of the development. This is probably the most logical way to proceed after the pages have been fully developed because the final EHSI system will have to contain such a dedicated processor.

Fourth, a data base should be incorporated into the EHSI system. This would be valuable in the development of all the pages of the EHSI. The benefits of data base are discussed in the future development sections of Chapters 7 and 8. Such a data base will have to be implemented in the final version of the EHSI to make it a highly marketable product.

9.3 Another Application

As was stated, the goal of our EHSI is to provide an affordable EHSI for the general-aviation public. This EHSI will ease the burden of the single pilot flying under IFR conditions. The EHSI will be designed for use in the small airplane. However, this will not be its only application. The EHSI would be invaluable in a teaching environment. It could be used to help a pilot obtaining an IFR rating to better understand what the instruments are telling him. This will be especially true on the NAVIGATIONAL and ILS PAGES of the EHSI. Oftentimes it is difficult to draw a mental picture of the plane's position relative to a navi-

gational fix or to get immediate feedback as to whether or not a maneuver will get the plane back onto a selected course or ILS approach path. The EHSI will provide this information for the pilot and speed up the learning process.

REFERENCES

[1]	Dyer, S.A., "A Proposed Electronic Horizontal
	Situation Indicator for use in General-Aviation
	Aircraft, " Proceedings of 1982 Position, Location
	and Navigation Symposium, pp. 198-205.

- [2] Lagerberg, J.D., <u>An Electronic Horizontal Situation Indicator and Development System</u>, Kansas State University, 1987.
- [3] Gruenbacher, D.J., <u>Low-level Software for an EHSI</u>
 <u>Development System</u>, Kansas State University, 1987.
- [4] Hewlett-Packard Inc., HP-1345A Digital Display Module Designers Manual, 1981.
- [5] UnderWare Inc., <u>BRIEF Quick Reference Card Version</u>
 1.33, 1984.

APPENDIX A

CHANGES TO THE DACT

First, dedicated interrupt lines were added to the DACI. This involved both hardware and software modifications to the DACI. This addition greatly improved the reliability of the system. Prior to this addition, noise on the communications bus was causing the DACI to interpret noise spikes as interrupt requests from the host. Since there was no interface command on the bus at the time of the interrupt, the interrupt was determined to be an illegal interface command and an error message resulted from the DACI.

The hardware changes to implement the dedicated interrupt lines can be seen in Fig. A.1[2] and Fig. A.2[2]. These dedicated lines are called IRQOUT and IRQIN. The IRQOUT line is used when the DACI wishes to interrupt the host. The IRQIN line is used when the DACI is being interrupted by the host. In Fig. A.1, the original OUTSHAKE line

of the DACI was moved from pin 19 of the EHSI parallel port connector to pin 25 of the parallel port connector. The dedicated IRQOUT line was then connected to pin 19 of the parallel port. The IRQOUT line originates from pin 14 of the control PIA in Fig. A.2. The SPAREIRQ line was originally connected to the control PIA at pin 14 and was not being used. The dedicated IRQIN line was implemented by cutting the original INSHAKE line going to pin 5 of U13 in Fig. A.2 and adding a 4.7k ohm resistor as a pullup. This IRQIN line is connected at pin 6 of the EHSI parallel port connector as seen in Fig. A.1. Pins 4, 23, 21, 26 of the parallel port were also grounded.

The software additions for the dedicated interrupt scheme were very simple. Only one additional routine was needed and it is entitled OUTSHAKE_IRQ. This routine is presently located at \$28DC in the RAM of the DACI. The code for OUTSHAKE_IRQ is seen in Fig. A.3. Additional changes to other DACI routines were also necessary to incorporate the OUTSHAKE_IRQ routine. Copies of the original DACI routines from [2] are at the end of this appendix. The changes made are noted in these routines. Note, PB4 of the control PIA in Fig. 2.A is configured as an output for the OUTIRQ line.

Second, a switch was added to the DACI to allow the input of an external clock to the clock interrupt line of the DACI. This was added to analyze the speed of the EHSI

system and to demonstrate the operation of the interrupt stack on the host-side. It also made it possible to retrieve data at specific frequencies for the design of digital filters for various portions of the EHSI. The hardware modifications that were made are seen in Fig. A.4. The CLKIRQ line was cut and a two-pole switch was added. One pole of the switch is connects the CLKIRQ line to the real-time clock interrupt of the system. The other pole connects the CLKIRQ line to an external clock input.

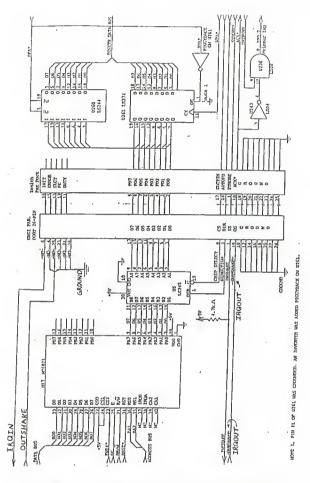


Figure A.1 DACI TO HOST PARALLEL PORT [2]
ADDITION OF IRQIN AND IRQUUT

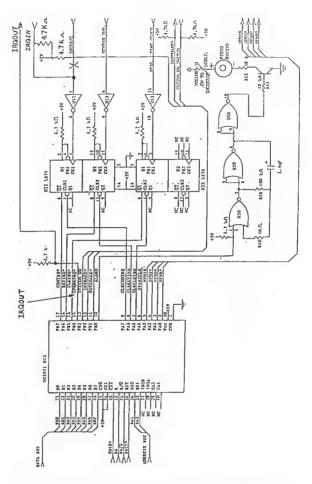


Figure A.2 CONTROL PIA AND ALARM HORN [2]
ADDITION OF IRQIN AND IRQOUT

SUBROUTINE OUTSHAKE_IRQ PURPOSE: To send out a pulse on the IRQOUT line of the parallel port. INITIAL CONDITIONS: This subroutine requires no parameter to be passed to it. ACTION: The pulse on the IRQOUT line has a pulse width of approximately 25 microseconds. The actual pulse width can be varied according to the value in DO.L. The expression for the pulse width is 3 * (DO + 1) + 4 microseconds. REGISTER USAGE: No registers are affected. ORG. \$208DC MOVE. L DO. \$9C8 SAVE REGISTER DO MOVE. L #6, DO SET PULSE LOW WIDTH BRING OUTTRO LOW

MOVE.L #6, DO

MOVE.L #6, DO

BCLR #4, \$30005

SELF5 DBRA DO, SELF5

BSET #4, \$30005

MOVE.L \$98C, DO

RTS

HOLD OUTIRG LOW
TAKE OUTIRG HIGH
RESTORE REGISTER DO
RETURN TO CALLING ROUTINE

Figure A.3 OUTSHAKE_IRQ ROUTINE

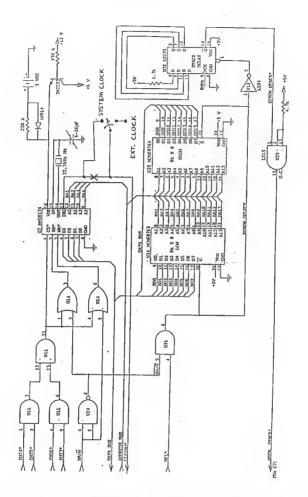


Figure A.4 EEPROM'S AND REAL TIME CLOCK [2] ADDITION OF EXTERNAL CLOCK SWITCH

APPENDIX A (CONTINUED)

CHANGES TO DACI ROUTINES FROM [2] NOTED

REVISION	JSR MOVE. B 3/3/87	INTPIA #*13, *30005 ② ① ENABLES PB4	
PIA#1PB	MOVE. L MOVE. B	#\$30005, AO #\$13, DO	INIT PIA #1 PORT B I/P O/P
PIA#1PA	MOVE.L MOVE.B JSR MOVE.B	#\$30001, AO #\$FF, DO INTPIA #\$0F, \$30001	INIT PIA #1 PORT A OUTPUT INIT CONTROL BITS
	DBRA MOVE.L MOVE.L MOVE.B TRAP	D3, SP1	
SP1	MOVE.B TRAP MOVE.L MOVE.B MOVE.B TRAP	#248,D7 #14 #30,D3 #\$20,D0 #248,D7 #14	OUTPUT CHARACTER HANDLER SEND THIRTY SPACES TO THE TERMINAL OUTPUT CHARACTER HANDLER
START LF1	ORG. MOVE.L MOVE.B MOVE.B TRAP DBRA MOVE.B	\$20900 #8,D3 #\$A,D0 #248,D7 #14 D3,LF1 #\$D,D0	SEND EIGHT LINE FEEDS TO THE TERMINAL OUTPUT CHARACTER HANDLER SEND CARRIAGE RETURN

LOGO	CLR.L MOVE.L MOVE.W JSR MOVE.W	D1 #\$3A9O, AO (AO)+, D1 SNDLST #\$COOO, DO	SET UP LOGO FOR DISPLAY ON THE HP DISPLAY SEND THE LOGO TO VGD SET UP ADDRESS POINTER
SYSON	BTST BEQ JSR MOVE.W JSR	#3, \$30005 SYSON SNDCMD #0, DO SNDCMD	WAIT FOR SYSTEM ON LOW OFF, HIGH ON SET UP ADDRESS PTR SO SEND A NOP WHERE JUMP WAS SO LOGO CAN BE
SYSUP	MOVE. B JSR MOVE. B JSR MOVE. L MOVE. B JSR	#\$65, \$30021 OUTSHAKE_IRQ (1) #\$2, D2 INSHAKE #\$30021, A0 #0, D0 INTPIA	DISPLAYED SEND SYS UP VEC TO HOST SEND STROBE TO HOST SET ERROR FLAG TO ACKERR WAIT FOR ACKNOWLEDGE MAKE PARALLEL PORT INPUT
ENIRGS	OR. B	#\$F0,\$30001	ENABLE ALL INTERRUPTS
ENCLKIRO	MOVE. B	#\$OC,\$3007F	ENABLE INT INTERVAL
MAIN	MOVE.B JSR NOP NOP MOVE.B BTST BNE	#24, \$1FA0 ANALOG \$30019, \$1FB1 #3, \$30005 MAIN	PUT # OF ENT IN DATA PAK PUT ANALOG VAL IN PACK NOP'S ADDED FOF CALL TO MODIFIED CLOCK_IRQ PUT BINARY I/P'S IN PACK CHECK TO SEE IF SYS IS ON IF ON, GO THRU MAIN AGAIN
SYSDN	AND. B MOVE. B BSET	#\$OF,\$30001 #\$2700,SR #0,\$30005	DISABLES INTERRUPTS DISABLES ALL INT LEVELS TURN ALARM OFF IF ON
SHUTN	MOVE. L CLR. L MOVE. W JSR MOVE. L MOVE. B JSR MOVE. B JSR MOVE. B JSR	#\$FF, DO INTPIA #\$66, (AO) OUTSHAKE_IRQ () #2, D2	DISPLAY SYS SHUTDOWN MSG MAKE PARALLEL PORT O/P SEND SHUTDN VEC TO HOST STROBE THE HOST SET ERROR FLAG TO ACKERR WAIT FOR ACKNOWLEDGE
REVISION	3/3/87	O CALLS TO OUTS	

KEYIRQ	ORG. MOVEM.L JSR CMPI.B BEQ CMPI.B BNE JSR BNA MOVE.L MOVE.B JSR MOVE.B JSR MOVE.B JSR MOVE.B JSR MOVE.B JSR	GETKEY #37, D2 NOKEY #36, D2 CONT5 SETCLK NOKEY #\$30021, A0 #\$FF, D0 INTPIA D2, \$30021 #2, D2	SAVE REGISTERS GET VAL OF KEY PRESSED WAS THERE KEY FOUND IF NO KEY THEN RETURN CHECK FOR SET CLOCK KEY IF NOT SET CLOCK KEY SET THE CLOCK/CALENDAR AFTER CLOCK IS SET, RET CHANGE PARALLEL PORT, O/P SEND VALID KEY, PAR PORT SET ERROR FLAG TO ACKERR STROBE THE HOST WAIT FOR ACKNOWLEDGE ENABLE PAR PORT TO INT'S CHANGE PAR PORT TO I/P
NOKEY	BCLR BSET MOVEM.L	#6,\$30001 #6,\$30001	CLEAR KEYBOARD INTERRUPT RESTORE REGISTERS
REVISION 3	RTS 3/ 3/87	CALL TO OUTSH	RETURN KAKE_IRQ ADDED

	MOVE. B AND. W MULU. W	\$3006F,D0 #\$F,D0 #10,D0	READ TENS OF HOURS CLEAR DATA4 - DATA7
	MOVE. B	\$3006D, D2	READ THE ONES OF HOURS
	AND. B	#\$F,D2 D2,D0	CLEAR DATA4 - DATA7
	MOVE. B	DO, - (AO)	ADD TENS AND ONES OF HRS
	MOVE. B	\$30073,DO	PUT IN THE DATA PACKAGE READ TENS OF DAYS
	AND. W	#\$F, DO	CLEAR DATA4 - DATA7
	MULU. W	#10. DO	OLEAR DATA4 - DATA/
	MOVE. B	\$30071,D2	READ THE ONES OF DAYS
	AND. B	#\$F, D2	CLEAR DATA4 - DATA7
	ADD. B	D2, D0	ADD TENS AND ONES OF DAYS
	MOVE. B	DO, -(AO)	PUT IN THE DATA PACKAGE
	MOVE. B	\$30075, DO	READ DAY OF WEEK
	AND. B	#\$F, DO	CLEAR DATA4 - DATA7
	MOVE. B	DO, -(AO)	PUT IN THE DATA PACKAGE
	MOVE. B	\$30079, DO	READ TENS OF MONTHS
	AND. W	#\$F, DO	CLEAR DATA4 - DATA7
	MULU.W	#10, D0	
	MOVE.B AND.B	\$30077, D2	READ THE ONES OF MTHS
	ADD. B	#\$F, D2 D2, D0	CLEAR DATA4 - DATA7
	MOVE. B	DO, -(AO)	ADD TENS AND ONES OF MTHS
	MOVE. B		PUT IN THE DATA PACKAGE READ DAY OF WEEK
(D)	MOVE. L	#\$30021, AO	MAKE PAR PORT O/P
	MOVE. B	#\$FF, DO	HARE FAR FORT U/P
	JSR	INTPIA	
	MOVE. B	#\$60,\$30021	SEND THE TIME VEC TO HOST
	JSR	OUTSHAKE IRQ (2	SEND DATA STROBE
	MOVE. B	#2, D2	SET ERROR FLAG TO ACKERR
	JSR	INSHAKE	WAIT FOR ACKNOWLEDGE
	BSET	#7,\$30001	ENABLE PAR PORT FOR INT'S
	MOVE. L	#\$30021, AO	CHANGE PAR PORT TO I/P
	MOVE. B	#0, D0	
FAST	JSR	INTPIA	
FASI	MOVE.B	\$3007F, DO	3 READS CLEARS CLOCK IRQ
	MOVE. B	\$3007F, DO	
	BCLR	\$3007F,D0 #5,\$30001	61.7.7.61.65
	BSET	#5,\$30001	CLEAR CLOCK INT LATCH
	MOVEM. L	\$9BC, DO/D2/AO	DECTOR DECTARDS
	RTS	4380, 807 827 A0	RESTORE REGISTERS RETURN
REVISION	3/3/87	A JUMP TO FA	ST CAN BE ADDED HERE FOR
		HIGH SPEED OF	PERATION WITH THE CLOCK
		INTERRUPT DI	SABLED. CLOCK VALUES
		BE INTERRUPT	IEVED AND HOST WILL NOT
		CALL TO OUTS	MAKE TEG ADDED
		9 3 10 30131	.uve_tva_KDDED

APPENDIX B

EHSI USER'S MANUAL

SYSTEM ON PROCEDURE

To bring the EHSI system to its operational state, complete the procedures of the following sections in order.

DACI ON

The procedure to start up the DACI contains the following steps:

- STEP 1) Make sure all the ribbon cables are connected to their appropriate ports on the DACI.
- STEP 2) Connect the green +5 volt cable to the +5 volt output of the power supply for the Motorola MC68000 Educational Board.
- Warning: Never turn off the power supply for the Educational Board or the DACI control program will be lost. This warning will be in affect until the EEPROM's are burned for the DACI control program.
- STEP 3) Turn on the power supply for the HP 1345A and

the DACI interface board.

- STEP 4) Turn on the Zenith Z-29 video terminal. This is the terminal connected to the Educational Board. The power switch is located on the back of the terminal at the bottom of the right-hand side.
- STEP 5) Press the "CAPS LOCK" key of the Z-29. This will place the Z-29 in the capital letter mode which is necessary for entering Educational Board commands.
- STEP 6) Press the black hardware reset button on the Educational Board.
- STEP 7) Press the red software abort button on the Educational Board.
- STEP 8) Make sure the EHSI system switch is in the "off" position. This switch is located above the control keyboard and should be in the down position.
- STEP 9) Run the DACI control program by entering the following command:

GO 2900

Then, hit the RETURN button on the Z-29. The following message should appear:

EHSI SYSTEM RUNNING INITIALIZATIONS COMPLETE

CONTINUE TO FLIGHT SIMULATOR

If any error messages occur, refer to [2] for actions to be taken.

This completes the necessary steps to turn on the DACI. Proceed to the next section.

HOST ON

Execute the following steps to turn on the host side of the EHSI system:

STEP 1) Turn on the Zenith Z-158 host computer. The power switch is located on the power strip on top of the lab bench to the right of Z-158.

This switch will also turn on the Z-158 monitor. The following prompt should be displayed:

C:\>

STEP 2) Enter the following command at the prompt: cd\ehsi

Then, hit the RETURN button on the Z-158. The following prompt should be displayed:

C:\ehsi>

STEP 3) Enter the following command at this prompt:

Press the RETURN button.

This completes the necessary steps to turn on host side of the EHSI. Proceed to the next section.

FLIGHT SIMULATOR ON

Execute the following steps to turn on the ATC-610 Flight Simulator:

- STEP 1) Get the flight simulator key and insert it into the ignition switch. Turn the key until the switch is in the "on" position.
- STEP 2) Push the master switch to the "on" position.

 this switch is located to the right of the ignition switch.
- STEP 3) For instructions on the use of the flight simulator proceed to the ATC-610 Flight Simulator User's Manual.

This completes the necessary steps to turn on the simulator. Proceed to the EHSI ON section.

EHSI ON

Turn the EHSI system switch located above the control keyboard to the "on" position. The switch will be in the up position. The following message will appear on the monitor of the Z-158:

SYSTEM SWITCH ON.

The DATA PAGE will now be displayed on the HP 1345A. The

EHSI is now fully operational. Proceed to the flight simulator.

This completes the SYSTEM ON PROCEDURE.

SYSTEM OFF PROCEDURE

To shut down the EHSI system, complete the procedures of the following sections in order.

DACI OFF

Turn the EHSI system switch located above the control keyboard to the "off" position. The switch will be in the up position. The following message will appear on the monitor of the Z-158 along with the DOS prompt:

SYSTEM SWITCH OFF.

C:\ehsi>

A shutdown message will also appear on the HP 1345A. Proceed to the next section.

FLIGHT SIMULATOR OFF

- STEP 1) Push the master switch to the "off" position.

 This switch is located to the right of the ignition switch.
- STEP 2) Turn the key in the ignition switch to the

"off" position. Remove the key and return it to the flight simulator log book.

This completes the necessary steps to turn off the flight simulator. Proceed to the next section.

HOST OFF

Turn off the Zenith Z-158 host computer. The power switch is located on the power strip on top of the lab bench to the right of the Z-158. The switch also turns off the Z-158's monitor. The host should now be in its off state. Proceed to the next section.

DACI OFF

Execute the following steps to turn off the DACI:

- STEP 1) Press the red software abort button on the Educational Board.
- STEP 2) Turn off the Zenith Z-29 video terminal.

 This is the terminal connected to the
 Educational Board. The power switch is
 located on the back of the terminal at the
 bottom of the right-hand side.
- STEP 3) Turn off the power supply for the HP 1345A and the DACI interface board.
- STEP 4) Disconnect the green +5 volt cable from the output of the power supply for the Motorola

MC68000 Educational Board.

- Warning: Never turn off the power supply for the Educational Board or the DACI control program will be lost. This warning will be in affect until the EEPROM's are burned for the DACI control program.
- STEP 5) Disconnect the ribbon cables going to the host, flight simulator, and control keyboard.

This completes the SYSTEM OFF PROCEDURE

APPENDIX C

EHSI HOST PROGRAM MAINTENANCE

All the host programs for the EHSI are located in the ehsi directory. To get to the ehsi directory, type in the following command at the prompt:

C: \>cd\ehs1

Then, press the RETURN key. The following prompt will appear on the monitor:

C:\ehsi>

The EHSI host programs can now be edited, compiled, and linked.

EDITING

All editing for the host programs of the EHSI is done using BRIEF, Version 1.33. The is a full-screen editor designed specifically for the editing of programs. To begin editing a file, type the following command at the prompt:

C:\ehsi>b XXXXXX.c

The program name to be edited should be inserted at XXXXXXX. Press the RETURN key to execute the editor. The name of the file being edited will appear at the top of the screen. To exit the editor, press the Alt key followed by the X key and answer the questions which appear in the command line at the bottom of the screen. The various editing commands available can be seen in Fig. C.1[5] and C.2[5]. This is the BRIEF Quick Reference Card. For a more detailed description of the commands refer to the BRIEF manual.

COMPILING AND LINKING WITHOUT USING MAKE UTILITY

COMPILING

The Microsoft C compiler, Version 4.0, is the compiler used for the EHSI. Compiling of the host programs for the EHSI can be done in two different ways. The first method is to compile the programs directly. The second method is to compile the programs from inside the BRIEF editor using a macro.

To compile the programs directly type in the following command at the prompt:

C:\ehsi>msc XXXXXX;

The program name to be compiled should be inserted at XXXXXXX.

Press the RETURN key to execute the compilation. The following message will be displayed:

Buffer Commands

Search/Translate Commands

Alt-b	Buffer list	Alt-c	Toggle case sensitivity
Alt-e	Edit file (create buffer)	F5, Alt-s	Search forward
Alt-f	Display buffer file name	F6, Alt-t	Translate
Alt-n	Edit next buffer	Shift-F5	Search backward
Ctrl-minus	Edit previous buffer	Shift-F6	Search again
Alt-r	Read file into buffer	Alt-F5	Incremental search forward
Alt-o	Change output file name	Alt-F6	Toggle regular expressions
Alt-w	Write buffer		08 1-8 orat exhictments

1

Regular Expressions/Wildcards

Cursor Commands

Alt-minua

		•	
		?	
Down arrow	Down one line	e	
Left arrow	Left one column		
Right arrow	Right one column	•	
Up arrow	Up one line	}	
PgDn	Page down	\ <num></num>	
PgUp	Page up	,	
Home	Top of window	1	
End	End of window	ī~	
Ctr1-PgDn	End of line	i	
Ctrl-PgUp	Beginning of line	-	
Ctrl-Home	Top of buffer	1	
Ctrl-End	End of buffer		
Alt-g	Goto line	\n	
Ctrl-Left arrow	Previous word	\t	
Ctrl-Right arrow	Next word	<, %	
Ctrl-b	Line to bottom of window	>, \$	
Ctrl-c	Center line in window	\c	
Ctrl-d	Scroll down one line		
Ctrl-t	Line to top of window		
Ctrl-u	Scroll up one line		
Shift-PgUp	Left edge of window		

Right edge of window

Delete current buffer

Zero or more of any characters Any one character Zero or more of previous character or group Begin group End group Substitute text matched by <num>th group Begin inclusive character group Begin exclusive character group End character group Define range in character group Either the preceding or the following character or group Newline character Tab character Beginning of line End of line (excluding newline) Place cursor at specified character

Basic Text Commands

Shift-PgDn

Backspace	Backspace and delete/overwrite
De1	Delete character
Enter	Insert new line/Move to next lin
Tab	Insert tab/Next tab stop

Figure C.1 BRIEF Quick Reference Card [5]

Alt-d Delete line Macro Commands
Alt-k Delete/Kill to end of line
Ctrl-Enter Open new line after current line F7

F8

F9

FIO

Shift-F9

Record/Stop recording

Play back recording

Load macro file

Execute macro

Delete macro file

Block Text Commands

Window Commands

F1 Change window
F2 Reize window
D3 Create window
P4 Destroy window
P5hift-Down arrow
Shift-Left arrow
Shift-Right arrow
Shift-Right arrow
Change to right window
Shift-Op arrow
Change to window above

Miscellaneous Commands

Eac

Alt-h Help Alt-1 Toggle insert mode Alt-u Undo Undo Gray/Kaypad * Alt-v Display version ID Alt-x Exit Alt-E Suspend session Ctrl-Braak Halt search or macro Ctrl-w Toggle backup files Shift-Flo Insert key code in buffer Alt-F10 Compile program in buffer Ctrl-n Locate next error Ctrl-p Display previous error text Ctrl-r Repeat following command

Escape/Cancel

Figure C. 2 BRIEF Quick Reference Card [5]

126

Microsoft (R) C Compiler Version 4.00 Copyright (C) Microsoft Corp 1984, 1985, 1986. All rights reserved.

If there are any errors or warnings, the number of each respectively will also be displayed. The XXXXXX.ERR can then be viewed to note what error or warning occurred as well as the location of the error or warning. The error file can be viewed inside the BRIEF editor or by executing the type DOS command.

To compile EHSI program inside the BRIEF editor, up the file to be compiled as was discussed in the EDITING section above. Execute the compiler macro by pressing the Alt key followed by the FiO key. The program will then The command line at the bottom will show begin compiling. what compiler is being used and the options that are select-When compilation is complete, the location of the first error or warning will be indicated by the cursor. The command line will display the nature of the error or warning. To move to the next error or warning, press the Ctrl key followed by the N key. When all the error and warning messages have been displayed, the following message will appear in the command line:

No more errore

To execute the error handler again, repeat the key sequence for looking at the next error or warning.

Both of the key sequences for compiling and viewing the

errors and warnings are under the Miscellaneous Commands of the quick reference card in Fig. C.1[5].

WARNING

When the Make Utility is not being used to compile and edit, special care will have to be taken when making modifications to the clock_pkg and alarm_pkg data structures of the host EHSI program. These structures are in a file entitled data_str.h which is included in the following routines:

- 1) insert_new_freq
- 2) set_timer
- dat_pg_dynamic
- 4) nav_pg_dynamic
- 5) ils_pg_dynamic
 - 6) altitude
 - 7) dme
 - 8) airspeed
 - 9) timer
- 10) climb_rate
- 11) set_estimated_wind
- 12) set_altitude
- 13) ehsi

These routines will have to be recompiled and added to the library whenever a change is made to these data structures.

If this procedure is not followed, the results of the host EHSI program will be unpredictable.

LIBRARY

All the routines used by the host EHSI program are contained in a library which is used when linking the EHSI routine. Linking the EHSI routine will be discussed in the next section. The libraries for the host EHSI program are called ehsi.lib and key_ehsi.lib. When a new compiled routine is ready to be added to the library, type in the following command at the prompt:

C:\ehsi>lib ehsi+XXXXXX

Insert the name of the routine to be added at XXXXXXX. Press the RETURN key to begin installation of the routine into the library. The following will then appear:

Microsoft (R) Library Manager Version 3.04 Copyright (C) Microsoft Corp 1983, 1984, 1985, 1986 All rights reserved.

List File:

Enter cross after the List file: prompt and hit the RETURN key. This will provide a cross listing of all the files presently contained in the library.

To add a new version of a routine which already exists in the library, type in the following at the prompt:

C:\ehsi>lib ehsi-XXXXXX+YYYYYY

Insert the name of the old version of the routine contained

in the library at XXXXXX. Insert the name of the new version of the routine at YYYYYY. Usually, XXXXXX and YYYYYY will be the same. Then, press the RETURN key and proceed as before.

LINKING

To produce the executable file for the host EHSI program, the EHSI routine must be linked to the other routines of the EHSI. This is done by typing in the following command at the prompt:

C:\ehsi>link ehsi XXXXXX/stack:4000

Then, press the RETURN key. Insert any new routine being tested at XXXXXX. Note, more than one routine can be linked with the EHSI routine. Put a space between any additional routines being tested. The stack option is used to increase the stack size allotted for the program. The following message will now appear:

Microsoft (R) Overlay Linker Version 3.51 Copyright (C) Microsoft Corp 1983,1984,1985,1986 All rights reserved.

Run File [EHSI.EXE]: List File [NUL.MAP]: Libraries [.LIB]:

Press the RETURN key after the Run File and List File prompts. For an explanation of these, refer to the linker portion of the manuals. Enter ehsi and key_ehsi after the Libraries prompt and press the RETURN key. The new routines

being tested and the routines in ehsi.lib and key_ehsi.lib will now be linked to the EHSI routine. This will produce an executable file. Any error messages will be displayed as the linking process progresses.

COMPILING AND LINKING WITH THE MAKE UTITILITY

Compiling and linking of the EHSI host programs can be done in one step with the make utility. The make file for the EHSI can be seen at the end of the appendix and is entitled ehsi in the ehsi directory. This file contains the directory names of the EHSI host programs and the dependencies for each of the programs. The file also contains the the link list for the EHSI program and other necessary parameters.

The make utility recompiles any routine whose source file is never than the latest object file for the routine. Routines which are also dependent on certain include files will also be recompiled when the include file is never than the latest object file for the routine. After compiling all the routines specified in the make file, the EHSI program is linked with the link list specified in the make file along with the listed parameters.

To execute the make utility available, enter the following command at the prompt:

C:\ehsi>make ehsi

Press the RETURN key to execute the compilation and linking of EHSI programs into an exectuable EHSI file. If no errors occur while the make utility progresses through the compilation and linking process, a new executable file will be produced in the directory entitled ehsi.exe. If there are any errors or warnings, the number of each respectively will be displayed. The XXXXXX.ERR can then be viewed to note what error or warning occurred as well as the location of the error or warning. The error file can be viewed insided the BRIEF editor or by executing the type DOS command.

Note, when any new routines are added to the EHSI program, changes must be made to the ehsi make file. Refer to the programming user's manual for formats when making these changes to the make file.

warning, when compiling programs which are added to the ehsi.lib and key_ehsi.lib without using the make utility, make sure to erase the latest object file generated for the source file being compiled. If this is not done, the latest version will not be added to the appropriate library when the make utility is run and the latest version will not be linked with the EHSI program.

APPENDIX C (CONT.)

EHSI MAKE FILE

File: ehsi

SOURCE FILE: ehsi

FUNCTION: None.

DESCRIPTION: Make file for the EHSI host program.

Used when running make utility.

DOCUMENTATION

FILES: None.

ARGUMENTS: None.

RETURN: None.

FUNCTIONS

CALLED: None.

AUTHOR: DAVE GRUENBACHER AND CHUCK ROBERTSON

DATE CREATED: 10Apr87 Version 1.0

REVISIONS: None.

insert.cbj : insert.c
 msc insert;
 lib ehsi-+insert,ehsi.crs;

time_gen.obj : time_gen.c
 msc time_gen;
 lib ehsi-+time_gen,ehsi.crs;

line.obj : line.c
 msc line;
 lib ehsi-+line,ehsi.crs;

clim_box.obj : clim_box.c
 msc clim_box;
 lib ehsi-+clim_box,ehsi.crs;

clim_hsh.obj : clim_hsh.c
 msc clim_hsh;
 lib ehsi-+clim_hsh,ehsi.crs;

climrate.obj : climrate.c
 msc climrate;
 lib ehsi-+climrate,ehsi.crs;

climfilt.obj : climfilt.c climfilt.h
 msc climfilt;
 lib ehsi-+climfilt,ehsi.crs;

arc_circ.obj : arc_circ.c
 msc arc_circ;
 lib ehsi-+arc_circ,ehsi.crs;

str_gen.obj : str_gen.c msc str_gen; lib ehsi-+str_gen,ehsi.crs;

plane.obj : plane.c
 msc plane;
 lib ehsi-+plane,ehsi.crs;

waypoint.cbj : waypoint.c
 msc waypoint;
 lib ehsi-+waypoint,ehsi.crs;

vortac.cbj : vortac.c
 msc vortac;
 lib ehsi-+vortac,ehsi.crs;

box.obj : box.c msc box; lib ehsi-+box,ehsi.crs;

compass.obj : compass.c msc compass; lib ehsi-+compass,ehsi.crs;

clim_aro.obj : clim_aro.c
 msc clim_aro;
 lib ehsi-+clim_aro,ehsi.crs;

arrow.obj : arrow.c

msc arrow; lib ehsi-+arrow, ehsi.crs;

ndb.obj : ndb.c

msc ndb;

lib ehsi-+ndb,ehsi.crs;

heading.obj : heading.c msc heading; lib ehsi-+heading,ehsi.crs;

runway.obj : runway.c msc runway; lib ehsi-+runway,ehsi.crs;

zero_pad.obj : zero_pad.c
 msc zero_pad;
 lib ehsi-+zero_pad,ehsi.crs;

timer.obj : timer.c data_str.h
 msc timer;
 lib ehsi-+timer,ehsi.crs;

dme.obj : dme.c data_str.h
 msc dme;
 lib ehsi-+dme,ehsi.crs;

ndb_angl.obj : ndb_angl.c
 msc ndb_angl;
 lib ehsi-+ndb_angl,ehsi.crs;

airspeed.obj : airspeed.c data_str.h
 msc airspeed;
 lib ehsi-+airspeed,ehsi.crs;

climrate.obj : climrate.c data_str.h
 msc climrate;
 lib ehsi-+climrate,ehsi.crs;

ils_cmps.obj : ils_cmps.c
 msc ils_cmps;
 lib ehsi-+ils_cmps,ehsi.crs;

hdg_brg.obj : hdg_brg.c msc hdg_brg; lib ehsi-+hdg_brg,ehsi.crs; key_alrm.obj : key_alrm.c msc key_alrm; lib key_ehsi-+key_alrm,key_ehsi.crs;

rest rey_airm, key_ensi.crs;

key_dat.obj : key_dat.c
 msc key_dat;
 lib key_ehsi-+key_dat, key_ehsi.crs;

key_nav.obj : key_nav.c msc key_nav; lib key_ehsi-+key_nav,key_ehsi.crs;

key_ils.obj : key_ils.c
 msc key_ils;
 lib key_ehsi-+key_ils, key_ehsi.crs;

key_entr.obj : key_entr.c
 msc key_entr;
 lib key_ehsi-+key_entr,key_ehsi.crs;

key_cler.obj : key_cler.c
 msc key_cler;
 lib key_ehsi-+key_cler,key_ehsi.crs;

key_freq.obj : key_freq.c data_str.h
msc key_freq;
lib key_ehsi-+key_freq,key_ehsi.crs;

key_stmr.obj : key_stmr.c data_str.h
msc key_stmr;
lib key_ehsi-+key_stmr, key_ehsi.crs;

key_math.obj : key_math.c msc key_math; lib key_ehsi-+key_math,key_ehsi.crs;

key_buff.obj : key_buff.c
 msc key_buff;
 lib key_ehsi-+key_buff,key_ehsi.crs;

key_alt.obj : key_alt.c data_str.h
 msc key_alt;
 lib key_ehsi-+key_alt, key_ehsi.crs;

key_wind.obj : key_wind.c data_str.h
 msc key_wind;
 lib key_ehsi-+key_wind, key_ehsi.crs;

key_cmd3.obj : key_cmd3.c

- msc key_cmd3;
 lib key_ehsi-+key_cmd3,key_ehsi.crs;
- dat_pg_s.obj : dat_pg_s.c datpg_xy.h
 msc dat_pg_s;
- dat_pg_d.obj : dat_pg_d.c data_str.h datpg_xy.h
 msc dat_pg_d;
- nav_pg_s.obj : nav_pg_s.c navpg_xy.h
 msc nav_pg_s;
- nav_pg_d.obj : nav_pg_d.c data_str.h navpg_xy.h
 msc nav_pg_d;
- ils_pg_s.obj : ils_pg_s.c ilspg_xy.h
 msc ils_pg_s;
- int_ehsi.obj : int_ehsi.asm
 masm int ehsi:
- com_ehsi.obj : com_ehsi.asm
 masm com_ehsi;
- ehsi.obj : ehsi.c data_str.h
 msc ehsi;
- link ehsi dat_pg_s nav_pg_s ils_pg_s dat_pg_d nav_pg_d
 ils_pg_d com_ehsi int_ehsi/stack:4000,ehsi,ehsi,
 ehsi.lib key_ehsi.lib;

APPENDIX D

EHSI HOST PROGRAM SOFTWARE LISTINGS

```
/***********************
    SOURCE FILE:
                    ehsi.c
   FUNCTION:
                    main()
   DESCRIPTION:
                    Controls the actions taken on receipt
                    of an interrupt vector from the
                    interrupt vector stack. Iniatializa-
                    tion and restoration are also perform-
                    ed from this routine. The data pack-
                    age, SCREEN array interrupt vector
                    stack, and calculator stack are
                    declared within this routine.
    DOCUMENTATION
    FILES:
                    None.
   ARGUMENTS:
                    None.
   RETURN:
                    None.
   FUNCTIONS
   CALLED:
                   None.
   AUTHOR:
                    Dave Gruenbacher & Chuck Robertson
   DATE CREATED:
                  19Jan87
                               Version 1.0
   REVISIONS:
                    None.
***********************
#define ehsi_main
#include <stdio.h>
#include "data_str.h"
void main()
   static unsigned short int_depth = 0, int_stack[90]=(0);
   int page_number = 1, i, int_number;
```

```
char
        key buffer[50]:
 double x_buffer, y buffer:
 htov
        INITIALIZE():
 void
        RESTORE():
 void
        dat_pg_dynamic():
 void
        nav_pq_dynamic():
 void
        ils pg dynamic():
 void
        dat_pg static():
 void
        nav_pq_static();
 vo1d
        ils pg static():
 void
       update key buffer():
void roll stack();
btov
      set altitude():
void
      set estimated wind():
void
      exit():
void
        clear_stack():
void
        insert_new_freq():
void
        set timer():
void reset_alarm();
void
       do math():
void call cmd3();
       display_ils_page();
void
biov
       display_data_page();
void
       display_nav_page();
CLOCK_PKG clock_pkg;
ALARM_PKG alarm_pkg;
clock_pkg.timer_min = 0;
clock_pkg.timer_sec = 0:
clock_pkg.time_out_min = 0;
clock_pkg.time_out_sec = 0;
clock_pkg.adf_freq = 242.0:
clock_pkg.com1_freq = 119.1;
clock_pkg.com2_freq = 121.9;
clock_pkg.vor1_freq = 112.6;
clock_pkg.vor2_freq = 110.1;
clock_pkg.assigned_altitude = 0:
clock_pkg. mda_dh = 0;
clock_pkg.estimated_wind = 0;
clock_pkg.timer_operation_flag = NULL_TIMER;
clock_pkg.timer_status_flag = TIMER_OFF;
clock_pkg.math_operation_flag = 0;
alarm_pkg.airspeed_alarm_flag = ALARM_OFF;
alarm_pkg.assigned_altitude_alarm_flag = ALARM_OFF;
alarm_pkg.mda_dh_alarm_flag = ALARM_OFF;
alarm_pkg.time_out_alarm_flag = ALARM OFF:
alarm_pkg.alarm_status_flag = ALARM_OFF;
```

```
alarm_pkg.assigned_altitude_enable_flag = DISABLED;
 alarm_pkg.mda_dh_enable_flag = DISABLED;
 INITIALIZE(&int_depth, int_stack);
 key_buffer[0] = '\0';
for (;;)
     if (int_depth != 0)
/ *
           if (int_stack[0] != 0x60)
             printf("%X %X\n", int_depth, int_stack[0]);
         int_number = int_stack[0];
         int_depth -= 1;
         for (i=0;i!=int_depth;i++)
             int_stack[i] = int_stack[i+1];
    switch (int_number) (
    case 0x60:
        int_number = 0;
        switch (page_number) {
            case 1:
                  dat_pg_dynamic(&clock_pkg, &alarm_pkg);
                  break:
            case 2:
                 nav_pg_dynamic(&clock_pkg, &alarm_pkg);
                 break:
            case 3:
                 ils_pg_dynamic(&clock_pkg, &alarm_pkg);
                 break:
            default:
                 break:
        break:
    case 0x65:
        int_number = 0:
        printf("\n\nSYSTEM SWITCH ON.\n\n");
        /* set up vector table */
        dat_pg_static();
                           /* Install data page static
                               mem. */
        for (i=0;i!=100;i++);
       nav_pg_static(); /* Install nav. page static
                              mem. */
```

```
for (i=0;i!=100;i++);
    ils_pg_static(); /* Install ils page static
                          mem. */
    break:
case 0x66:
    RESTORE():
    printf("\n\nSYSTEM SWITCH OFF.\n\n");
    exit();
    break:
default:
    if ((int_number == 0) || (int_number > 0x23))
       break:
   switch (int number) {
   case 0x04:
                       /* O received from keypad */
   case 0x05:
                       /* . received from keypad */
   case OxOA:
                       /* 1 received from keypad */
   case OxOB:
                       /* 2 received from keypad */
   case 0x0C:
                      /* 3 received from keypad */
   case 0x10:
                      /* 4 received from keypad */
   case Ox11:
                      /* 5 received from keypad */
   case 0x12:
                      /* 6 received from keypad */
   case 0x16:
                      /* 7 received from keypad */
                       /* 8 received from keypad */
   case 0x17:
   case 0x18:
                       /* 9 received from keypad */
       /* clear the buffer if a math operation was
          nust completed.
       if (clock_pkg.math_operation_flag == 1)
          key_buffer[0] = '\0';
          clock_pkg.math_operation_flag = 0;
       update_key_buffer(int_number, key_buffer):
       int_number = 0:
       break:
   case 0x02:
                       /* ENTER hit on keypad
       roll_stack(key_buffer, &x_buffer, &y_buffer);
       int_number = 0:
       break:
   case 0x08:
                       /* CLEAR hit on keypad
       int_number = 0;
      clear_stack(key_buffer, &x_buffer, &y_buffer);
      break;
```

```
case 0x01:
                                 /* new COM1 freg. entered */
             case 0x07:
                                 /* new COM2 freq. entered */
             case OxOD:
                                 /* new VOR1 freq. entered */
             case Ox13:
                                 /* new VOR2 freq. entered */
             case 0x19:
                                 /* new ADF freq. entered */
                insert_new_freq(int_number, key_buffer.
                                 &clock pkg);
                 int number = 0:
                 break:
             case Ox1R:
                                 /* new mda/dh entered
             case Ox1C:
                                 /* new asgn. alt. entered */
                set_altitude(int_number, key_buffer,
                              &clock pkg, &alarm pkg):
                 int_number = 0:
                 break:
             case Ox1D:
                                  /* new est. wind entered */
                 int_number = 0:
break:
                 set_estimated_wind(key_buffer, &clock_pkg);
             case Ox22:
                                    /* SET TIMER hit
                                                            */
                 int number = 0:
                 set_timer(key_buffer, &clock_pkg);
                clock_pkg.timer_operation_flag = SET_TIMER;
                break:
            case Ox1F:
                                   /* START TIMER hit
                int_number = 0:
                clock_pkg.timer_operation_flag
                                  = START_TIMER;
                break:
            case 0x23:
                                    /* RESET TIMER hit
                                                           */
                int_number = 0:
                clock_pkg.timer_operation_flag
                                 = RESET_TIMER:
                break:
            case Ox1E:
                                    /* SET/RST ALRM hit */
                int_number = 0;
                reset_alarm();
                break:
            case 0x03:
                                /* div key hit on keypad */
            case 0x09:
                                /* mult key hit on keypad */
            case OxOF:
                                /* add key hit on keypad */
            case 0x15:
                                /* sub key hit on keypad */
                do_math(int_number, key_buffer, &x_buffer,
                        &y_buffer);
```

```
int_number = 0;
                 clock_pkg.math_operation_flag = 1;
                 break:
             case 0x21:
                 call_cmd3(key_buffer, &x_buffer, &y_buffer);
                 int number = 0:
                 break;
             case Ox1A:
                 int_number = 0;
                 page_number = 1;
                 display_data_page();
                 break:
            case 0x14:
                 int_number = 0:
                 page_number = 2;
                 display_nav_page();
                 break;
            case OxOE:
                 int_number = 0;
                 page_number = 3:
                 display_ils_page();
                break:
            default:
                int_number = 0;
                break;
            break;
        }
      }
   }
}/* end main */
```

SOURCE FILE: dat_pg.c

FUNCTION: dat_pg_static()

DESCRIPTION: This routine generates the static data for the DATA PAGE. It is only

called once when the SYSTEM SWITCH is

turned on.

DOCUMENTATION

×

FILES: None.

ARGUMENTS: None.

RETURN: None.

FUNCTIONS

CALLED: UPDATE_SCREEN()

strcpy()
string_gen()
insert()
box()
line()
arc_circ()
climb box()

clim hsh()

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 15Jan87 Version 1.0

REVISIONS: None.

#include<stdio.h>

#include "datpg_xy.h"

```
#include "data_str.h"
void dat pg static()
    unsigned short conv[300];
                                  /* conversion array */
   unsigned short address ptr;
                                   /* address pointer */
   unsigned short line_char;
                                   /* line characteristic */
   unsigned short internal_jmp;
                                   /* HP internal jump */
   int XO, YO;
                                   /* plot coordinates */
   int p:
                                   /* conversion array
                                                  index */
   int length;
                                   /* insertion length */
   int i:
                                   /* loop counter */
   int total sent;
                                   /* total words sent to
                                      vector memory
   int error:
   char final_string(30);
                                   /* string arrays */
   char cat_string[10];
   double size;
                                   /* character size */
   int insert():
   void string_gen();
   void box();
   void line():
   void arc_circ();
   void climb_box();
   void clim_hsh();
   address_ptr = 0xC000;
                             /* point at first byte in
                                vector memory
                                                           */
   internal_jmp = 0x8001;
                             /* store jump to 0001 in
                                vector memory
                                                           */
   line_char = 0x7818;
                            /* set line characteristics */
   size = 1.5;
                             /* set character size */
   p=0:
   SCREEN[p++] = address_ptr;
   SCREEN[p++] = internal_jmp;
  SCREEN(p++) = line_char;
         strcpy(final_string, "HEADING:");
         string_gen(final_string, heading_XO, heading_YO,
                    size, &length, conv);
         p = insert(p, length, conv);
        strcpy(final_string, "AIRSPEED:
        string_gen(final_string, airspeed_CAS_XO,
                    airspeed_CAS_YO, size, &length, conv);
        p = insert(p,length,conv);
```

```
box(col1, 1890, size, length-2, &length, conv);
 p = insert(p, length, conv);
 strcpy(final string, "
                                       TAS"):
 string_gen(final_string, airspeed_TAS_XO,
             airspeed_TAS_YO, size, &length, conv);
 p = insert(p,length,conv);
 strcpy(final_string, "GNDSPEED:");
 string_gen(final_string, gndspeed_XO,
            gndspeed_YO, size, &length, conv);
p = insert(p, length, conv);
strcpy(final_string, "ASSIGNED:"):
string_gen(final_string, assigned XO.
            assigned_YO, size, &length, conv);
p = insert(p, length, conv);
strcpy(final_string, "ALTITUDE:
string_gen(final_string, altitude_XO,
            altitude_YO, size, &length, conv);
p = insert(p, length, conv);
box(altitude_XO, altitude_YO, size, length-2,
    &length, conv):
p = insert(p, length, conv);
strcpy(final_string, "MDA/DH:");
string_gen(final_string, mda_dh_XO,
            mda_dh_YO, size, &length, conv):
p = insert(p, length, conv);
strcpy(final_string, "TIMER:
string_gen(final_string, timer_XO,
            timer_YO, size, &length, conv);
p = insert(p, length, conv);
box(timer_XO, timer_YO, size, length-2, &length, conv);
p = insert(p, length, conv);
strcpy(final_string, "TIME-OUT: ");
string_gen(final_string, time_out_XO,
            time_out_YO, size, &length, conv);
p = insert(p, length, conv);
strcpy(final_string, *COM1:
string_gen(final_string,com1_XO,com1_YO,
           size, &length, conv);
p = insert(p, length, conv);
```

```
box(com1_X0, com1_Y0, size, length-2, &length, conv);
  p = insert(p, length, conv);
  strcpy(final_string, "COM2:");
  string_gen(final_string,com2_XO,com2_YO,
             size, &length, conv);
  p = insert(p,length,conv);
 strcpy(final_string, "NAV1:");
 string gen(final_string, nav1 XO.
             nav1_YO, size, &length, conv);
 p = insert(p, length, conv);
 strcpy(final_string, "NAV2:");
 string_gen(final_string, nav2_XO,
             nav2_YO, size, &length, conv);
 p = insert(p, length, conv);
 strcpy(final_string, "RNAV: WP1
                                           . "):
string_gen(final_string, rnav_wp1_XO,
             rnav_wp1_YO, size, &length, conv);
 p = insert(p, length, conv);
strcpy(final_string, "
                            WP2
                                          . "):
 string_gen(final_string, rnav_wp2_XO.
             rnav_wp2_YO, size, &length, conv);
 p = insert(p, length, conv);
 strcpy(final_string, "ADF:");
 string_gen(final_string, adf_XO, adf_YO, size,
             &length, conv):
 p = insert(p, length, conv);
strcpy(final_string, "TEMP: ");
string_gen(final_string, temp_XO,
            temp_YO, size, &length, conv);
 p = insert(p, length, conv);
strcpy(final_string, "BAROMETER:");
string_gen(final_string, barometer_XO,
            barometer_YO, size, &length, conv);
p = insert(p, length, conv);
/* PUT UP REAL TIME CLOCK INFORMATION */
strcpy(final_string, "TIME: ");
string_gen(final_string, time XO.
            time_YO, size, &length, conv);
p = insert(p, length, conv);
```

```
strcpy(final_string, "EDT");
 string_gen(final_string, time_edt_XO,
             time_edt_YO, size, &length, conv);
 p = insert(p, length, conv);
 strcpy(final_string, "ZULU");
 string_gen(final_string, time_zulu_XO,
             time_zulu_YO, size, &length, conv);
p = insert(p, length, conv);
strcpy(final_string, "SINCE L/O"):
string_gen(final_string, time_since_lo_XO,
            time_since_lo_YO, size, &length, conv);
p = insert(p, length, conv);
/* PUT UP STATIC INFORMATION FOR CLIMB
    INDICATOR
/* climb indicator lettering */
strcpy(final_string, "C");
string_gen(final_string, C_XO, C_YO, size, &length,
            conv);
p = insert(p, length, conv);
strcpy(final_string, "L");
string_gen(final_string, L_XO, L_YO, size, &length,
            conv);
p = insert(p, length, conv);
strcpy(final_string, "I");
string_gen(final_string, I_XO, I_YO, size, &length,
            conv);
p = insert(p,length,conv);
strcpy(final_string, "M");
string_gen(final_string, M_XO, M_YO, size, &length,
            conv):
p = insert(p, length, conv);
strcpy(final_string, "B");
string_gen(final_string, B_XO, B_YO, size, &length,
            conv);
p = insert(p, length, conv);
/* climb indicator arc */
arc_circ(climb_arc_XO, climb_arc_YO, -20.0, 20.0,
         1000, &length, conv);
p = insert(p,length,conv);
```

```
/* climb indicator center lines */
       line(climb_centerline1_XO, climb_centerline1_YO,
            climb_centerline1_X1, climb_centerline1 Y1.
             &length, conv);
       p = insert(p, length, conv);
       line(climb_centerline2_XO, climb_centerline2_YO,
            climb_centerline2_X1, climb_centerline2_Y1,
            &length, conv):
       p = insert(p,length,conv);
       line(climb_centerline3_XO, climb_centerline3_YO,
            climb_centerline3_X1, climb_centerline3_Y1.
            &length, conv);
       p = insert(p, length, conv);
       line(climb_centerline4_XO, climb_centerline4_YO,
            climb_centerline4_X1, climb_centerline4_Y1,
            &length, conv);
       p = insert(p,length.conv):
       line(climb_centerline5_XO, climb_centerline5_YO,
            climb_centerline5_X1, climb_centerline5_Y1,
            &length, conv):
       p = insert(p, length, conv);
       /* box climb indicator lettering */
      climb_box(climb_arc_XO, climb_arc_YO, &length, conv);
      p = insert(p, length, conv);
      /* climb indicator arc hash marks */
      clim_hsh(climb_arc_XO, climb_arc_YO, -20.0, 20.0,
                1000, &length, conv);
      p = insert(p, length, conv);
      /* COMMAND LINE */
      line(command_line_XO, command_line_YO,
           command_line_X1, command line Y1.
           &length, conv);
      p = insert(p,length,conv);
SCREEN(p++) = 0x8700;
SCREEN[p++] = 0xC700;
SCREEN[p++] = 0x0000;
SCREEN[p++] = 0x8FFF;
```

```
SCREEN[p++] = OxCEOO;
SCREEN[p++] = OxOOOO;
SCREEN[p++] = Ox6FFF;
SCREEN[p++] = OxFFFF;

error = 1;
while (error != 0)
    error = SEND_SCREEN( );
} /* end of dat_pg_static */
```

SOURCE FILE:

dat_pg_d.c

FUNCTION:

*

dat_pg_dynamic()

DESCRIPTION:

This function generates all the dynamic data for the DATA PAGE. It retrieves raw data from the data package and makes necessary conversions. Presently it is displaying the following:

- 1) real time clock
- 2) plane heading
- 3) present altitude
- 4) rate of climb and climb arrow
- 5) airspeed
- 6) adf frequency
- 7) com frequencies
- 8) nav frequencies
- 9) timer and time_out functions
- 10) alarm conditions

DOCUMENTATION

FILES:

None.

ARGUMENTS:

clock_pkg : pointer to clock package
alarm_pkg : pointer to alarm package

RETURN:

None.

FUNCTIONS CALLED:

¥

×

×

GET_DATA_PACKAGE()
TOGGLE_ALARM_SWITCH()
UPDATE_SCREEN()
insert()
string_gen()
time_string_gen()
get_heading()
clim_arrow()
box()

zero_pad()
timer()

```
airspeed()
                     altitude()
                     climb rate()
   AUTHOR:
                    CHUCK ROBERTSON
   DATE CREATED: 01Feb87 Version 1.0
   REVISIONS:
                    None.
***********************
#include "data_str.h"
#include "datpg_xy.h"
#include<stdlib.h>
#include<string.h>
#include<math.h>
#include<stdio.h>
void dat_pg_dynamic(clock_pkg, alarm_pkg)
CLOCK_PKG *clock_pkg;
ALARM_PKG *alarm_pkg;
{
   unsigned short conv[300];
                               /* conversion array */
   int p:
                                /* conversion array
                                    index */
   int length;
                                 /* insertion length */
   int i:
                                /* loop counter */
   int vert speed:
                                /* rate of climb */
   int plane_heading_deg;
                                /* plane heading */
   int times:
                                /* request counter */
   int error:
                                /* masm return error
                                   code */
   int page_number = 1;
                                /* page being
                                   displayed */
   char final_string[20];
                                /* conversion string
                                   arrays */
   char cat_string[5];
   double size:
                               /* character size */
   double climb_degrees:
                                /* angle of climb
                                   arrow */
   int insert();
```

```
void string_gen();
    void time_string gen();
    void get heading():
    void clim arrow():
    void box():
    void zero_pad();
    void timer():
    void airspeed();
    void altitude();
    void climb rate():
    times = 0:
    error = 1:
    while (error i= 0)
         times++:
         error = GET_DATA_PACKAGE( );
/ ¥
     if(times != 1)
      printf("CK DP times = %d\n", times);
                                                */
   p=0:
   SCREEN[p++] = 0xC700;
   SCREEN[p++] = 0x7818:
                                         /* set line size */
   size = 1.5:
       CONVERT TIME TO CHAR STRING */
   time_string_gen(final_string);
   string_gen(final_string, time_edt_X1, time_edt_Y1, size,
               &length, conv);
   p = insert(p, length, conv);
   /* PUT UP HEADING */
   get_heading(&plane_heading_deg);
   zero_pad((double)(plane_heading_deg), 100.0,6,
            "int", final_string);
   string_gen(final_string, heading_X1, heading_Y1, size,
               &length, conv);
   p = insert(p, length, conv);
   /* PUT UP ASSIGNED ALTITUDE */
   zero_pad((double)(clock_pkg->assigned_altitude), 1000.0,
            6, "int", final_string);
```

```
if(alarm_pkg->assigned_altitude_enable_flag == DISABLED)
    string_gen(final_string, assigned_X1, assigned_Y1,
                size, &length, conv):
 else
    strcat(final_string, " ALM");
    string_gen(final_string, assigned_X1, assigned Y1.
               size, &length, conv);
    3
p = insert(p, length, conv);
 /* PUT UP ALTITUDE */
altitude(clock_pkg,alarm_pkg,altitude_X1,altitude_Y1,
          &length, conv):
p = insert(p, length, conv);
    PUT UP MDA/DH */
zero_pad((double)(clock_pkg->mda_dh), 1000.0,6, "int",
          final_string);
if(alarm_pkg->mda_dh_enable_flag == DISABLED)
   string_gen(final_string, mda_dh_X1, mda_dh_Y1, size,
               &length, conv):
else
   strcat(final_string, " ALM");
   string_gen(final_string, mda_dh_X1, mda_dh_Y1, size.
               &length, conv):
p = insert(p, length, conv);
   PUT UP CLIMB VALUES */
climb_rate(clock_pkg, alarm_pkg, vert_speed_XO,
           vert_speed_YO, &vert_speed,
           &length, conv);
p = insert(p,length,conv);
box(vert_speed_X0, vert_speed_Y0, size, length-2,
    &length, conv);
p = insert(p, length, conv);
/* calculate climb arrow */
climb_degrees = (double)(vert_speed / 100.0);
clim_arrow(climb_arc_XO,climb_arc_YO,climb_degrees,
           1000, &length, conv);
```

```
p = insert(p, length, conv);
 /* PUT UP AIRSPEED */
 airspeed(page_number, clock_pkg, alarm_pkg,
          airspeed_CAS_X1, airspeed CAS Y1.
          &length.conv):
 p = insert(p, length, conv);
 /* PUT UP ADF */
zero_pad(clock_pkg->adf_freg, 100.0, 6, "int",
          final string):
string_gen(final_string, adf_X1, adf_Y1, size, &length,
            conv):
p = insert(p, length, conv);
   PUT UP COM1
zero_pad(clock_pkg->com1_freq, 100.0, 6, "double",
          final_string);
string_qen(final_string,com1_X1,com1_Y1,size,&length,
            conv):
p = insert(p, length, conv);
/* PUT UP COM2 */
zero_pad(clock_pkg->com2_freq, 100.0, 6, "double",
          final string):
string_gen(final_string,com2_X1,com2_Y1,size,&length,
            conv):
p = insert(p, length, conv);
/* PUT UP VOR1 */
zero_pad(clock_pkg->vor1_freq, 100.0, 6, "double".
         final_string);
string_gen(final_string, nav1_X1, nav1_Y1, size, &length,
           conv);
p = insert(p, length, conv);
/* PUT UP VOR2 */
zero_pad(clock_pkg->vor2_freq, 100.0, 6, "double",
         final_string);
string_gen(final_string, nav2_X1, nav2_Y1, size, &length,
           conv):
p = insert(p, length, conv);
/* PUT UP TIMER */
```

SOURCE FILE: nav_pg_s.c

FUNCTION: nav_pg_static()

DESCRIPTION: This routine generates the static data

for the NAV PAGE. It is only called once when the SYSTEM SWITCH is turned

on.

DOCUMENTATION

FILES: None.

ARGUMENTS: None.

RETURN: None.

FUNCTIONS

CALLED: UPDATE_SCREEN()

strcpy()
string_gen()
insert()
box()
line()
arc_circ()

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 15Jan87 Version 1.0

REVISIONS: None.

*

#include<stdlib.h>
#include<string.h>

#include<stdio.h>

#include "navpg_xy.h"

#include "data_str.h"

```
void nav_pg_static()
   unsigned short conv[300];
                                   /* conversion array */
   unsigned short address_ptr;
                                    /* address pointer */
   unsigned short line_char;
                                    /* line characteristic */
   unsigned short internal jmp;
                                    /* HP internal jump */
   int XO, YO;
                                    /* plot coordinates */
   int p:
                                    /* conversion array
                                       index */
   int length:
                                   /* insertion length */
   int i:
                                   /* loop counter */
   int total_sent;
                                   /* total words sent to
                                      vector memory
   int error:
                                   /* masm return error */
   char final_string[30];
                                   /* string arrays */
   char cat_string[10];
   double size:
                                   /* character size */
   int insert();
   void string_gen();
   void box():
   void line():
   void arc circ():
   void plane();
   address_ptr = 0xC300;
                            /* point at 0x0300 word in
                                vector memory
   line_char = 0x7818:
                            /* set line characteristics */
   size = 1.5;
                             /* set character size */
   p=0:
   SCREEN[p++] = address_ptr;
   SCREEN[p++] = line_char;
         /* PUT UP AIRSPEED */
         strcpy(final_string, "AIRSPD:");
         string_gen(final_string, airspeed_XO, airspeed_YO,
                    size, &length, conv);
         p = insert(p, length, conv):
         /* PUT UP GROUNDSPEED */
        strcpy(final_string, "GNDSPD: ");
        string_gen(final_string, gndspeed_XO, gndspeed_YO,
                    size, &length, conv);
        p = insert(p, length, conv);
        /* PUT UP TRACK */
```

```
strcpy(final_string, "TRK:");
 string_gen(final_string, track_XO, track_YO.
             size, &length, conv);
 p = insert(p, length, conv);
 /* PUT UP ALTITUDE */
 strcpy(final_string, "ALT:");
 string_gen(final_string, altitude_XO, altitude_YO,
            size, &length, conv);
 p = insert(p,length,conv);
 /* PUT UP DME */
 strcpy(final_string, "DME:
string_gen(finel_string, dme_XO, dme_YO, size,
            &length, conv):
p = insert(p, length, conv);
box(dme_X0, dme_Y0, size, length-2, &length, conv);
p = insert(p, length, conv);
/* PUT UP PLANE */
plane(plane_XO, plane_YO, &length, conv);
p = insert(p, length, conv);
/* PUT UP STATIC ARC OF COMPASS */
arc_circ(compass_arc_XO, compass_arc_YO,
          compass_arc_init_angle,
          compass_arc_end_angle,
          compass_arc_radius, &length, conv);
p = insert(p,length,conv):
/* PUT UP HEADING BOX AND POINTER */
box(heading_box_X0, heading_box_Y0, size,
    heading_box_no_char, &length, conv);
p = insert(p, length, conv);
line(heading_ptr_line1_XO, heading_ptr_line1_YO,
     heading_ptr_line1_X1, heading_ptr_line1_Y1,
     &length, conv);
p = insert(p, length, conv);
line(heading_ptr_line2_XO, heading_ptr_line2_YO,
     heading_ptr_line2_X1, heading_ptr_line2_Y1,
     &length, conv);
p = insert(p,length,conv);
```

SOURCE FILE: nav_pg_d.c

FUNCTION: nav_pg_dynamic()

DESCRIPTION: This function generates all the

dynamic data for the NAV PAGE. It retrieves raw data from the data package and makes necessary conversions. Presently it is displaying the following:

1) airspeed

2) ground speed

3) ground track

4) dme

5) plane heading

6) compass

7) vortac's

8) ndb's

9) alarm conditions

10) altitude

DOCUMENTATION

FILES: None.

ARGUMENTS: clock_pkg : pointer to clock package

alarm_pkg : pointer to alarm package

RETURN: None.

FUNCTIONS

×

*

*

¥

CALLED: GET_DATA_PACKAGE()

TOGGLE_ALARM_SWITCH()

UPDATE_SCREEN()

insert()

string_gen()

get_heading()
clim arrow()

box()

zero_pad()

vortac()

ndb()

```
line()
                     compass()
                     heading_and_bearing()
                     airspeed()
                     altitude()
                     dme()
                     get_ndb_plot_angle()
                     timer()
 ¥
    AUTHOR:
                     CHUCK ROBERTSON
 ¥
   DATE CREATED: 09Feb87 Version 1.0
 ¥
 ¥
   REVISIONS:
                    None.
**********************
#include "data_str.h"
#include "navpg_xy.h"
#include<stdlib.h>
#include<string.h>
#include<math.h>
#include<stdio.h>
void nav_pg_dynamic(clock_pkg, alarm pkg)
CLOCK_PKG *clock_pkg;
ALARM_PKG *alarm pkg;
   unsigned short conv[300];
                                 /* conversion array */
   int p;
                                  /* conversion array
                                     index */
   int length;
                                 /* insertion length */
   int i:
                                 /* loop counter */
   int plane_heading_deg;
                                 /* plane heading */
   int times;
                                 /* request counter */
   int radix:
                                 /* conversion radix */
   int int_dme;
int X, Y;
                                 /* integer dme value */
                                 /* plot coordinates */
   int error:
                                 /* masm error return */
   int page_number = 2;
                                 /* page being
                                     displayed */
   char final_string[20];
                                 /* conversion string
                                    arrays */
```

```
char cat_string[20];
int
         insert():
void string gen();
void box();
void get_heading();
void vortac();
void ndb();
void line();
void compass();
       heading_and_bearing();
zero_pad();
void
void zero_pad();
void airspeed();
void altitude();
void dme();
       get_ndb_plot_angle();
void
void
        timer():
double size;
                              /* character size
                                                         */
double real_delta_x;
                              /* converted delta x and */
double real_delta_y;
                              /* delta y relative to
                                  VOR
double plane_delta_x;
                             /* converted delta x and */
double plane_delta_y;
                             /* delta y relative to
                                 plane */
double theta_rad;
                             /* calculation angle
                                 degrees */
double theta_deg;
                              /* calculation angle
                                 radians */
double screen_scale;
double PI:
double DEGREE_TO_RAD;
double vor_theta_rad;
                              /* vor plot angle in
                                 radians */
double vor_theta_deg;
                              /* vor plot angle in
                                 degrees */
double radius;
                              /* plot radius
                                                         */
double real_dme;
                              /* real dme value
                                                        */
double vor_distance_scale;
                             /* scale for vor
                                 distance
double ndb_theta_deg;
                             /* ndb plot angle in
                                 degrees */
                              /* ndb plot angle in
double ndb theta rad:
                                 radians */
```

```
times = 0:
   error = 1;
    while (error != 0)
         times++;
         error = GET_DATA_PACKAGE( );
/* if (times != 1)
         printf("NV DP times = %d\n", times); */
   screen_scale = 9.5/12.5;
   PI = 3.14159265;
   DEGREE_{TO}RAD = (2 * PI / 360.0);
   size = 1.5;
   p=0;
   SCREEN[p++] = OxC900;
   SCREEN[p++] = 0x7818;
                                         /* set line size */
   /* PUT UP AIRSPEED */
   airspeed(page_number, clock_pkg, alarm_pkg,
             airspeed_X1, airspeed_Y1,
             &length, conv);
   p = insert(p, length, conv);
   /* PUT UP GROUNDSPEED */
   /* yet to be implemented */
   /* PUT UP TRACK */
   /* yet to be implemented */
   /* PUT UP ALTITUDE */
   altitude(clock_pkg, alarm_pkg, altitude_X1, altitude_Y1,
             &length, conv);
   p = insert(p, length, conv);
   /* PUT UP DME */
   dme(clock_pkg, alarm_pkg, dme_X1, dme_Y1, &length.conv):
   p = insert(p, length, conv);
   /* PUT UP COMPASS AND HEADING */
   get_heading(&plane_heading_deg);
```

```
zero_pad((double)(plane_heading_deg), 100.0,6,
         "int", final_string);
string_gen(final_string, heading_XO, heading_YO, size,
           &length, conv);
p = insert(p, length, conv);
compass(compass_arc_XO, compass_arc_YO, plane_heading_deg,
        compass_arc_radius, &length, conv);
p = insert(p, length, conv);
/* PUT UP VORTAC */
/* calculate real dme value */
real_dme = (double)((data_pkg.DME - 2.0) * 0.207);
/* plot vortac if plane is within 20 miles */
if(real_dme <= 30.0)
      /* calculate delta x and y relative to vor */
      real_delta_x = (double)
                      ((data_pkg.DELTA_X - 126.0)
                        * 0.392):
      real_delta_y = (double)
                      ((data_pkg.DELTA_Y - 126.0)
                       * 0.415);
      /* calculate delta x and y relative to plane */
      plane_delta_x = -(real_delta_x);
      plane_delta_y = -(real_delta_y);
      /* CALCULATE POSITION OF VOR RELATIVE TO PLANE */
      if( (plane delta y != 0.0)
          && (plane_delta_x != 0.0) )
             /* calculate vor angle relative to plane
                position
             theta_rad = atan2(plane_delta_y,
                               plane_delta x);
             /* add in the planes heading */
             vor_theta_rad = theta rad +
                             (plane_heading_deg *
```

```
DEGREE_TO RAD);
       vor_theta_deg = vor_theta_rad /
                               DEGREE TO RAD:
       if(vor_theta_deg >= 360.0)
           vor_theta_deg = vor_theta_deg - 360.0;
   }
else
   vor_theta_deg = 0.0:
   /* calculate new vor_theta_rad */
   vor_theta_rad = vor_theta_deg * DEGREE TO RAD:
   /* calculate plot radius */
   if(real dme <= 0.0)
      radius = 0.0:
   else
      radius = 406.2 * log10(real_dme);
   /* calculate vortac plot coordinates */
   X = (int)(radius * cos(vor_theta_rad)
                  * screen scale):
   Y = (int)(radius * sin(vor_theta_rad));
   /* plot vortac */
   vortac(X + plane_X0, Y + plane_Y0, &length, conv);
   p = insert(p, length, conv);
   /* put up vortac line */
   line(plane_XO, plane_YO, plane_XO + X,
        plane_YO + Y, &length, conv);
   p = insert(p, length, conv);
   /* put up bearing and heading if radius
      > 400 */
   if(radius >= 400.0)
   /* put vortac heading and bearing */
   heading_and_bearing(plane_XO, plane_YO,
                       plane_heading deg,
```

```
vor_theta_deg,
                               radius, &length, conv);
          p = insert(p, length, conv);
          3
   }
/* PUT UP NDB */
get_ndb_plot_angle(&ndb_theta deq);
ndb_theta_rad = ndb_theta_deg * DEGREE TO RAD:
radius = 700.0:
/* calculate ndb plot coordinates */
X = (int)(radius * cos(ndb_theta_rad) * screen_scale);
Y = (int)(radius * sin(ndb_theta_rad));
/* plot ndb */
ndb(X + plane_XO, Y + plane_YO, &length, conv);
p = insert(p, length, conv);
/* put up ndb line */
line(plane_XO, plane_YO, plane_XO + X, plane_YO + Y,
     &length, conv);
p = insert(p, length, conv);
/* put up ndb heading and bearing */
heading_and_bearing(plane_XO, plane_YO, plane_heading_deg,
                     ndb_theta_deg, radius, &length, conv);
p = insert(p, length, conv);
/* CALL TIMER */
timer(page_number, clock_pkg, alarm_pkg, &length, conv);
p = insert(p, length, conv);
SCREEN[p++] = 0x8E00:
SCREEN[p++] = OxFFFF;
times = 0:
error = 1;
while (error != 0)
```

```
times++;
error = SEND_SCREEN();
}

/* if (times != 1)
    printf("NV US times = %d\n", times); */
) /* end of nav_pg_dynamic */
```

```
SOURCE FILE:
                     ils_pg_s.c
    FUNCTION:
                     ils_pg_static()
    DESCRIPTION:
                     This routine generates the static
                     data for the ILS PAGE. It is only
                     called once when the SYSTEM SWITCH
                     is turned on.
    DOCUMENTATION
   FILES:
                     None.
   ARGUMENTS:
                     None.
   RETURN:
                   None.
   FUNCTIONS
   CALLED:
                     UPDATE_SCREEN()
                     string_gen()
                     insert()
                     arc_circ()
                   ·line()
   AUTHOR:
                   CHUCK ROBERTSON
   DATE CREATED:
                    16Feb87
                                Version 1.0
 ×
   REVISIONS:
                     None.
***********************
#include<stdlib.h>
#include<string.h>
#include<stdio.h>
#include "ilspg_xy.h"
#include "data_str.h"
void ils_pg_static()
```

```
unsigned short conv[300];
                                /* conversion array */
unsigned short address_ptr;
                                /* address pointer */
unsigned short line_char;
                                /* line characteristic */
int p;
                                /* conversion array
                                   index */
int length;
                                /* insertion length */
int 1;
                                /* loop counter */
int total_sent;
                                /* total words sent to
                                   vector memory
int error;
                               /* masm error return */
char final_string[30];
                               /* string arrays */
char cat_string[10];
double size;
                               /* character size */
void string_gen();
int insert();
void box();
void arc_circ();
void line():
size = 1.5;
                         /* set character size */
address_ptr = 0xC500;
                         /* point at 0x500 word in
                             vector memory
line_char = 0x7818;
                         /* set line characteristics */
p=0;
SCREEN[p++] = address_ptr;
SCREEN[p++] = line_char;
      /* PUT UP ALTITUDE */
      strcpy(final_string, "ALT: "):
      string_gen(final_string, altitude_XO, altitude_YO,
                 size, &length, conv);
      p = insert(p, length, conv);
      box(altitude_XO, altitude_YO, size, length-2,
          &length, conv);
      p = insert(p, length, conv);
      /* PUT UP DECISION HEIGHT */
      strcpy(final_string, "DH:");
      string_gen(final_string, mda_dh_XO, mda_dh_YO, size,
                 &length, conv);
      p = insert(p, length, conv);
      /* PUT UP DME */
     strcpy(final_string, "DME: ");
```

```
string_gen(final_string, dme_XO, dme_YO, size,
                  &length, conv);
      p = insert(p, length, conv);
      /* PUT UP STATIC ARC OF COMPASS */
      arc_circ(compass_arc_XO, compass_arc_YO,
                compass_arc_init_angle,
                compass_arc_end_angle,
                compass_arc_radius, &length, conv);
      p = insert(p, length, conv);
      /* PUT UP HEADING BOX AND POINTER */
      box(heading_box_XO, heading_box_YO, size,
          heading_box_no_char, &length, conv);
      p = insert(p, length, conv);
      line(heading_ptr_line1_XO, heading_ptr_line1_YO,
            heading_ptr_line1_X1, heading_ptr_line1_Y1,
            &length, conv);
      p = insert(p, length, conv);
      line(heading_ptr_line2_XO, heading_ptr_line2_YO,
            heading_ptr_line2_X1, heading_ptr_line2_Y1,
            &length, conv);
      p = insert(p, length, conv);
SCREEN[p++] = 0x8C00;
SCREEN[p++] = 0xCCOO:
SCREEN[p++] = 0x0000:
SCREEN[p++] = 0x8FFF:
SCREEN[p++] = OxFFFF;
error = 1:
while (error i= 0)
    error = SEND SCREEN( ):
/* end of ils pg static */
```

SOURCE FILE: ils_pg_d.c

FUNCTION: ils pg dynamic()

DESCRIPTION: This function generates all the

dynamic data for the ILS PAGE. It retrieves raw data from the data package and makes necessary conversions. Presently it is displaying the following:

1) altitude

2) decision height

3) plane heading 4) dme

5) ils cross hairs

6) runway

7) alarm conditions

DOCUMENTATION

¥

¥

¥

FILES: None.

ARGUMENTS: clock_pkg : pointer to clock package

'alarm_pkg : pointer to alarm package

RETURN: None.

FUNCTIONS

CALLED: GET_DATA_PACKAGE()

TOGGLE_ALARM_SWITCH()

UPDATE_SCREEN()

insert()

string_gen()

get_heading()

box()

zero_pad()

line()

runway() altitude()

dme()

zero pad() ils_compass()

```
×
                  airspeed()
 ×
                    timer()
 ¥
 ×
   AUTHOR:
                    CHUCK ROBERTSON
   DATE CREATED:
                    16Feb87 Version 1.0
   REVISIONS:
                    None.
***********************
#include "data str.h"
#include "ilspg_xy.h"
#include<stdlib.h>
#include<string.h>
#include<math.h>
#include<stdio.h>
void ils_pg_dynamic(clock_pkg, alarm_pkg)
CLOCK_PKG *clock pkg;
ALARM_PKG *alarm_pkg;
   unsigned short conv[3001; /* conversion array */
   int times;
                               /* request times */
   int
                               /* array index */
          p;
   int
         length;
                               /* insertion length */
        i;
   int
                               /* counter */
   int
        radix:
                               /* conversion */
   int
         error:
                               /* masm error return */
         page_number = 3;
   int
                               /* page being displayed */
   int
         plane_heading_deg;
                               /* plane heading */
       glideslope;
   int
                               /* glideslope plot
                                  coord */
   int
         course_deviation;
                               /* course_deviation
                                  coord */
   int
                               /* runway heading */
         runway_hdg_deg;
   int
        insert();
   void
         string gen():
   void
         runway();
   void box();
   void line():
   void get_heading();
   void altitude();
```

```
void
       dme();
void
       zero_pad();
void ils_compass();
void
       airspeed():
void timer();
int
       lower_X bound:
                               /* plot boundries */
int
       upper_X_bound;
       lower_Y_bound:
int
int
       upper Y bound:
int
       middle_X;
                               /* cross hair
                                  coordinates */
int
       middle Y;
int
       lower_pegout_bound;
                               /* pegout bounds */
int
       upper_pegout_bound;
int
       left_pegout_bound;
int
       right_pegout_bound;
unsigned char
                 peg_left;
                               /* sensitivity bounds */
unsigned char
                 peg_up;
unsigned char
                peg_right;
unsigned char
                peg_down;
double delta x;
                               /* plot divisions */
double delta_y;
double real_dme;
                               /* real dme value */
double MIN_runway_scale;
                              /* minimum runway scale */
double MAX_runway_scale;
                              /* maximum runway scale */
double MAX_plot_dme;
                              /* maximum plot dme */
double delta_runway_scale;
                              /* scale multiplier */
double runway_scale;
                              /* runway scale */
double crab_angle_deg;
                              /* crab angle of runway */
       final_string[20];
char
                              /* conversion string
                                 arrays */
char
      cat_string[20];
double size;
                              /* character size */
times = 0;
error = 1;
while (error != 0)
    times++;
    error = GET_DATA_PACKAGE( ):
 if (times != 1)
    printf("ILS DP %d\n", times); */
```

```
size = 1.5;
radix = 10:
 runway_hdg_deg = 280;
MIN_runway_scale = 1.0;
MAX_runway_scale = 5.0;
MAX_plot_dme = 15.0:
delta_runway_scale = (MAX_runway_scale -
                        MIN_runway scale) /
                        MAX_plot_dme;
p=0;
SCREEN[p++] = OxCCOO;
SCREEN(p++) = 0x7818:
                                       /* set line size */
/* CHECK AIRSPEED */
airspeed(page_number, clock_pkg, alarm_pkg,
          airspeed_X1, airspeed_Y1,
          &length, conv);
p = insert(p, length, conv);
/* PUT UP ALTITUDE */
altitude(clock_pkg, alarm_pkg, altitude_X1, altitude_Y1,
          &length, conv):
p = insert(p, length, conv);
/* PUT UP DECISION HEIGHT */
zero_pad((double)(clock_pkg->mda_dh),1000.0,6, "int",
         final_string);
string_gen(final_string, mda_dh_X1, mda_dh_Y1, size,
            &length, conv);
p = insert(p, length, conv);
/* PUT UP COMPASS AND HEADING */
get_heading(&plane_heading_deg);
zero_pad((double)(plane_heading_deg), 100.0,6,
         "int", final string);
string_gen(final_string, heading_XO, heading_YO, size,
           &length, conv);
p = insert(p, length, conv);
ils_compass(compass_arc_XO, compass_arc_YO,
            plane_heading_deg, compass_arc_radius,
            &length, conv);
p = insert(p, length, conv);
```

```
/* PUT UP DME */
    dme(clock_pkg, alarm_pkg, dme_X1, dme_Y1, &length, conv);
    p = insert(p, length, conv);
    real_dme = (double)((data_pkg.DME - 2.0) * 0.207);
    if(real_dme < 0.0)
       real_dme = - real_dme;
    /* PUT UP CDI */
      strcpy(final_string, "CDI:");
    itoa(data_pkg.COURSE_DEVIATION, cat_string, radix);
    strcat(final_string, cat_string);
    string_gen(final_string, 1500, 1850, size, &length, conv);
    p = insert(p, length, conv); */
    /* PUT UP GLDSLP */
/ *
      strcpy(final_string, "GLDSLP:");
    itoa(data_pkg.GLIDESLOPE, cat_string, radix);
   strcat(final_string, cat_string);
   string_gen(final_string, 1500, 1750, size, &length, conv);
   p = insert(p, length, conv); */
   /* IF PLANE IS CLOSE ENOUGH PLOT CROSSHAIRS */
   if(real_dme <= 20.0)
         /* CALCULATE RUNWAY SCALE */
         runway_scale = MAX_runway_scale -
                         real_dme * delta_runway_scale;
         /* CALCULATE BOUNDRIES FOR RUNWAY PLOT */
         lower_X_bound = (int)(75 * runway_scale);
         upper_X_bound = (int)(2048 - (75 *
                                runway_scale)):
         middle_X = (int)(lower_X_bound +
                          (upper_X_bound -
                           lower_X_bound) / 2.0);
         lower_Y_bound = (int)(50 * runway_scale);
         upper_Y_bound = (int)(1850 -
                               (150 * runway_scale));
         middle_Y = (int)(lower_Y_bound +
                          (upper_Y_bound -
```

```
lower_Y_bound) / 2.0):
/* PUT UP INITIAL INDICATOR CROSSHAIRS */
if(real_dme > MAX_plot_dme)
      line(middle_X, lower_Y_bound, middle_X,
           upper_Y_bound, &length, conv);
      p = insert(p, length, conv);
      line(lower_X_bound, middle_Y, upper_X_bound,
          middle_Y, &length, conv);
      p = insert(p,length,conv);
   }
/* IF PLANE IS CLOSE ENOUGH PLOT RUNWAY */
if(real_dme <= MAX_plot_dme)
   {
      /* ADD IN SENSITIVITY */
      if(real_dme >= 8.0)
         {
             peg_left = peg_up = 50;
             peg_right = peg_down = 205:
     else if(real_dme >= 7.0)
             peg_left = peg_up = 52;
             peg_right = peg_down = 202;
     else if(real_dme >= 6.0)
        {
            peg_left = peg_up = 62;
            peg_right = peg_down = 191;
     else if(real_dme >= 5.0)
            peg_left = peg_up = 75;
            peg_right = peg_down = 181;
     else if(real_dme >= 4.0)
            peg_left = peg_up = 81;
            peg_right = peg_down = 174;
     else if(real_dme >= 3.0)
```

```
{
        peg_left = peg_up = 87;
        peg_right = peg_down = 165;
else if(real_dme >= 2.0)
        peg_left = peg_up = 92;
       peg_right = peg_down = 158;
else if(real dme >= 1.0)
       peg_left = peg_up = 106:
       peg_right = peg_down = 147:
else if(real_dme >= 0.0)
       peg_left = peg_up = 117;
       peg_right = peg_down = 135;
   }
if(data_pkg.COURSE_DEVIATION < peg_left)
   data_pkg.COURSE_DEVIATION = peg_left;
if(data_pkg.COURSE_DEVIATION > peg_right)
   data_pkg.COURSE_DEVIATION = peg_right;
if(data_pkg.GLIDESLOPE < peg_up)</pre>
   data_pkg.GLIDESLOPE = peg_up;
if(data_pkg.GLIDESLOPE > peg_down)
   data_pkg.GLIDESLOPE = peg_down;
/* CALCULATE COORDINATES FOR RUNWAY */
delta_x = (double)(
               (upper_X_bound -
               lower X bound) /
              (peg_right - peg_left));
delta_y = (double)(
              (upper_Y_bound -
               lower_Y_bound) /
              (peg_down - peg_up));
course_deviation
          = lower_X_bound +
            (int)
            ((data_pkg.COURSE DEVIATION -
```

```
peg_left) * delta_x);
 glideslope
            = upper_Y_bound -
             (int)
              ((data_pkg.GLIDESLOPE - peg_up)
               * delta_y);
 /* PUT UP PEG-OUT BOUNDARIES */
 lower_pegout_bound = (int)(lower_Y_bound +
                              2 * delta_y);
upper_pegout_bound = (int)(upper_Y_bound -
                              2 * delta y);
left_pegout_bound = (int)(lower_X_bound +
                              2 * delta x);
right_pegout_bound = (int)(upper_X_bound -
                              2 * delta_x);
line(left_pegout_bound, lower_pegout_bound,
      left_pegout_bound, upper_pegout_bound,
      &length, conv);
p = insert(p, length, conv);
line(left_pegout_bound, upper_pegout_bound,
     right_pegout_bound, upper_pegout_bound,
     &length, conv):
p = insert(p, length, conv);
line(right_pegout_bound, upper_pegout_bound,
     right_pegout_bound, lower_pegout_bound,
     &length, conv);
p = insert(p,length,conv);
line(right_pegout_bound, lower_pegout_bound,
     left_pegout_bound, lower_pegout_bound,
     &length, conv);
p = insert(p, length, conv);
/* PUT UP APPROACH CROSSHAIRS */
line(middle_X,lower_pegout_bound,
     middle_X, upper_pegout_bound,
     &length, conv);
p = insert(p, length, conv);
```

```
line(left_pegout_bound, middle_Y,
                       right_pegout_bound, middle Y.
                       &length, conv):
                  p = insert(p, length, conv);
                  /* CALCULATE CRAB ANGLE */
                  crab_angle_deg = -(double)(
                                     (plane_heading_deg + 360)
                                      - (runway_hdg_deg
                                      % 360));
                  if(crab_angle_deg <= 360.0)
                     crab_angle_deg = crab_angle_deg + 360.0;
                 if(crab_angle_deg > 20.0)
                     crab_angle_deg = 20.0;
                 if(crab_angle_deg < -20.0)
                    crab_angle_deg = -20.0;
                 /* PLOT RUNWAY */
                 runway(course_deviation, glideslope,
                        crab_angle_deg, runway_scale, &length,
                        conv);
                 p = insert(p, length, conv);
              }
        }
    /* CHECK TIMER */
    timer(page_number, clock_pkg, alarm_pkg, &length, conv);
    p = insert(p, length, conv);
    SCREEN[p++] = 0x8E00;
    SCREEN[p++] = OxFFFF;
    times = 0;
    error = 1:
    while (error != 0)
         times++;
         error = SEND_SCREEN( );
/ *
      if (times != 1)
         printf("ILS US times = %d\n", times); */
```

} /* end of iIs_pg_dynamic */

APPENDIX D (CONT.)

GENERAL-PURPOSE TOOLS

```
¥
    SOURCE FILE:
                   insert.c
   FUNCTION:
                   insert()
   DESCRIPTION:
                   This function inserts the conversion
                   array into the screen array. The
                   length of the conversion array is
                   passed to the routine.
    DOCUMENTATION
    FILES:
                   None.
   ARGUMENTS:
                         : index for the SCREEN array
                   length : length of the conversion
                          array
                         : pointer to the conversion
                   Conv
                          array
   RETURN:
                   р
                         : index for the SCREEN array
   FUNCTIONS
   CALLED:
                  None.
 ¥
   AUTHOR:
                  CHUCK ROBERTSON
   DATE CREATED:
 ×
                  22Jan87
                              Version 1.0
   REVISIONS:
                 None.
insert(p, length, conv)
unsigned short p;
unsigned short length;
unsigned short conv[];
{
  int i;
                               /* loop counter */
```

```
extern unsigned short SCREEN[];  /* screen data array */
for(i=0;i < length ;i++)
          SCREEN[p++] = conv[i];
return (p);
} /* end of insert */</pre>
```

¥ SOURCE FILE: line.c FUNCTION: line() DESCRIPTION: This program generates the HP code to put up a line between two points on vector graphics display. DOCUMENTATION FILES: None. ARGUMENTS: XO, YO : beginning coordinates for the line X1. Y1 : ending coordinates for the line length ptr : pointer to the length of conv[] array Conv : pointer to converion array RETURN: None. FUNCTIONS CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 22Jan87 Version 1.0 REVISIONS: None. ************************* void line(XO, YO, X1, Y1, length_ptr, conv)

unsigned short XO, YO; unsigned short X1, Y1; unsigned short *length_ptr; unsigned short conv[];

SOURCE FILE: arc_circ.c

FUNCTION: arcd_circ()

DESCRIPTION: This routine generates the HP code to

put up an arc from 0 to 360 degrees with plot origin at center of arc.

DOCUMENTATION

¥

FILES: None.

ARGUMENTS: x_center : x plot coordinate

for arc center
y_center : y plot coordinate

for arc center start_angle_deg : initial angle end_angle_deg : final angle

end_angle_deg : final angle
radius : radius of arc
length ptr : pointer to leng

conv : pointer to conversion array

RETURN: None.

FUNCTIONS
CALLED: None.

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 25Jan87 Version 1.0

REVISIONS: None.

#include<stdio.h>

void arc_circ(x_center, y_center, start_angle_deg,

```
end angle deg, radius, length ptr, conv)
unsigned short conv[];
unsigned short *length_ptr;
int x_center, y_center;
int radius:
double start_angle_deg;
double end_angle_deg;
{
    int.
          X. Y:
                               /* plot coordinate
                                  variables
    int
          final_delta_cnt;
                               /* iteration number */
   int
          delta_count;
                              /* loop counter */
   int
          1;
                               /* loop counter */
   int
          index:
                              /* conversion index */
   double PI:
   double DEGREE_TO_RAD;
   double start angle rad:
                               /* beginning angle in
                                  radians
   double end_angle_rad;
                               /* ending angle in radians */
   double delta_theta_deg;
                               /* subinterval of total angle
                                  in degrees */
   double delta_theta_rad;
                              /* subinterval of total angle
                                  in radians */
   double theta;
                              /* calculation angle */
   double total_angle;
                              /* total arc angle */
   double screen_scale;
                              /* screen scale for X
                                 direction */
   PI = 3.14159265;
   screen scale = 9.5/12.5:
   DEGREE_TO_RAD = (2 * PI) / 360.0:
   index = 0:
       convert beginning and ending angles to radians */
   start_angle_rad = start_angle_deg * DEGREE_TO_RAD:
   end_angle_rad = end_angle_deg * DEGREE_TO_RAD;
   /* determine angle increment based on radius */
   if(radius <= 100)
     delta_theta_deg = 12.0;
  else if((radius > 100) && (radius <= 200))
     delta_theta_deg = 11.25;
  else if((radius > 200) && (radius <= 300))
     delta_theta_deg = 10.0;
```

else if((radius > 300) && (radius <= 400))

```
delta theta deg = 9.0:
 else if((radius > 400) && (radius <= 500))
    delta_theta_deg = 8.0;
 else if((radius > 500) && (radius <= 600))
    delta_theta_deg = 7.2:
 else if((radius > 600) && (radius <= 700))
    delta_theta_deg = 6.0:
 else if((radius > 800) && (radius <= 900))
    delta_theta_deg = 5.0;
 else if((radius > 900) && (radius <= 1000))
    delta_theta_deg = 4.0;
 else if((radius > 1000) && (radius <= 1200))
    delta_theta_deg = 3.0:
 else if((radius > 1200) && (radius <= 1400))
    delta_theta_deg = 2.0;
 else if((radius > 1400) && (radius <= 2000))
    delta_theta_deg = 1.0;
 else printf("INVALID RADIUS INPUTED\n"):
 /* convert to radians */
delta_theta_rad = delta_theta_deg * DEGREE_TO_RAD;
/* initialize calculation angle */
theta = start_angle_rad;
/* calculate total arc angle */
total_angle = end_angle_deg - start_angle_deg;
/* calculate iteration # */
final_delta_cnt = (int)(total_angle/delta_theta_deg);
/* calculate initial coordinates */
X = (int)(radius * cos(theta) * screen_scale);
Y = (int)(radius * sin(theta));
conv[index++] = 0x0000 | (X + x_center); /* plot x
                                             off
conv[index++] = 0x1000 ! (Y + y_center); /* plot y
                                            off
                                                    */
/* calculate rest of coordinates */
for(delta_count = 1; delta_count <= final_delta_cnt
    ; delta_count ++)
```

```
theta = theta + delta_theta_rad;

X = (int)(radius * cos(theta) * screen_scale);
Y = (int)(radius * sin(theta));

conv[index++] = 0x0000 | (X + x_center);
conv[index++] = 0x1800 | (Y + y_center);
}

*length_ptr = index;
} /* end arc_circ */
```

SOURCE FILE: str_gen.c FUNCTION: string_gen() DESCRIPTION: This routine generates the HP code to put a string at a certain location on the vector graphics display. DOCUMENTATION FILES: None. ARGUMENTS: XO. YO : coordinates for the string length_ptr : pointer to the length of conv[] array size : size of the character to be generated conv : pointer to converion array string : pointer to string array RETURN: None. FUNCTIONS CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 22Jan87 Version 1.0 REVISIONS: None.

Rone

#include<string.h>
#include<stdio.h>

void string_gen(string, XO, YO, size, length_ptr, conv)

```
unsigned short XO, YO;
 unsigned short *length_ptr;
 unsigned short conv[]:
 double size:
 char string[]:
    int i;
                      /* loop counter */
   /* convert coordinates to hp code */
   conv[0] = 0x0000 ! XO; /* plot x off */
   conv[1] = 0x1000 | YO; /* plot y off */
   /* convert size to hp code */
   if(size == 1.0)
      conv[2] = 0x4100:
   else if(size == 1.5)
      conv[2] = 0x4900:
   else if(size == 2.0)
      conv[2] = 0x5100;
   else if(size == 2.5)
      conv[2] = 0x5900:
   else printf("INVALID SIZE INPUTED\n");
   /* find string length */
   *length_ptr = strlen(string);
   /* add 2 to length pointer for XO and YO code */
   *length_ptr = *length_ptr + 2;
   /* convert string to hp code */
  conv[2] = conv[2] | string[0];
  for(i=3; i < *length_ptr; i++)
     conv[i] = string[i-2] | 0x4000;
 /* end of string gen */
}
```

SOURCE FILE: box.c FUNCTION: box() DESCRIPTION: This function generates the code to put up a box on the vector graphics display around a given word at a given location. DOCUMENTATION FILES: None. ARGUMENTS: XO, YO : plot coordinates for plane size : size of the characters no_char : number of characters to be boxed length_ptr : pointer to length of conv[] array CONV : pointer to conversion array RETURN: None. FUNCTIONS CALLED: None. AUTHOR: CHUCK ROBERTSON

DATE CREATED:

25Jan87

Version 1.0

REVISIONS:

None.

*********************** #include<stdio.h>

void box(XO, YO, size, no_char, length_ptr, conv)

```
unsigned short conv[];
unsigned short *length_ptr:
unsigned short XO, YO;
unsigned short no_char;
double size;
  int X, Y;
                     /* coordinate variables */
  int width;
                     /* character width */
  int height;
                     /* character height */
  if(size == 1.0)
      { width = 36:
       height = 36; }
  else if(size == 1.5)
     { width = 54;
       height = 54; }
  else if(size == 2.0)
     { width = 72:
       height = 72; }
  else if(size == 2.5)
     { width = 90:
       height = 90: }
  else printf("invalid size inputed\n");
  /* move to initial coordinates and generate box
     coordinates
  X = X0 - 10;
  Y = Y0 - 10;
  conv[0] = 0x0000- | X; /* plot x off */
  conv[1] = 0x1000 | Y; /* plot y off */
 X = X + 20 + no_char * width;
 conv[2] = 0x0000 | X;
 conv[3] = 0x1800 | Y;
 Y = Y + 20 + height;
 conv[4] = 0x00000 | X;
 conv[5] = 0x1800 | Y;
 X = X - 20 - no_char * width;
 conv[6] = 0x0000 | X;
 conv[7] = 0x1800 | Y:
 Y = Y - 20 - height;
 conv[8] = 0x0000 | X;
 conv[9] = 0x1800 | Y;
 *length_ptr = 10;
```

} /* end of box */

SOURCE FILE: zero_pad.c

FUNCTION: zero_pad()

DESCRIPTION: This function zero pads a given

integer or double given the number to

be padded, the pad limit, the

precision, and type of number to be padded.

padded

DOCUMENTATION

¥

FILES: None.

ARGUMENTS: pad_value : value to be padded

pad_limit : zero padding limit precision : precision desired

type_string : type of number to be

padded

pad_string : pointer to conversion

array

RETURN: None.

FUNCTIONS

CALLED: strcpy()

strcmp()
strcat()
itoa()
gcvt()
printf()

AUTHOR: Chuck Robertson

DATE CREATED: 25Feb87 Version 1.0

REVISIONS: None.

```
#include <stdlib.h>
#include <string.h>
#include <stdio.h>
void zero_pad(pad_value, pad_limit, precision, type_string,
                   pad_string)
char pad_string[20];
double pad value;
double pad limit:
int precision;
char type_string[];
   char
          cat_string[19];
        radix = 10:
   int
   /* start pad_string with EOL for strcat() to detect */
   pad_string[0] = '\0':
   /* take care of negative case */
   if(pad_value < 0.0)
         strcpy(pad_string, "-");
         pad_value = pad_value * -1.0:
      3
   /* determine if padding is necessary */
   if( ((pad_limit / 10.0) <= pad_value) &&
       (pad_value < pad_limit) )
      strcpy(pad_string, "0");
   else if( ((pad_limit / 100.0) <= pad_value ) &&
            (pad_value < (pad_limit / 10.0)) )
      strcpy(pad_string, "00");
   else if( ((pad_limit / 1000.0) <= pad_value ) &&
            (pad_value < (pad_limit / 100.0)) )
      strcpy(pad_string, "000");
   if( (pad_value == 0.0) && (pad_limit >= 1000.0) )
      strcpy(pad_string, "000");
   else if( (pad_value == 0.0) && (pad_limit >= 100.0) )
      strcpy(pad_string, "00");
  else if( (pad_value == 0.0) && (pad_limit >= 10.0) )
      strcpy(pad_string, "O");
  /* convert pad_value to a character string */
```

```
if( strcmp(type_string, "int") == 0)
   itoa((int)(pad_value), cat_string, radix);
else if( strcmp(type_string, "double") == 0)
   gcvt(pad_value, precision, cat_string);
else
   printf("ILLEGAL TYPE INPUTED IN ZERO PAD\n");

/* append pad_value to pad_string */
   strcat(pad_string, cat_string);
} /* end of zero_pad */
```

/***********************

SOURCE FILE: altitude.c

FUNCTION: altitude()

DESCRIPTION: This routine retrieves the altitude

value from the data package and converts it into feet. It then converts this value into a zero-

padded plot string.

DOCUMENTATION

×

FILES: None.

ARGUMENTS: clock_pkg : pointer to clock

package

alarm_pkg : pointer to alarm

XO, YO package

XO, YO : plot coordinates length_ptr : pointer to length of

conversion array

altitude_conv : pointer to conversion

array

RETURN: None.

FUNCTIONS

CALLED: zero pad()

string_gen()

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 23Mar87 Version 1.0

REVISIONS: 04Apr87 Version 1.1

Added alarm capability.

```
#include <string.h>
#include <stdio.h>
#include "data str.h"
                        /* data package structure
                            declaration. */
void altitude(clock_pkg, alarm_pkg, XO, YO, length_ptr,
              altitude conv)
CLOCK_PKG *clock pkg;
ALARM PKG *alarm pkg;
int XO. YO:
int *length_ptr;
unsigned short altitude_conv[];
{
   unsigned short string_conv[30];
   int altitude;
   int string_length;
   int index = 0;
   int i:
   char final_string[20];
   double size1 5 = 1.5;
   void string gen();
   void zero pad();
   altitude = (int)((data_pkg.ALTITUDE - 15) * 29.7);
   if(altitude < 0)
      altitude = 0:
   zero_pad((double)(altitude),1000.0,6, "int",
            final_string):
   string_gen(final_string, XO, YO, size1_5, &string_length,
              string conv):
   for(i=0;i<string_length;i++)</pre>
       altitude_conv[index + i] = string_conv[i];
   index = index + string_length;
   /* CHECK MDA_DH ALARM CONDITION */
   /* IF ALTITUDE IS APPROACHING MDA_DH SOUND ALARM */
   if((alarm_pkg->mda_dh_alarm_flag == ALARM_OFF)
    && (altitude <= (clock_pkg->mda_dh + 50))
         && (alarm_pkg->mda_dh_enable_flag == ENABLED))
```

```
alarm_pkg->mda_dh_alarm_flag = ALARM_ON;
    if(alarm_pkg->alarm_status_flag == ALARM_OFF)
       TOGGLE_ALARM_SWITCH();
       alarm_pkg->alarm_status_flag = ALARM ON:
    )
 if((alarm_pkg->mda_dh_alarm_flag == ALARM_ON)
   && (altitude > (clock_pkg->mda_dh + 50))
   && (alarm_pkg->mda_dh_enable_flag == ENABLED))
    alarm_pkg->mda_dh_alarm_flag = ALARM_OFF:
    if((alarm_pkg->alarm_status_flag == ALARM_ON) &&
      (alarm_pkg->assigned_altitude_alarm_flag ==
      ALARM OFF)
     && (alarm_pkg->airspeed_alarm_flag == ALARM_OFF)
     && (alarm_pkg->time_out_alarm_flag == ALARM_OFF))
      TOGGLE_ALARM_SWITCH():
      alarm_pkg->alarm_status_flag = ALARM_OFF;
   )
/* IF ALARM IS ON DISPLAY MDA/DH APPROACHING MESSAGE */
if((alarm_pkg->mda_dh_alarm_flag == ALARM ON)
  && (alarm_pkg->mda_dh_enable_flag == ENABLED))
      strcpy(final_string, ">>>MDA/DH<<<");
      string_gen(final_string, 680, 80,
                 size1_5, &string_length, string_conv);
      for(i=0;i<string_length;i++)
          altitude_conv[index + il = string_conv[i];
      index = index + string_length;
   3
/* CHECK ASSIGNED ALTITUDE ALARM CONDITION */
/* IF ALTITUDE IS EXCEEDING ASSIGNED ALTITUDE LIMIT
  SOUND ALARM
if( (alarm_pkg->assigned_altitude_alarm_flag ==
   ALARM OFF)
 && !( (altitude <= (clock_pkg->assigned_altitude +
```

```
100))
   && (altitude >= (clock_pkg->assigned_altitude -
       100))
   && (alarm_pkg->assigned_altitude_enable flag ==
     ENABLED) )
   alarm_pkg->assigned_altitude_alarm_flag = ALARM_ON;
   if(alarm_pkg->alarm_status_flag == ALARM_OFF)
      TOGGLE_ALARM_SWITCH();
      alarm_pkg->alarm_status_flag = ALARM_ON;
    }
if( (alarm_pkg->assigned_altitude_alarm_flag ==
    ALARM ON)
  && ( (altitude <= (clock_pkg->assigned_altitude +
    100))
  && (altitude >= (clock_pkg->assigned_altitude -
    100)))
  && (alarm_pkg->assigned_altitude_enable_flag ==
    ENABLED) )
   alarm_pkg->assigned_altitude_alarm_flag = ALARM_OFF;
   if((alarm_pkg->alarm_status_flag == ALARM_ON) &&
     (alarm_pkg->mda_dh_alarm_flag == ALARM_OFF)
     && (alarm_pkg->airspeed_alarm_flag == ALARM_OFF)
     && (alarm_pkg->time_out_alarm_flag == ALARM_OFF))
      TOGGLE_ALARM_SWITCH();
      alarm_pkg->alarm_status_flag = ALARM_OFF;
   )
/* IF ALARM IS ON DISPLAY ASSIGNED ALTITUDE MESSAGE */
if((alarm_pkg->assigned_altitude_alarm_flag == ALARM_ON)
 && (alarm_pkg->assigned_altitude_enable_flag ==
   ENABLED))
  {
     strcpy(final_string, ">>ASSIGNED<<");
     string_gen(final_string, 680, 10,
                size1_5, &string_length, string_conv);
     for(i=0;i<string_length;i++)
         altitude_conv[index + i] = string_conv[i];
     index = index + string_length;
```

```
*length_ptr = index;
} /* end altitude */
```

}

SOURCE FILE: dme.c

FUNCTION: dme()

DESCRIPTION: This routine retrieves the dme value

from the data package and converts it into nautical miles. It then converts this value into a zero-padded plot

string.

DOCUMENTATION

FILES: None.

ARGUMENTS: clock_pkg : pointer to clock

package

alarm_pkg : pointer to alarm

package

XO, YO : plot coordinates length_ptr : pointer to length of

conversion array dme_conv : pointer to conversion

array

RETURN: None.

FUNCTIONS

CALLED: zero_pad()

string_gen()

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 23Mar87 Version 1.0

REVISIONS: None.

#define ALARM_OFF 0 #define ALARM_ON

```
#include <stdlib.h>
#include <string.h>
#include <stdio.h>
#include "data_str.h"
                        /* data package structure
                            declaration.
extern int TOGGLE_ALARM_SWITCH( );
void dme(clock_pkg, alarm_pkg, XO, YO, length_ptr, dme_conv)
CLOCK_PKG *clock_pkg;
ALARM_PKG *alarm_pkg;
int XO, YO;
int *length_ptr;
unsigned short dme_conv[];
(
    int dme;
    int length:
    char final_string[20];
    double size = 1.5;
    void string_gen();
    void zero_pad();
    dme = (int)((data_pkg.DME - 2.0) * 0.207);
    zero_pad((double)(dme), 10.0, 6, "int", final_string);
    string_gen(final_string, XO, YO, size, &length, dme_conv);
    *length_ptr = length;
} /* end dme */
```

SOURCE FILE: airspeed.c

FUNCTION: airspeed()

DESCRIPTION: This routine retrieves the airspeed

value from the data package and converts it into knots. It then converts this value into a zero-

padded plot string.

DOCUMENTATION

FILES: None.

ARGUMENTS: page_number : page being displayed

clock_pkg : pointer to clock

package
alarm_pkg : pointer to alarm

package airspeed_XO : plot coord.

airspeed_YO : plot coord.
length ptr : pointer to 1

length_ptr : pointer to length of conversion array

airspeed_conv : pointer to conversion

array

RETURN: None.

FUNCTIONS

CALLED: zero_pad()

string_gen() strcpy()

AUTHOR:

* * *

¥

¥

CHUCK ROBERTSON

DATE CREATED: 23Mar87 Version 1.0

REVISIONS: 04Apr87 Version 1.1

Added alarm capability.

```
¥
#include <stdlib.h>
#include <string.h>
#include <stdio.h>
#include "data_str.h" /* data package structure
                          declaration.
void airspeed(page_number, clock_pkg, alarm_pkg,
             airspeed_XO, airspeed_YO,
             length_ptr, airspeed_conv)
int page_number;
CLOCK_PKG *clock_pkg;
ALARM_PKG *alarm_pkg;
int airspeed_XO, airspeed_YO;
int *length_ptr;
unsigned short airspeed conv[]:
{
   unsigned short string_conv[30];
   int airspeed;
   int string_length;
   int index = 0;
   int i:
   char final_string[20];
   double size1_5 = 1.5;
   void string gen();
   void zero_pad();
   airspeed = (int)((data_pkg.AIRSPEED + 60) * 0.8427);
   if(airspeed <= 51)
      airspeed = 0;
   /* DISPLAY AIRSPEED ON DATA PAGE AND NAV PAGE */
   if(page_number != 3)
     {
        zero_pad((double)(airspeed),100.0,6,"int",
                 final string):
        string_gen(final_string, airspeed_XO, airspeed_YO,
                  size1_5, &string_length, string_conv);
       for(i=0;i<string_length;i++)</pre>
           airspeed_convlindex + il = string_convlil;
       index = index + string_length;
```

```
/* IF AIRSPEED IS AT STALL SPEED SOUND ALARM */
     if((alarm_pkg->airspeed_alarm_flag == ALARM_OFF)
       && (airspeed < 70))
       alarm_pkg->airspeed_alarm_flag = ALARM_ON;
        if(alarm pkg->alarm_status_flag == ALARM_OFF)
           TOGGLE_ALARM_SWITCH();
           alarm pkg->alarm_status_flag = ALARM_ON;
        3
     if((alarm_pkg->airspeed_alarm_flag == ALARM_ON)
       && (airspeed > 75))
       alarm_pkg->airspeed_alarm_flag = ALARM_OFF:
       if((alarm_pkg->alarm_status_flag == ALARM ON) &&
          (alarm_pkg->assigned_altitude_alarm_flag ==
          ALARM_OFF)
         && (alarm_pkg->mda_dh_alarm_flag == ALARM_OFF)
         && (alarm_pkg->time_out_alarm_flag == ALARM_OFF))
          TOGGLE_ALARM SWITCH():
          alarm_pkg->alarm_status_flag = ALARM_OFF;
       }
    /* IF ALARM IS ON DISPLAY IMPENDING STALL MESSAGE */
    if(alarm_pkg->airspeed_alarm_flag == ALARM_ON)
       {
          strcpy(final_string, ">>>STALL<<<");
          string_gen(final_string, 1365, 80,
                     size1_5, &string_length, string_conv);
          for(i=0;i<string_length;i++)</pre>
              airspeed_conv[index + i] = string_conv[i]:
          index = index + string_length;
       }
   *length_ptr = index;
} /* end airspeed */
```

}

SOURCE FILE: arrow.c

FUNCTION: arrow()

DESCRIPTION: This routine generates the HP code to

put up the head of an arrow for a given radius, location, angle, and direction. The direction determines whether or not the arrow head points "to" or "from" the plot coordinates.

DOCUMENTATION

FILES: None.

ARGUMENTS: x_center : x plot coordinate

y_center : y plot coordinate angle_deg : plot angle in

degrees

radius : length of arrow dir_string : direction string length_ptr : pointer to length of

conv : pointer to conversion array

RETURN: None.

FUNCTIONS

CALLED: None.

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 15Feb87 Version 1.0

REVISIONS: None.

#include<math.h>
#include<string.h>

```
#include<stdio.h>
void arrow(x_center, y_center, angle_deg, radius, dir_string,
           length_ptr.conv)
unsigned short conv():
unsigned short *length_ptr;
int x_center, y_center;
double radius:
double angle_deg;
char dir_string[];
   int
         х, у;
                                 /* plot coordinates */
    int
         arrow_head_end_X;
                                 /* arrow head end coord */
    int
         arrow_head_end_Y;
         arrow_head_bgn_X;
   int
                                 /* arrow head base coord */
         arrow_head_bgn_Y;
   int
   int
         disp:
                                 /* plot displacement */
   double PI;
   double angle_rad;
                                 /* plot angle in radians */
   double screen_scale;
   PI = 3.14159265;
   screen_scale = 9.5/12.5;
       convert angle to radians */
   angle_rad = angle_deg * (2 * PI) / 360.0;
   /* figure displacement for arrow direction */
   if( strcmp(dir_string, "to") == 0 )
      disp = -50;
   else if( strcmp(dir_string, "from") == 0 )
      disp = 50;
   else
      printf("illegal arrow direction inputed\n");
   /* calculate coordinates */
   arrow_head_end_X = (int)(radius * cos(angle_rad)
                            * screen_scale);
  arrow_head_end_Y = (int)(radius * sin(angle_rad));
  conv[0] = 0x0000 | (arrow_head_end_X + x_center);
  conv[1] = 0x1000 | (arrow_head_end_Y + y_center);
  arrow_head_bgn_X = (int)((radius + disp)
                             * cos(angle_rad)
```

SOURCE FILE: heading.c FUNCTION: get heading() DESCRIPTION: This routine gets the raw heading from the data package and uses a lookup table along with interpolation to convert the raw heading into the actual plane heading. DOCUMENTATION FILES: None. ARGUMENTS: plane_heading_deg_ptr : pointer to the heading variable in calling routine RETURN: None. **FUNCTIONS** CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 16Feb87 Version 1.0 REVISIONS: None. ****************** #include "data_str.h" #include<stdlib.h> typedef struct { double a_to_d; double value; } CONVERSION; /* lookup table for heading conversion */ CONVERSION heading[15] = {

```
{0.0,99.0},
                              {6.0,90.0},
                              {30.0,60.0},
                              (50.0,30.0),
                              {73.0,0.0},
                              (94.0, 330.0),
                              {115.0,300.0},
                              (137.0, 270.0),
                              {158.0,240.0},
                              (180.0, 210.0).
                              {201.0, 180.0},
                              {223.0, 150.0},
                              {245.0, 120.0},
                              (251.0, 110.0),
                              (255.0, 100.0)
                           };
void get_heading(plane_heading_deg_ptr)
int *plane_heading_deg_ptr;
    int i;
                          /* loop counter */
    double raw_heading;
    /* GET RAW HEADING FROM A/D AND CONVERT
       TO DEGREES USING LOOKUP TABLE
    raw_heading = (double)(data_pkg.COMPASS);
    if( (raw_heading > 73.0) && (raw_heading <= 94.0))
       heading[4]. value = 360.0;
    else
       heading[4].value = 0.0;
    for(i = 0; i < 14; i++)
       if( (raw_heading >= heading[i].a_to_d) &&
           (raw_heading <= heading[i + 1].a_to_d) )</pre>
             /* INTERPOLATE */
             *plane_heading_deg_ptr
                      (int)
                      ( (raw_heading - heading[i].a_to_d) /
                      (heading[i+1].a to d -
                       heading[i].a_to_d) *
                      (heading[i+1].value -
                       heading[i].value) +
                       heading[i].value);
             break;
```

} /* end of get_heading */

}

/*********************

SOURCE FILE: timer.c

FUNCTION: timer()

DESCRIPTION: This function implements the TIMER and

TIME_OUT portion of the DATA PAGE. The function keeps track of the timer value and sounds an alarm when the

time-out is reached.

DOCUMENTATION

×

FILES: None.

ARGUMENTS: page_number: page being displayed

clock_pkg : pointer to clock package
alarm_pkg : pointer to alarm package
length_ptr : point to length of conv[]

array

timer_conv : pointer to conversion

array

RETURN: None.

FUNCTIONS

CALLED: string_gen()

zero_pad()

AUTHOR: Chuck Robertson

DATE CREATED: 26Feb87 Version 1.0

REVISIONS: 04Apr87 Version 1.1

Added alarm capability.

#include <string.h>

#include <stdio.h>

```
#include "data_str.h"
                        /* data package structure
                           declaration.
#include "datpg_xy.h"
void timer(page_number, clock_pkg, alarm_pkg, length_ptr,
           timer conv)
CLOCK_PKG *clock_pkg;
ALARM_PKG *alarm_pkg;
unsigned short timer_conv[];
int *length_ptr;
int page_number;
   static int real_time_start_sec;
   static int total_time_out_sec;
   unsigned short string_conv[30];
   int real_time_present_sec;
   int total_timer_present_sec;
   int string_length;
   int i:
   int index = 0;
   double size1_5 = 1.5;
   char final_string[20], cat_string[5];
   void string gen():
   void zero_pad();
   /* SET TIMER */
   if(clock_pkg->timer_operation_flag == SET_TIMER)
         /* turn off timer */
         clock_pkg->timer_status_flag = TIMER_OFF:
         /* clear operation flag */
         clock_pkg->timer_operation_flag = NULL_TIMER;
         if(alarm_pkg->time_out_alarm_flag == ALARM_ON)
            1
               alarm_pkg->time_out_alarm_flag = ALARM_OFF;
               if((alarm_pkg->alarm_status_flag ==
                  ALARM ON) &&
                 (alarm_pkg->assigned_altitude_alarm_flag
```

```
== ALARM OFF)
               && (alarm_pkg->mda_dh_alarm_flag ==
                    ALARM_OFF)
               == alarm_pkg->airspeed_alarm_flag ==
                   ALARM OFF))
               {
                  TOGGLE_ALARM_SWITCH();
                  alarm_pkg->alarm_status_flag =
                                              ALARM OFF:
               }
          ١
    }
 /* DETERMINE RESET CONDITION */
 if(clock_pkg->timer_operation_flag == RESET_TIMER)
       clock_pkg->timer_min = 0;
       clock pkg->timer_sec = 0;
       clock_pkg->timer_status_flag = TIMER_OFF;
      clock_pkg->timer_operation_flag = NULL_TIMER;
      if(alarm_pkg->time_out_alarm_flag == ALARM_ON)
            alarm_pkg->time_out_alarm_flag = ALARM_OFF;
            if((alarm_pkg->alarm_status_flag ==
                ALARM ON) &&
               (alarm_pkg->assigned_altitude_alarm_flag
                == ALARM_OFF)
              && (alarm_pkg->mda_dh_alarm_flag ==
                   ALARM_OFF)
              == (alarm_pkg->airspeed_alarm_flag ==
                  ALARM_OFF))
                 TOGGLE_ALARM_SWITCH():
                 alarm_pkg->alarm_status_flag =
                                             ALARM OFF:
              }
         )
/* START CLOCK */
if(clock_pkg->timer_operation_flag == START_TIMER)
      total_time_out_sec = ((60 *
                             clock_pkg->time_out_min) +
```

```
clock_pkg->time_out_sec);
       real_time_start_sec = (int)
                              ((3600 * data_pkg.HOURS) +
                               (60 * data_pkg.MINUTES) +
                               data_pkg.SECONDS);
       clock_pkg->timer_status_flag = TIMER_ON;
       clock_pkg->timer_operation_flag = NULL_TIMER;
       if(alarm_pkg->time_out_alarm_flag == ALARM_ON)
             alarm_pkg->time_out_alarm_flag = ALARM_OFF;
             if((alarm_pkg->alarm_status_flag ==
                 ALARM ON) &&
               (alarm_pkg->assigned_altitude_alarm_flag
                == ALARM_OFF)
               && (alarm_pkg->mda_dh_alarm_flag ==
                   ALARM OFF)
              && (alarm_pkg->airspeed_alarm_flag ==
                  ALARM OFF))
                 TOGGLE_ALARM_SWITCH();
                 alarm_pkg->alarm_status_flag =
                                             ALARM_OFF:
              3
         }
   }
/* CALCULATE TIMER VALUE AND DISPLAY */
if(clock_pkg->timer_status_flag == TIMER_ON)
      real_time_present_sec = (int)
                               ((3600 * data_pkg.HOURS) +
                                (60 * data_pkg.MINUTES) +
                               data_pkg.SECONDS);
      if(real_time_present_sec < real_time_start_sec)</pre>
         real_time_present_sec = real_time_present_sec +
                                  (24 * 3600):
     total_timer_present_sec = real_time_present_sec -
                                real_time_start_sec;
     /* convert present timer value into minutes and
        sec
                                                       * /
```

```
clock_pkg->timer_sec = total_timer_present_sec
                                % 60:
       clock_pkg->timer_min = ((total_timer_present_sec -
                                clock_pkg->timer_sec)
                                / 60):
       /* check for time_out alarm */
       if((total_timer_present_sec >= total_time_out_sec)
          && (alarm_pkg->time_out_alarm_flag ==
              ALARM OFF))
          {
             alarm_pkg->time_out_alarm_flag = ALARM_ON;
             if(alarm_pkg->alarm_status_flag ==
                ALARM OFF)
               TOGGLE_ALARM_SWITCH();
               alarm_pkg->alarm_status_flag = ALARM_ON;
          }
      /* is timer done ? */
      if(total_timer_present_sec == 3600)
         clock_pkg->timer_status_flag = TIMER_OFF:
   ¥
/* IF ALARM IS ON DISPLAY TIME-OUT MESSAGE */
if(alarm_pkg->time_out_alarm_flag == ALARM_ON)
      strcpy(final_string, ">>TIME-OUT<<");
      string_gen(final_string, 1365, 10,
                 size1_5, &string_length, string_conv);
      for(i=0;i<string_length;i++)
          timer_conv[index + i] = string_conv[i];
      index = index + string_length;
   }
/* DISPLAY TIMER AND TIME-OUT VALUES IF PAGE
   BEING DISPLAYED IS THE DATA PAGE
if(page_number == 1)
     /* DISPLAY TIMER VALUE */
```

```
zero pad((double)(clock_pkg->timer_min), 10.0, 6,
                 "int", final string):
       strcat(final_string, ":");
       zero_pad((double)(clock_pkg->timer_sec), 10.0,6,
               "int", cat_string);
       strcat(final_string, cat_string);
       string_gen(final_string,timer_X1,timer_Y1,size1_5,
                  &string_length, string_conv);
       for(i=0;i<string_length;i++)</pre>
           timer_conv[index + i] = string_conv[i];
       index = index + string_length;
       /* DISPLAY TIME-OUT VALUE */
       zero_pad((double)(clock_pkg->time_out_min), 10.0,6,
                "int", final_string);
      strcat(final_string, ":");
      zero_pad((double)(clock_pkg->time_out_sec), 10.0,6,
               "int", cat_string);
      strcat(final_string, cat_string);
      string_gen(final_string, time_out_X1, time_out_Y1,
                  size1_5, &string_length, string_conv);
      for(i=0;i<string_length;i++)
         timer_conv[index + i] = string_conv[i];
      index = index + string_length;
   3
 /* send back length of conversion array */
 *length_ptr = index;
/* end of timer */
```

APPENDIX D (CONT.)

PAGE-SPECIFIC TOOLS

SOURCE FILE: time_gen.c

FUNCTION: time_string_gen()

DESCRIPTION: Generates the string depicting the

time as the values of hours, minutes, and seconds of DATA_PKG show. Spaces and zeros are inserted in the

appropriate spots so that the length

of the final string is always

constant.

DOCUMENTATION

FILES: None.

ARGUMENTS: None.

RETURN: char *

pointer to the string named time_string.

FUNCTIONS
CALLED: None.

AUTHOR: Dave Gruenbacher

DATE CREATED: 07Feb87 Version 1.0

REVISIONS: None.

#include <stdlib.h>
#include <string.h>

#include <stdio.h>

#include "data_str.h" /* data package structure declaration. *

time_string_gen(time_string)
char time_string[9];

```
{
     char temp_string[3];
     char *strcat(), *itoa();
     int
           radix = 10:
     /* start time_string with EOL for strcat() to
         detect.
     time_string[0] = '\0':
     /* convert data_pkg.HOURS to a character
     itoa(data_pkg. HOURS, temp_string, radix);
     /* if data_pkg.HOURS is less then 10, then insert */
     /* a blank where the tens digit would be.
     if (data_pkg. HOURS < 10)
         strcat(time_string, " ");
    /* append hours and a colon to time_string. */
    strcat(time_string, temp_string);
    strcat(time_string, ":");
    /* if data_pkg.MINUTES is less then 10, insert */
    /* a zero where the tens digit would be. */
    if (data_pkg.MINUTES <10)
        strcat(time_string, "0");
    /* convert data_pkg.MINUTES to a character
       string.
    itoa(data_pkg.MINUTES, temp_string, radix);
    /* append minutes and a colon to time_string. */
    strcat(time_string, temp_string);
    strcat(time_string, ":");
    /* if data_pkg.SECONDS is less then 10, insert a */
    /* zero where the tens digit would be.
    if (data_pkg.SECONDS < 10)
        strcat(time_string, "O");
    /* convert data_pkg.SECONDS to a character string. */
    itoa(data_pkg.SECONDS, temp_string, radix);
    /* append seconds and a colon to time_string. */
  strcat(time_string, temp_string);
    /* return pointer to time_string to calling routine. */
   return:
} /* time_string_gen */
```

```
SOURCE FILE:
                    clim_box.c
    FUNCTION:
                    clim_box()
    DESCRIPTION:
                    This program generates the HP code to
                    put up the climb indicator box for the
                    DATA PAGE on the vector graphics
                    display.
    DOCUMENTATION
    FILES:
                    None.
    ARGUMENTS:
                    XO, YO
                              : plot coordinates for the
                                box
                    length_ptr
                              : pointer to the length of
                                conv() array
                    CONV
                              : pointer to converion
                                array
   RETURN:
                    None.
   FUNCTIONS
   CALLED:
                   None.
   AUTHOR:
                   CHUCK ROBERTSON
   DATE CREATED:
                   27Jan87
                               Version 1.0
   REVISIONS:
                  None.
*************************
void climb_box(XO, YO, length_ptr, conv)
unsigned short XO, YO;
unsigned short *length_ptr;
unsigned short conv[];
  int i;
                         /* loop counter */
```

```
int X, Y;
                              /* plot coordinate variables */
   /* move to initial coordinates and generate
      box coordinates
   X = XO:
   Y = YO;
   conv[0] = 0x0000 | X; /* plot x off */
   conv[1] = 0x1000 | Y; /* plot y off */
   X = X - 60:
   Y = Y + 315;
   conv[2] = 0x0000 | X;
   conv[3] = 0x1800 | Y:
   X = X - 80;
   conv[4] = 0x0000 | X;
   conv[5] = 0x1800 | Y:
   Y = Y - 630:
   conv[6] = 0x00000 | X;
   conv[7] = 0x1800 | Y;
   X = X + 80:
   conv[8] = 0x0000 | X;
   conv[9] = 0x1800 | Y:
   X = X + 60;
   Y = Y + 315:
   conv[10] = 0x0000 | X;
   conv[11] = 0x1800 | Y:
   *length_ptr = 12;
} /* end of clim_box */
```

*********** SOURCE FILE: clim hsh.c FUNCTION: clim hsh() DESCRIPTION: This routine generates the HP code to put up the hash marks of the climb indicator of the DATA PAGE. DOCUMENTATION FILES: None. ARGUMENTS: x_center : x plot coordinate for hash arc center y_center : y plot coordinate for hash arc center start_angle_deg : initial hash angle end_angle_deg : final hash angle radius : radius of hash arc length_ptr : pointer to length of conv[] array conv : pointer to conversion array RETURN: None. **FUNCTIONS** CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 25Jan87 Version 1.0 REVISIONS: None.

```
unsigned short conv[];
unsigned short *length ptr;
int x_center, y_center;
int radius:
double start_angle_deg;
double end angle deg:
    int X, Y;
                              /* plot coordinate
                                  variables
          final_delta_cnt;
   int
                              /* iteration number */
   int
          delta_count;
                              /* loop counter */
   int
          1:
                              /* loop counter */
   int
          index:
                              /* conversion index */
   double PI:
   double DEGREE TO RAD:
   double start_angle_rad;
                              /* beginning angle in
                                  radians
   double end_angle rad:
                               /* ending angle in
                                  radians
   double delta theta deg:
                              /* subinterval of total
                                  angle in degrees
   double delta_theta_rad;
                              /* subinterval of total
                                  angle in radians
                                                       */
   double theta;
                              /* calculation angle */
   double total_angle:
                              /* total arc angle */
   double screen_scale;
                              /* screen scale for X
                                 direction
   PI = 3.14159265;
   screen_scale = 9.5/12.5;
   DEGREE_TO_RAD = (2 * PI) / 360.0;
   index = 0:
       convert beginning and ending angles to radians */
   start_angle_rad = start_angle_deg * DEGREE_TO_RAD;
   end_angle_rad = end_angle_deg * DEGREE_TO_RAD;
   /* initialize calculation angle */
   theta = start_angle_rad;
   /* calculate total arc angle */
   total_angle = end_angle_deg - start_angle_deg;
   /* calculate subinterval */
```

```
delta_theta_deg = total_angle/4.0;
delta_theta_rad = delta_theta_deg * DEGREE_TO_RAD;
final_delta_cnt = 4; /* iteration # */
/* calculate coordinates */
for(delta_count = 0; delta_count <= final_delta_cnt;
    delta_count++)
      if(delta_count != 2)
          X = (int)((radius - 23) * cos(theta) *
                     screen_scale);
          Y = (int)((radius - 23) * sin(theta));
          conv[index++1 = 0x0000 | (X + x_center);
          conv[index++] = 0x1000 | (Y + y_center);
          X = (int)((radius + 23) * cos(theta) *
                     screen_scale);
          Y = (int)((radius + 23) * sin(theta));
          conv[index++] = 0x0000 + (X + x_center);
          conv[index++] = 0x1800 | (Y + y_center);
        }
      theta = theta + delta_theta_rad;
   }
*length_ptr = index;
) /* end of clim_hsh */
```

/**************************

SOURCE FILE: clim_aro.c

FUNCTION: clim_arrow()

DESCRIPTION: This routine generates the HP code to

put up the climb arrow for the rate of climb indicator on the DATA PAGE. It makes the calculation for a given

location, radius, and angle.

DOCUMENTATION

FILES: None.

ARGUMENTS: x_center : x plot coordinate

for arrow

y_center : y plot coordinate for arrow

angle_deg : plot angle in

degrees

radius : length of arrow

conv : pointer to

conversion array

RETURN: None.

FUNCTIONS

CALLED: None.

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 02Feb87 Version 1.0

REVISIONS: None.

#include<math.h>

```
void clim_arrow(x_center, y_center, angle_deg, radius,
               length_ptr, conv)
unsigned short conv[]:
unsigned short *length_ptr;
int x_center, y_center;
int radius:
double angle_deg;
   int
          х, у;
                               /* plot coordinates */
   int
          arrow_head_end_X;
                              /* arrow head end coord */
   int
          arrow_head end Y:
   int
          arrow_head_bgn_X;
                              /* arrow head base coord */
   int
         arrow_head bgn Y;
   double PI:
   double angle_rad;
                              /* plot angle in radians */
   double screen scale;
   PI = 3.14159265;
   screen_scale = 9.5/12.5;
      convert angle to radians */
   angle_rad = angle_deg * (2 * PI) / 360.0;
   /* calculate coordinates */
   arrow_head_end_X = (int)(radius * cos(angle_rad)
                          * screen_scale);
   arrow_head_end_Y = (int)(radius * sin(angle_rad));
  conv[2] = 0x0000 | (arrow_head_end_X + x_center);
  conv[3] = 0x1800 | (arrow_head_end_Y + y_center);
  arrow_head_bgn_X = (int)((radius - 50) *
                           cos(angle_rad) *
                           screen_scale);
  arrow_head_bgn_Y = (int)((radius - 50) *
                           sin(angle_rad));
  y = (int)(20 * cos(angle_rad));
  x = (int)(-20 * sin(angle_rad) * screen_scale);
  conv[4] = 0x0000 | (x_center + arrow_head_bgn_X + x);
  conv[5] = 0x1800 | (y_center + arrow_head_bgn_Y + y);
  conv[6] = 0x0000 | (arrow_head_end_X + x_center);
```

```
conv[7] = 0x1000 ! (arrow_head_end_Y + y_center);
conv[8] = 0x0000 ! (x_center + arrow_head_bgn_X - x);
conv[9] = 0x1800 i (y_center + arrow_head_bgn_Y - y);
*length_ptr = 10;
} /* end of clim_arrow */
```

SOURCE FILE: climrate.c

FUNCTION: climb_rate()

DESCRIPTION: This routine retrieves the vertical

speed value from the data package and converts it into feet/minute. It then converts this value into a zero-

padded plot string.

DOCUMENTATION

¥

FILES: None.

ARGUMENTS: clock_pkg : pointer to clock

package

alarm_pkg : pointer to alarm

package

XO, YO : plot coordinates

vert_speed : pointer to climb rate

length_ptr : pointer to length of

conversion array

climb_rate_conv : pointer to

conversion array

RETURN: None.

FUNCTIONS

CALLED: zero_pad()

string_gen()

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 23Mar87 Version 1.0

REVISIONS: 11Apr87

Added call to climb_filter()

Dave Gruenbacher

```
**********************
#include <stdlib.h>
#include <string.h>
#include <stdio.h>
#include "data_str.h"
                       /* data package structure
                           declaration.
void climb_rate(clock_pkg, alarm_pkg, XO, YO, vert_speed,
                length_ptr,climb_rate_conv)
CLOCK_PKG *clock_pkg;
ALARM_PKG *alarm_pkg;
int XO, YO;
int *length ptr;
int *vert speed;
unsigned short climb_rate_conv[];
    int climb_rate;
    int length;
    int radix = 10:
   char final_string[20]:
   double size = 1.5;
    float climb_filter();
   void string gen():
    if(data_pkg. VERT SPEED <= 115)
       data_pkg.VERT_SPEED = data_pkg.VERT_SPEED + 16;
   data_pkg. VERT_SPEED = (unsigned char)
                       (climb_filter(data_pkg.VERT_SPEED));
   climb_rate = (int)((data_pkg.VERT_SPEED - 136.0)
                       * 20.3);
   itoa(climb_rate, final_string, radix);
   string_gen(final_string, XO, YO, size, &length,
              climb_rate_conv);
   *vert_speed = climb_rate;
   *length_ptr = length;
} /* end climb_rate */
```

/********************************** * SOURCE FILE: plane.c FUNCTION: plane() * DESCRIPTION: This function generates the code to put up a plane on the vector graphics display at a given location. DOCUMENTATION FILES: None. ARGUMENTS: XO, YO : plot coordinates for plane length_ptr : pointer to length of conv[] array CONV : pointer to conversion array RETURN: None. FUNCTIONS CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 15Jan87 Version 1.0 REVISIONS: None. ****************** void plane(X0, Y0, length_ptr, conv) unsigned short conv[]; unsigned short *length_ptr; unsigned short XO, YO;

/* coordinate variables */

{ int X, Y:

```
/* draw fusilage */
   X = XO;
   Y = Y0 - 140;
   conv[0] = 0x0000 | X;
   conv[1] = 0x1000 | Y;
   X = X - 20;
   Y = Y + 180;
   conv[2] = 0x0000 | X;
   conv[3] = 0x1800 | Y:
   X = X + 20:
   Y = Y + 20;
   conv[4] = 0x0000 | X;
   conv[5] = 0x1800 | Y:
   X = X + 20;
   Y = Y - 20;
   conv[6] = 0x0000 | X;
   conv[7] = 0x1800 | Y:
   X = X - 20;
   Y = Y - 180:
   conv[8] = 0x0000 | X;
   conv[9] = 0x1800 | Y:
/* draw wing */
   X = X0 - 100;
   Y = Y0 - 20;
   conv[10] = 0x0000 | X;
   conv[11] = 0x1000 | Y:
   Y = Y + 40;
   conv[12] = 0x0000 | X:
   conv[13] = 0x1800 | Y:
   X = X + 200:
   conv[14] = 0x00000 | X;
   conv[15] = 0x1800 | Y:
  Y = Y - 40:
   conv[16] = 0x00000 | X:
   conv[17] = 0x1800 | Y:
  X = X - 200;
  conv[18] = 0x0000 | X:
  conv[19] = 0x1800 | Y:
/* draw aeleron */
  X = X0 - 30:
```

```
Y = Y0 - 120;
  conv[20] = 0x00000 i X;
  conv[21] = 0x1000 ! Y;
  Y = Y + 20;
  conv[22] = 0x0000 | X;
  conv[23] = 0x1800 | Y;
  X = X + 60;
 conv[24] = 0x0000 | X;
  conv[25] = 0x1800 | Y;
  Y = Y - 20;
  conv[26] = 0x00000 | X;
  conv[27] = 0x1800 + Y;
 X = X - 60;
 conv[28] = 0x0000 | X;
  conv[29] = 0x1800 | Y;
  *length_ptr = 30;
}/* end plane */
```

/******************** SOURCE FILE: waypoint.c FUNCTION: waypoint() DESCRIPTION: This function generates the code to put up a waypoint on the vector graphics display at a given location. ¥ DOCUMENTATION FILES: None. ARGUMENTS: XO, YO : plot coordinates for plane ¥ length_ptr : pointer to length of conv[] array conv : pointer to conversion array ¥ RETURN: None. FUNCTIONS CALLED: None. ¥ AUTHOR: CHUCK ROBERTSON DATE CREATED: 25Jan87 Version 1.0 REVISIONS: None. ************* void waypoint(XO, YO, length_ptr, conv) unsigned short conv[]: unsigned short *length_ptr; unsigned short XO. YO: { unsigned short X, Y;

/* coordinate variables */

```
double screen scale = 9.5/12.5:
                                   /* scale screen */
X = X0 - (unsigned short)(13 * screen_scale);
Y = Y0 + 13;
conv[0] = 0x0000 | X;
conv[1] = 0x1000 | Y;
X = X + (unsigned short)(13 * screen_scale);
Y = Y + 37:
conv[2] = 0x0000 | X;
conv[3] = 0x1800 | Y;
X = X + (unsigned short)(13 * screen_scale);
Y = Y - 37;
conv[4] = 0x00000 | X:
conv[5] = 0x1800 | Y:
X = X + (unsigned short)(38 * screen_scale);
conv[6] = 0x0000 | X;
conv[7] = 0x1800 | Y;
X = X - (unsigned short)(27 * screen_scale);
Y = Y - 19:
conv[8] = 0x0000 | X;
conv[9] = 0x1800 | Y:
X = X + (unsigned short)(14 * screen_scale);
Y = Y - 40:
conv[10] = 0x0000 + X;
conv[11] = 0x1800 | Y;
X = X - (unsigned short)(38 * screen_scale);
Y = Y + 27;
conv[12] = 0x00000 | X:
conv[13] = 0x1800 | Y;
X = X - (unsigned short)(38 * screen_scale);
Y = Y - 27:
conv[14] = 0x0000 | X:
conv[15] = 0x1800 | Y:
X = X + (unsigned short)(14 * screen_scale);
Y = Y + 40:
conv[16] = 0x0000 | X:
conv[17] = 0x1800 | Y:
X = X - (unsigned short)(27 * screen_scale);
Y = Y + 19;
conv[18] = 0x0000 | X:
conv[19] = 0x1800 | Y:
```

```
X = X + (unsigned short)(38 * screen_scale);
conv[20] = 0x0000 | X;
conv[21] = 0x1800 | Y;

*length_ptr = 22;
} /* end waypoint */
```

/******************************** SOURCE FILE: vortac.c FUNCTION: vortac() DESCRIPTION: This function generates the code to put up a vortac on the vector graphics display at a given location. * DOCUMENTATION FILES: None. ARGUMENTS: XO, YO : plot coordinates for plane length_ptr : pointer to length of conv[] array conv : pointer to conversion array RETURN: None. FUNCTIONS CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 25Jan87 Version 1.0 REVISIONS: None. *********************** void vortac(XO, YO, length_ptr, conv) unsigned short conv[]; unsigned short *length_ptr; unsigned short XO, YO; { unsigned short X, Y; /* coordinates variables */

```
double screen_scale = 9.5/12.5; /* scale screen */
   X = X0 - (unsigned short)(25 * screen_scale);
   Y = YO + 40:
   conv[0] = 0x0000 | X:
   conv[1] = 0x1000 | Y;
   X = X + (unsigned short)(50 * screen_scale);
   conv[2] = 0x0000 | X;
   conv[3] = 0x1800 | Y:
   X = X + (unsigned short)(30 * screen_scale);
   Y = Y - 40:
   conv[4] = 0x0000 | X;
   conv[5] = 0x1800 | Y:
   X = X - (unsigned short)(30 * screen_scale);
   Y = Y - 40;
   conv[6] = 0x0000 | X:
   conv[7] = 0x1800 | Y:
   X = X - (unsigned short)(50 * screen_scale);
   conv(8) = 0x0000 | X;
   conv[9] = 0x1800 | Y:
   X = X - (unsigned short)(30 * screen_scale);
   Y = Y + 40:
   conv[10] = 0x00000 | X;
   conv[11] = 0x1800 | Y;
   X = X + (unsigned short)(30 * screen scale);
   Y = Y + 40:
   conv[12] = 0x00000 | X;
   conv[13] = 0x1800 | Y:
   Y = Y + 30:
   conv[14] = 0x00000 | X:
   conv[15] = 0x1800 | Y;
   X = X + (unsigned short)(50 * screen_scale);
   conv[16] = 0x00000 | X;
   conv[17] = 0x1800 + Y;
   Y = Y - 30;
   conv[18] = 0x00000 | X;
   conv[19] = 0x1800 | Y:
/* fill in */
  Y = Y + 6;
  conv[20] = 0x0000 | X;
```

```
conv[21] = 0x1000 | Y:
   X = X - (unsigned short)(50 * screen_scale);
   conv[22] = 0x0000 | X;
   conv[23] = 0x1800 | Y:
   Y = Y + 6:
   conv[24] = 0x00000 | X;
   conv[25] = 0x1000 | Y:
   X = X + (unsigned short)(50 * screen_scale);
   conv[26] = 0x00000 | X;
   conv[27] = 0x1800 | Y:
   Y = Y + 6:
   conv[28] = 0x0000 | X;
   conv[29] = 0x1000 | Y:
   X = X - (unsigned short)(50 * screen_scale);
   conv[30] = 0x0000 | X:
   conv[31] = 0x1800 | Y:
   Y = Y + 6:
   conv[32] = 0x0000 | X;
   conv[33] = 0x1000 | Y;
   X = X + (unsigned short)(50 * screen_scale);
   conv[34] = 0x00000 | X;
   conv[35] = 0x1800 | Y:
   Y = Y + 6;
   conv[36] = 0x0000 | X:
   conv[37] = 0x1000 | Y:
  X = X - (unsigned short)(50 * screen_scale);
   conv[38] = 0x0000 | X;
  conv[39] = 0x1800 | Y:
/* bottom right knob */
  X = X0 + (unsigned short)(25 * screen_scale);
  Y = Y0 - 40;
  conv[40] = 0x0000 | X:
  conv[41] = 0x1000 | Y:
  X = X + (unsigned short)(24 * screen_scale);
  Y = Y - 18;
  conv[42] = 0x0000 | X;
  conv[43] = 0x1800 | Y:
  X = X + (unsigned short)(30 * screen_scale);
```

```
Y = Y + 40;
  conv[44] = 0x00000 | X;
  conv[45] = 0x1800 | Y:
  X = X - (unsigned short)(24 * screen_scale);
  Y = Y + 18:
  conv[46] = 0x00000 | X;
  conv[47] = 0x1800 | Y:
/* fill in */
  X = X + (unsigned short)(4 * screen_scale);
  Y = Y - 3;
  conv[48] = 0x00000 | X;
  conv[49] = 0x1000 | Y:
  X = X - (unsigned short)(30 * screen_scale);
  Y = Y - 40;
  conv[50] = 0x00000 | X:
  conv[51] = 0x1800 | Y;
  X = X + (unsigned short)(4 * screen scale);
  Y = Y - 3:
  conv[52] = 0x0000 | X;
  conv[53] = 0x1000 | Y;
  X = X + (unsigned short)(30 * screen_scale);
  Y = Y + 40:
  conv[54] = 0x00000 | X;
  conv[55] = 0x1800 | Y;
  X = X + (unsigned short)(4 * screen_scale);
  Y = Y - 3:
  conv[56] = 0x0000 | X;
  conv[57] = 0x1000 | Y;
  X = X - (unsigned short)(30 * screen_scale);
  Y = Y - 40;
  conv[58] = 0x0000 | X;
  conv[59] = 0x1800 | Y:
  X = X + (unsigned short)(4 * screen_scale);
  Y = Y - 3:
  conv[60] = 0x0000 | X:
  conv[61] = 0x1000 | Y;
  X = X + (unsigned short)(30 * screen_scale);
  Y = Y + 40;
  conv[62] = 0x0000 | X;
  conv[63] = 0x1800 | Y:
```

```
X = X + (unsigned short)(4 * screen_scale);
   Y = Y - 3:
   conv[64] = 0x0000 | X;
   conv[65] = 0x1000 | Y;
   X = X - (unsigned short)(30 * screen_scale);
   Y = Y - 40:
   conv[66] = 0x0000 | X;
   conv[67] = 0x1800 | Y:
   X = X + (unsigned short)(4 * screen_scale);
   Y = Y - 3:
   conv[68] = 0x0000 | X;
   conv[69] = 0x1000 | Y;
   X = X + (unsigned short)(30 * screen_scale);
   Y = Y + 40:
   conv[70] = 0x0000 | X:
   conv[71] = 0x1800 | Y;
/* bottom left knob */
   X = X0 - (unsigned short)(55 * screen_scale);
   Y = YO:
   conv[72] = 0x0000 | X;
   conv(73) = 0x1000 | Y:
   X = X - (unsigned short)(24 * screen_scale);
   Y = Y - 18;
   conv[74] = 0x0000 | X;
   conv[75] = 0x1800 | Y:
   X = X + (unsigned short)(30 * screen_scale);
   Y = Y - 40:
   conv[76] = 0x00000 | X;
   conv[77] = 0x1800 | Y:
   X = X + (unsigned short)(24 * screen_scale);
   Y = Y + 18;
   conv[78] = 0x0000 | X:
   conv[79] = 0x1800 | Y:
/* fill in */
   X = X - (unsigned short)(4 * screen_scale);
   Y = Y - 3:
  conv[80] = 0x0000 | X:
  conv[81] = 0x1000 | Y:
  X = X - (unsigned short)(30 * screen_scale);
  Y = Y + 40:
  conv[82] = 0x0000 | X:
```

```
conv[83] = 0x1800 | Y;
X = X - (unsigned short)(4 * screen_scale);
Y = Y - 3;
conv[84] = 0x0000 | X;
conv[85] = 0x1000 | Y;
X = X + (unsigned short)(30 * screen_scale);
Y = Y - 40;
conv[86] = 0x0000 | X;
conv[87] = 0x1800 | Y;
X = X - (unsigned short)(4 * screen scale);
Y = Y - 3;
conv[88] = 0x0000 | X;
conv[89] = 0x1000 | Y:
X = X - (unsigned short)(30 * screen scale);
Y = Y + 40;
conv[90] = 0x0000 | X;
conv[91] = 0x1800 | Y;
X = X - (unsigned short)(4 * screen_scale);
Y = Y - 3;
conv[92] = 0x00000 | X;
conv[93] = 0x1000 | Y;
X = X + (unsigned short)(30 * screen scale);
Y = Y - 40;
conv[94] = 0x00000 | X:
conv[95] = 0x1800 | Y:
X = X - (unsigned short)(4 * screen_scale);
Y = Y - 3;
conv[96] = 0x0000 | X;
conv[97] = 0x1000 | Y;
X = X - (unsigned short)(30 * screen_scale);
Y = Y + 40;
conv[98] = 0x0000 | X;
conv[99] = 0x1800 | Y:
X = X - (unsigned short)(4 * screen_scale);
Y = Y - 3;
conv[100] = 0x0000 | X;
conv[101] = 0x1000 | Y:
X = X + (unsigned short)(30 * screen_scale);
Y = Y - 40:
conv[102] = 0x0000 | X:
```

```
conv[103] = 0x1800 | Y;
 *length_ptr = 104;
}/* end vortac */
```

```
/************************
   SOURCE FILE:
                   ndb.c
   FUNCTION:
                    ndb()
   DESCRIPTION:
                    This routine generates the HP code to
                    put up a ndb at a given location on
 ¥
                    the vector graphics display.
   DOCUMENTATION
   FILES:
                    None.
   ARGUMENTS:
                    XO. YO
                                   : plot coordinates
                    length_ptr
                                   : pointer to length of
                                    conv[] array
                    conv
                                   : pointer to
 ¥
                                    conversion array
   RETURN:
                    None.
   FUNCTIONS
   CALLED:
                    arc_circ()
 ×
   AUTHOR:
                   CHUCK ROBERTSON
   DATE CREATED:
                    30Jan87
                               Version 1.0
 ×
   REVISIONS:
                    None.
*************************
void ndb(X0, Y0, length_ptr, conv)
unsigned short conv[];
unsigned short *length ptr;
int XO, YO;
( unsigned short i;
                                   /* loop counter */
  unsigned short arc_circ_conv[100]; /* conversion
```

*/

```
unsigned short arc_circ_length;
                                       /* insertion
                                           length
                                                     */
  double start_ang, end_ang;
         radius1, radius2, radius3;
  int
  int
         total length:
                                       /* total insertion
                                          length
  void arc circ():
  total_length = 0;
  start_ang = 0.0;
  end ang = 360.0:
  radius1 = 25:
  radius2 = 50:
  radius3 = 75:
  arc_circ(XO, YO, start_ang, end_ang, radius1,
           &arc_circ_length, arc_circ_conv);
  for(i=0;i<arc_circ_length;i++)
     conv[total_length + i] = arc_circ_conv[i];
  total_length = total_length + arc_circ_length;
  arc_circ(XO, YO, start_ang, end_ang, radius2,
           &arc_circ_length, arc circ conv):
  for(i=0;i<arc_circ_length;i++)
     conv[total_length + i] = arc circ conv[i]:
  total_length = total_length + arc_circ_length;
  arc_circ(XO, YO, start_ang, end_ang, radius3,
           &arc_circ_length, arc circ conv):
 for(i=0;i<arc_circ_length;i++)
     conv[total_length + i] = arc_circ_conv[i];
 total_length = total_length + arc_circ_length;
  *length_ptr = total_length;
} /* end of ndb */
```

SOURCE FILE: compass.c

FUNCTION: compass()

DESCRIPTION: This routine generates the HP code to

put up the hash marks and appropriate markings for a compass showing 90 degrees at a given radius for a given

heading.

DOCUMENTATION

FILES: None.

ARGUMENTS: x_center : x plot coordinate

for compass center y_center : y plot coordinate

for compass center heading_deg : heading in degrees

radius : radius of compass length_ptr : pointer to length of

compass_conv : pointer to

conversion array

RETURN: None.

FUNCTIONS

CALLED: None.

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 04Feb87 Version 1.0

REVISIONS: None.

#include<math.h>
#include<string.h>

#include<stdlib. h>

```
void compass(x_center, y_center, heading_deg, radius,
             length_ptr, compass_conv)
unsigned short compass conv[]:
unsigned short *length_ptr;
int x_center, y_center;
int radius;
int heading deg;
    int
           X. Y:
                                    /* plot coordinates */
    int
           1:
                                    /* loop counter */
    int
           length:
                                    /* insertion length */
    int
           heading_upper_bound;
                                    /* heading bounds */
    int
           heading_lower_bound;
    int
           degrees_off center;
                                    /* degrees off center
                                       from nearest tick
                                       mark
    int
           nearest_tick;
                                    /* nearest tick to
                                       heading
                                                        */
    int
           upper_tick, lower_tick;
                                    /* max and min tick
                                       values
    int
           tick_value;
                                    /* value of tick mark */
           num_iterations;
    int
                                    /* num of loop
                                       iterations */
    int
           count;
                                    /* loop counter */
    int
           index:
                                    /* conversion array
                                       index
                                                         * /
    int
           disp;
                                    /* displacement for
                                       display
                                                         * /
   int
           str_length;
                                    /* string insertion
                                       length
   int
           tick_value_XO;
                                    /* tick value plot
                                       coord
   int
           tick_value YO;
   double PI:
   double delta_theta_deg;
                                    /* angle increment
                                       degrees
                                                        * /
   double delta theta rad:
                                    /* angle increment
                                       radians
   double theta_rad, theta_deg;
                                    /* calculation angle */
   double screen_scale;
   double DEGREE_TO_RAD;
   double start_angle rad;
                                    /* first calculation
                                       angle
   double start_angle_deg;
   double size:
                                    /* character size */
```

char

hdg_string[3];

```
unsigned short string conv[40];
void string gen():
PI = 3.14159265;
delta_theta_deg = 5.0;
DEGREE\_TO\_RAD = (2 * PI / 360.0);
screen scale = 9.5/12.5:
size = 1.0:
length = 0;
/* calculate upper and lower bounds of heading */
heading_upper_bound = (heading deg + 45) % 360:
heading_lower_bound = (heading_deg + 315) % 360;
/* calculate degrees off center of closest tick mark
   from heading
degrees_off_center = heading_deg % 5;
/* calculate what tick is nearest heading */
nearest_tick = heading deg - degrees off center:
/* calculate upper and lower ticks */
upper_tick = (nearest_tick + 45) % 360;
lower_tick = (nearest_tick + 315) % 360:
/* get lower_tick within range */
if(lower_tick < heading_lower_bound)
   lower_tick = (lower_tick + 5) % 360;
/* calculate start plot angle for lower_tick */
if(lower_tick < heading_lower bound)
   start_angle_deg = (double)(135.0 -
                     ((lower tick + 360) -
                     heading_lower_bound));
else
   start_angle_deg = (double)(135.0 -
                     (lower_tick -
                      heading_lower_bound));
/* convert angles to radians */
delta_theta_rad = delta_theta_deg * DEGREE_TO_RAD;
start_angle_rad = start_angle_deg * DEGREE_TO_RAD;
```

```
/* INITIALIZE CALCULATION ANGLE, TICK_VALUE, AND
   INDEX
theta_rad = start_angle_rad;
tick_value = lower_tick;
index = 0;
/* CALCULATE # OF ITERATIONS */
if((heading_upper_bound % 5) == 0)
   num iterations = 19:
else
   num_iterations = 18;
/* CALCULATE COORDINATES AND CHARACTERS FOR HP CODE */
for(count = 0; count < num_iterations; count++)</pre>
   (
      X = (int)(radius * cos(theta_rad) * screen_scale);
      Y = (int)(radius * sin(theta_rad));
      compass_conv[index++] = 0x0000 | (X + x_center);
     compass_conv[index++] = 0x1000 | (Y + y_center);
      if((tick_value % 10) == 0)
         disp = 100;
     else
         disp = 50:
     X = (int)((radius - disp) * cos(theta rad)
                 * screen_scale);
     Y = (int)((radius - disp) * sin(theta_rad));
     compass_conv[index++] = 0x0000 | (X + x_center);
     compass_conv[index++] = 0x1800 | (Y + y_center);
     theta_rad = theta_rad - delta_theta rad;
     if((tick_value % 90) == 0)
           switch (tick_value)
                 case 0:
                   strcpy(hdg string, "N"):
                   break:
                 case 90:
                   strcpy(hdg_string, "E");
                   break;
                 case 180:
                   strcpy(hdg_string, "S");
```

```
break:
                  case 270:
                    strcpy(hdg_string, "W");
                    break:
                  default:
                    break:
            str_length = 1;
          }
      else
            if((tick_value % 10) == 0)
                  itoa(tick_value, hdg_string, 10);
                  str_length = strlen(hdg_string);
      /* CALCULATE COORDINATES FOR TICK_VALUE */
      switch (str_length)
         {
            case 1:
                disp = 18:
                break:
            case 2:
                disp = 36;
                break:
            case 3:
                disp = 54;
                break;
         }
      tick_value_XO = X + x_center - disp;
      tick_value_Y0 = Y + y_center - 50;
      if((tick_value % 10) == 0)
           string_gen(hdg_string, tick_value_XO,
                       tick_value_YO, size, &length,
                       string_conv);
           for(i=0;i<length:i++)
              compass_conv[index + i] = string_conv[i];
           index = index + length;
        }
      tick_value = (tick_value + 5) % 360;
   }
*length_ptr = index;
```

}/* end of compass */

/*********************

SOURCE FILE: hdg_brg.c

FUNCTION: heading and bearing()

DESCRIPTION: This routine generates the HP code to

put up the heading and bearing of an ndb or vortac. It calculates the appropriate value and plots it with a corresponding arrow indicating a heading "to" or a bearing "from" an

ndb or vortac.

DOCUMENTATION

FILES: None.

ARGUMENTS: plane_XO

: X plot coord. plane_YO : Y plot coord.

plane_heading_deg : plane heading theta_deg

: plot angle in degrees radius : plot radius

length_ptr : pointer to length of conv[] array

hdg_brg_conv : pointer to conversion

array

RETURN: None.

FUNCTIONS

¥

CALLED: insert()

string_gen() line() arrow()

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 23Feb87 Version 1.0

```
REVISIONS:
                    None.
*************************************
#include<stdlib.h>
#include<string.h>
#include<math.h>
#include<stdio.h>
void heading_and_bearing(plane_XO, plane_YO,
                        plane_heading_deg, theta_deg,
                        radius, length_ptr, hdg_brg_conv)
int plane_XO, plane_YO;
int plane_heading_deg;
double theta_deg;
double radius;
unsigned short hdg_brg_conv[];
unsigned short *length_ptr;
{
   unsigned short conv[50]; /* conversion array */
   int length;
                              /* insertion length */
   int i:
                              /* loop counter */
   int index;
                              /* array index */
   int radix;
                              /* conversion radix */
   int
          insert();
   int string gen();
   int
         line():
         arrow();
   int
   char final_string[20]:
                             /* conversion strings */
   char cat_string[5]:
   double size;
                              /* character size */
   double theta_rad;
                              /* conversion angle in
                                 radians
   double screen_scale;
                              /* screen scale */
   double PI;
   double DEGREE_TO_RAD:
   int
          heading deg;
                             /* heading in degrees */
   int
          hdg_X, hdg_Y;
                             /* heading plot coord */
                              /* relative to plane
   int
          hdg_label_X;
                              /* heading plot coord */
   int
          hdg_label Y;
   double hdg_label_theta;
                              /* final heading plot
                                 angle
```

```
double hdg_label_phi;
                             /* intermediate hdg plot
                                angle
 double hdg_label_radius;
                             /* hdg plot radius */
 double hdg_delta_X;
                            /* hdg intermed x plot
                                value
 double hdg_delta_Y;
                            /* hdg intermed y plot
                                value
 double hdg_delta_X_sqrd;
 double hdg_delta_Y_sqrd;
        bearing_deg;
 int
                            /* bearing in degrees */
 int
        brg_X, brg_Y;
                            /* bearing plot coord */
                            /* relative to plane */
 int
        brg_label_X;
                            /* bearing plot coord */
 int
        brg_label_Y;
 double brg_label_theta;
                            /* final bearing plot
                               angle
double brg_label_phi;
                            /* intermediate brg plot
                               angle
double brg_label_radius;
                            /* brg plot radius */
double brg_delta_X;
                            /* brg intermed x plot
                               value
double brg_delta_Y;
                            /* brg intermed y plot
                               value
                                                    */
double brg_delta_X_sqrd;
double brg_delta_Y_sqrd;
screen_scale = 9.5/12.5;
PI = 3.14159265:
DEGREE_{TO}RAD = (2 * PI / 360.0);
index = 0:
size = 1.0;
radix = 10:
/* convert plot angle to radians */
theta_rad = theta_deg * DEGREE_TO_RAD;
/* calculate heading to ndb or vortac */
heading_deg = (int)(plane_heading_deg + 360
                           + (90 - theta_deg) );
if(heading_deg >= 360)
   heading_deg = heading_deg % 360;
/* calculate bearing from a ndb or vortac */
```

```
bearing_deg = (heading_deg - 180) + 360;
if(bearing_deg >= 360)
   bearing_deg = bearing_deg % 360;
/* assign delta_y constants */
hdg_delta_Y = 50.0:
brg_delta Y = 50.0:
/* calculate delta x values */
hdg_delta_X = 2.0 / 3.0 * radius;
brg_delta_X = 1.0 / 3.0 * radius;
/* calculate heading and bearing phi angle */
hdg_label_phi = atan2(hdg_delta Y.
                              hdg delta X):
brg_label_phi = atan2(brg_delta_Y,
                              brg_delta X);
/* calculate radii to plot heading and bearing */
hdg_delta_X_sqrd = pow(hdg_delta_X, 2.0);
hdg_delta_Y_sqrd = pow(hdg_delta_Y, 2.0);
hdg_label_radius = sqrt(hdg_delta_X_sqrd
                               + hdg_delta_Y_sqrd);
brg_delta_X_sqrd = pow(brg_delta_X, 2.0);
brg_delta_Y_sqrd = pow(brg_delta_Y, 2.0);
brg_label_radius = sqrt(brg_delta_X_sqrd
                              + brg_delta_Y_sqrd);
/* figure out which side of radius to plot heading
   and bearing
if((theta_deg <= 0.0) || (theta_deg >= 180.0))
   (hdg_label_theta = theta_rad +
                             hdg_label_phi;
   brg_label_theta = theta rad +
                             brg_label_phi:
else
   (hdg_label_theta = theta_rad -
                             hdg_label_phi;
```

```
brg_label_theta = theta_rad -
                               brg_label_phi;
     }
 /* calculate heading plot coordinates */
 hdg_label_X =(int)(hdg_label_radius *
                            cos(hdg_label_theta) *
                            screen_scale);
 hdg_label_Y =(int)(hdg_label_radius *
                            sin(hdg_label_theta));
 hdg_X = hdg_label_X + plane_X0;
 hdg_Y = hdg_label_Y + plane_YO;
 /* plot heading */
 itoa(heading_deg, final_string, radix);
 string_gen(final_string, hdg_X, hdg_Y, size, &length, conv);
for(i=0;i<length;i++)
     hdg_brg_conv[index + i] = conv[i];
index = index + length;
arrow(plane_XO, plane_YO, theta_deg, hdg_delta_X, "to",
       &length, conv);
for(i=0;i<length;i++)
    hdg_brg_conv[index + i] = conv[i];
index = index + length;
/* calculate bearing plot coordinates */
brg_label_X =(int)(brg_label_radius *
                           cos(brg_label_theta) *
                           screen_scale);
brg_label_Y =(int)(brg_label_radius *
                           sin(brg_label_theta));
brg_X = brg_label_X + plane_X0;
brg_Y = brg_label_Y + plane_YO;
/* plot bearing */
itoa(bearing_deg, final_string, radix);
string_gen(final_string,brg_X,brg_Y,size,&length,conv);
```

```
for(i=0;i<length;i++)
    hdg_brg_conv[index + i] = conv[i];

index = index + length;

arrow(plane_XO, plane_YO, theta_deg, brg_delta_X, "from",
    &length, conv);

for(i=0;i<length;i++)
    hdg_brg_conv[index + i] = conv[i];

index = index + length;
    *length_ptr = index;
} /* end of heading_and_bearing */</pre>
```

```
SOURCE FILE: ndb_angl.c
   FUNCTION:
                   get_ndb_plot_angle()
   DESCRIPTION:
                   This routine gets the raw adf from the
                   data package and uses a lookup table
                   along with interpolation to convert
                   the raw adf into the ndb plot angle.
   DOCUMENTATION
   FILES:
                   None.
   ARGUMENTS:
                   ndb_theta_deg_ptr : pointer to the
                                     plot angle
                                     variable in
                                     calling routine
   RETURN:
                   None.
   FUNCTIONS
   CALLED:
                   None.
   AUTHOR:
                  CHUCK ROBERTSON
   DATE CREATED:
                   23Mar87
                             Version 1.0
  REVISIONS:
                  None.
*************************
#include "data str.h"
#include<stdlib. h>
typedef struct ( double a_to_d;
              double value; } CONVERSION;
/* lookup table for raw adf value conversion */
CONVERSION adf[18] = {
```

```
{0.0, 184.0},
                              {3.0, 180.0},
                              {19.0, 170.0},
                              {25.0, 150.0}.
                              {47.0, 120.0},
                              {68.0,90.0),
                              {89.0,60.0},
                              {110.0,30.0},
                              {131.0.0.0}.
                              {154.0,330.0},
                              {176.0,300.0}.
                              {199.0,270.0},
                              {221.0, 240.0},
                             {243.0,210.0},
                             {250.0, 200.0}.
                              {254.0,190.0},
                              (255.0, 185.0)
                       };
void get_ndb_plot_angle(ndb_theta_deg_ptr)
double *ndb_theta_deg_ptr;
{
                          /* loop counter */
    double raw_adf;
    /* GET RAW ADF READING FROM A/D AND FIND PLOT ANGLE OF
       NDB IN DEGREES USING LOOKUP TABLE
    raw_adf = (double)(data_pkg.ADF);
    if( (raw_adf > 131.0) && (raw_adf <= 154.0))
       adf[8].value = 360.0:
       adf[8].value = 0.0;
   for(i = 0; i < 17; i++)
       if( (raw_adf >= adf[i].a_to_d) &&
           (raw_adf <= adf[i + 1].a_to_d) )
          {
             /* INTERPOLATE */
             *ndb_theta_deg_ptr
                   = (double)( (raw_adf - adf[i].a_to_d) /
                              (adf[i+1].a_to_d -
                               adf[i].a_to_d) *
                              (adf[i+1].value -
                               adf[i].value) +
                               adf[i].value):
```

```
break;
)

/* end of get_ndb_plot_angle */
```

SOURCE FILE: runway.c FUNCTION: runway() DESCRIPTION: This function calculates the HP code to generate the runway for a given location, angle, and scale. It is drawn using vanishing point techniques. DOCUMENTATION FILES: None. ARGUMENTS: rho_deg : centerline angle x_center, y_center : plot coordinates scale : runway scale length ptr : pointer to length of conv[] array conv : pointer to conversion array RETURN: None. * FUNCTIONS CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 18Feb87 Version 1.0 REVISIONS: None.

#include<stdlib.h>
#include<math.h>

```
int x_center, y_center;
unsigned short runway_conv[];
unsigned short *length ptr:
double rho_deg;
double scale;
{
   int
           index:
                                   /* conversion array
                                      index
   int
           delta_x, delta_y;
                                   /* delta values for
                                      plot
                                                        * /
   int
           base_middle_X;
                                   /* coord of middle of */
   int
          base middle Y:
                                   /* base of runway
   int
        end_runway_radius;
                                   /* radius to runway
                                      end
   int
          end_middle_X;
                                   /* coord of middle of */
   int
         end_middle_Y;
                                  /* end of runway
                                                          */
   int
          base_left_X;
                                  /* corner coordinates */
   int
          base_left_Y;
                                  /* of runway
   int
          base right X:
   int
         base_right_Y;
   int
         end_left_X;
   int
         end_left_Y;
   int
         end_right_X;
   int
         end_right_Y;
   int
          base_centerline_radius; /* radii to base and */
          end_centerline_radius; /* end of runway
   int
                                                        */
                                   /* centerline
                                                        * /
   int
          base_centerline_X;
                                   /* centerline */
   int
          base_centerline Y:
                                  /* coordinates */
   int
          end_centerline_X;
   int
          end_centerline_Y;
   int
          touchdown_line_left_X; /* touchdown line */
   int
          touchdown_line_left_Y;
                                  /* coordinates
                                                     */
   int
          touchdown_line_right_X;
   int
          touchdown_line_right_Y;
   double rho_rad;
                               /* crab angle in radians
   double phi_deg;
                               /* plot angle in degrees
                                                          */
   double phi_rad;
                                /* plot angle in radians
   double screen_scale;
   double PI:
   double DEGREE_TO_RAD;
```

```
index = 0;
end_runway_radius = (int)(200 * scale);
base_centerline_radius = (int)(30 * scale);
end_centerline_radius = (int)(190 * scale);
screen_scale = 9.5/12.5;
PI = 3.14159265;
DEGREE_TO_RAD = (2 * PI / 360.0);
/* CONVERT RHO TO RADIANS */
rho_rad = rho_deg * DEGREE TO RAD:
/* CALCULATE MIDDLE OF BASE */
delta_x = (int)((-50 * scale) * tan(rho_rad)
                 * screen_scale);
delta_y = (int)(-50 * scale);
base_middle_X = x_center + delta_x;
base_middle_Y = y_center + delta y;
/* CALCULATE PHI */
phi_deg = 90.0 - rho_deg;
phi_rad = phi_deg * DEGREE TO RAD:
/* CALCULATE MIDDLE END OF RUNWAY */
delta_x = (int)(end_runway_radius * cos(phi rad)
                * screen_scale);
delta_y = (int)(end_runway_radius * sin(phi_rad));
end_middle_X = base_middle_X + delta_x;
end_middle_Y = base_middle_Y + delta_y;
/* CALCULATE FOUR CORNERS OF RUNWAY */
base_left_X = base_middle_X -
              (int)(50 * scale * screen scale);
base_left_Y = base_middle_Y;
base_right_X = base_middle_X +
               (int)(50 * scale * screen_scale);
base_right_Y = base_middle_Y;
end_left_X = end_middle_X -
             (int)(30 * scale * screen_scale);
end_left_Y = end_middle_Y;
```

```
end_right_X = end_middle_X +
              (int)(30 * scale * screen_scale);
end_right_Y = end_middle Y;
/* CALCULATE RUNWAY CENTERLINE POINTS */
delta_x = (int)(base_centerline_radius *
                cos(phi_rad) * screen_scale);
delta_y = (int)(base_centerline_radius * sin(phi_rad));
base_centerline_X = base_middle_X + delta_x;
base_centerline_Y = base_middle_Y + delta_y;
delta_x = (int)(end_centerline_radius *
                cos(phi_rad) * screen_scale);
delta_y = (int)(end_centerline_radius * sin(phi_rad));
end_centerline_X = base_middle_X + delta x;
end_centerline_Y = base_middle_Y + delta_y;
/* CALCULATE TOUCHDOWN LINE POINTS */
touchdown_line_left_X = x_center -
                        (int)(20 * scale)
                              * screen_scale);
touchdown_line_left_Y = y_center;
touchdown_line_right_X = x_center +
                         (int)(20 * scale
                               * screen_scale);
touchdown_line_right_Y = y_center;
/* CONVERT TO HP CODE */
/* plot runway outline */
runway_conv[index++] = 0x0000 | base left X:
runway_conv[index++] = 0x1000 | base_left_Y;
runway_conv[index++] = 0x0000 | end left X:
runway_conv[index++] = 0x1800 | end_left_Y;
runway_conv[index++] = 0x0000 | end_right X;
runway_conv[index++] = 0x1800 | end_right_Y;
runway_conv[index++] = 0x0000 | base_right_X;
runway_conv[index++] = 0x1800 | base_right_Y;
runway_conv[index++] = 0x0000 | base_left_X;
runway_conv[index++] = 0x1800 | base left Y;
```

```
/* plot centerline */
runway_conv[index++] = 0x0000 | base_centerline_X;
runway_conv[index++] = 0x1000 | base_centerline_Y;

runway_conv[index++] = 0x0000 | end_centerline_X;
runway_conv[index++] = 0x1800 | end_centerline_Y;

/* plot touchdown line */
runway_conv[index++] = 0x0000 | touchdown_line_left_X;
runway_conv[index++] = 0x1000 | touchdown_line_left_Y;

runway_conv[index++] = 0x0000 | touchdown_line_right_X;
runway_conv[index++] = 0x1800 | touchdown_line_right_Y;

*length_ptr = index;
} /* end of runway */
```

/*******************

SOURCE FILE: 11s_cmps.c

FUNCTION: 11s_compass()

DESCRIPTION: This routine generates the HP code to

put up the hash marks and appropriate markings for a compass showing 20 degrees at a given radius for a given

heading.

DOCUMENTATION

FILES: None.

ARGUMENTS: x_center : x plot coordinate

for compass center y_center : y plot coordinate

for compass center heading_deg : heading in degrees

radius : radius of compass
length_ptr : pointer to length of
conv[] array

compass_conv : pointer to

conversion array

RETURN: None.

FUNCTIONS
CALLED: None.

DATE CREATED: 30Mar87 Version 1.0

CHUCK ROBERTSON

REVISIONS: None.

#include<math.h>

#include<string.h>
#include<stdlib.h>

AUTHOR:

```
void ils_compass(x_center, y_center, heading_deg, radius,
                 length_ptr,compass_conv)
unsigned short compass_conv[];
unsigned short *length_ptr;
int x_center, y_center;
int radius;
int heading_deg;
    int
           X, Y:
                                    /* plot coordinates */
    int
           i;
                                    /* loop counter */
    int
           length;
                                    /* insertion length */
           heading_upper_bound;
    int
                                    /* heading bounds */
    int
           heading_lower_bound;
    int
         tick_value;
                                    /* value of tick mark */
           num_iterations;
    int
                                    /* num of loop
                                       iterations */
    int
           count:
                                    /* loop counter */
    int
           index;
                                    /* conversion array
                                       index
    int
                                    /* displacement for
           disp;
                                       display
                                                         * /
    int
           str_length;
                                   /* string insertion
                                       length
    int
           tick_value_XO;
                                    /* tick value plot
                                       coord
    int
           tick_value_YO;
    double PI:
    double delta_theta_deg;
                                  /* angle increment
                                       degrees
                                                        * /
   double delta_theta_rad;
                                    /* angle increment
                                       radians
    double theta_rad, theta_deg;
                                    /* calculation angle */
    double screen_scale;
   double DEGREE_TO_RAD;
   double start angle rad;
                                   /* first calculation
                                       angle
   double start_angle_deg;
   double size;
                                    /* character size */
   char
           hdg string[3];
   unsigned short string_conv[40];
   void string_gen();
```

PI = 3.14159265; delta_theta_deg = 1.0;

```
DEGREE_{TO}RAD = (2 * PI / 360.0):
screen_scale = 9.5/12.5;
size = 1.0:
length = 0:
/* calculate upper and lower bounds of heading */
heading_upper_bound = (heading_deg + 10) % 360;
heading_lower_bound = (heading_deg + 350) % 360;
start_angle_deg = 100.0:
/* convert angles to radians */
delta_theta_rad = delta_theta_deg * DEGREE_TO_RAD;
start_angle_rad = start_angle_deg * DEGREE_TO_RAD;
/* INITIALIZE CALCULATION ANGLE, TICK_VALUE, INDEX.
   AND # OF ITERATIONS */
theta_rad = start_angle_rad;
tick_value = heading_lower_bound;
index = 0:
num_iterations = 21;
/* CALCULATE COORDINATES AND CHARACTERS FOR HP CODE */
for(count = 0; count < num_iterations; count++)
     X = (int)(radius * cos(theta_rad) * screen_scale);
     Y = (int)(radius * sin(theta_rad));
     compass_conv[index++] = 0x0000 | (X + x_center);
     compass_conv[index++] = 0x1000 | (Y + y_center);
     if((tick_value % 5) == 0)
        disp = 100:
     else
        disp = 50:
     X = (int)((radius - disp) * cos(theta_rad)
                 * screen_scale):
     Y = (int)((radius - disp) * sin(theta_rad));
     compass_conv[index++] = 0x0000 | (X + x_center);
     compass_conv[index++] = 0x1800 | (Y + y_center);
     theta_rad = theta_rad - delta_theta_rad;
     if((tick_value % 90) == 0)
```

```
switch (tick value)
         {
            case 0:
              strcpy(hdg_string, "N");
              break:
            case 90:
              strcpy(hdg_string, "E");
              break;
           case 180:
              strcpy(hdg_string, "S");
              break:
           case 270:
              strcpy(hdg_string, "W");
              break;
           default:
              break:
      str_length = 1;
else if((tick_value % 10) == 0)
           itoa(tick_value, hdg_string, 10);
           str_length = strlen(hdg_string);
/* CALCULATE COORDINATES FOR TICK VALUE */
switch (str_length)
      case 1:
         disp = 18:
         break;
      case 2:
         disp = 36;
         break;
      case 3:
         disp = 54;
         break;
   }
tick_value_X0 = X + x_center - disp;
tick_value_Y0 = Y + y_center - 50;
if((tick_value % 10) == 0)
  (
     string_gen(hdg_string, tick value XO.
                tick_value YO, size, &length.
                string_conv);
```

APPENDIX D (CONT.)

HOST PROGRAM INCLUDE FILES

SOURCE FILE: data_str.h FUNCTION: None. DESCRIPTION: This is a file to be included in any function that needs to access a member of the data package. All the pieces of the data package are stored in the structure data pkg. Individual members are accessed by using the following name: data_pkg.ALTITUDE etc. DOCUMENTATION FILES: None. ARGUMENTS: None. RETURN: None. FUNCTIONS CALLED: None. AUTHOR: Dave Gruenbacher & Chuck Robertson DATE CREATED: 22Jan87 Version 1.0 REVISIONS: 13Apr87 Version 1.1 Added conditional statements. Dave Gruenabcher

typedef struct {
unsigned char BANK ;
unsigned char PITCH ;
unsigned char VERT_SPEED ;
unsigned char DELTA_X ;

```
unsigned char
                DELTA Y
                                                   ;
unsigned char
                MANIFOLD_PRESSURE :
                COURSE_DEVIATION
unsigned char
                                           ;
unsigned char
                GLIDESLOPE
unsigned char
                ALTITUDE
                                                   ;
unsigned char
                AIRSPEED
                                                   ;
unsigned char
                COMPASS
                                                   ;
unsigned char
                ADF
                                                            ÷
unsigned char
                DME
                                                            ;
unsigned char
                POWER
                                                   ;
unsigned char
                RPM
                                                            ;
unsigned char
                SPARE
                BINARY_INPUTS
unsigned char
                                           ;
unsigned char
                LAST_KEY
                                                   ŝ
unsigned char
                MONTH
                                                   ĵ
unsigned char
                DAY
                                                            ;
unsigned char
                DATE
                                                   ÷
unsigned char
                HOURS
                                   ÷
unsigned char
                MINUTES
                                                   ;
unsigned char
                SECONDS
                                                   ;
                                   } DATA_PKG;
typedef struct {
                  int
                         timer min;
                  int
                         timer_sec;
                  int
                         time_out min:
                  int
                         time_out_sec;
                  int
                         math_operation_flag;
                  int
                         timer_operation_flag;
                  int
                         timer_status_flag;
                  double adf_freq;
                  double comi_freq;
                  double com2_freq;
                  double vor1_freq;
                  double vor2_freq;
                  int
                         assigned_altitude;
                  int
                         mda dh:
                  int
                         estimated_wind;
                                     } CLOCK_PKG:
typedef struct {
                  int
                         airspeed_alarm flag:
                 int
                         assigned_altitude_alarm_flag;
                 int
                         mda_dh_alarm_flag;
                 int
                         assigned_altitude_enable_flag;
                 int
                         mda_dh_enable_flag;
                 int
                         time_out_alarm_flag;
                 int
                         alarm_status flag:
                                               } ALARM PKG;
```

```
#ifdef ehsi_main
DATA_PKG data_pkg = {0};
unsigned short SCREEN[1000] = {0xFFFF};
#else
extern DATA_PKG data_pkg;
extern unsigned short SCREEN[];
extern int GET_DATA_PACKAGE();
extern int SEND_SCREEN();
extern int RETRIEVE_SCREEN();
extern int TOGGLE_ALARM_SWITCH();
#endif
#define NULL_TIMER
#define START_TIMER
#define RESET_TIMER
#define SET_TIMER
#define TIMER_OFF
                        1
                        2
                         3
                         0
#define TIMER_ON
                         1
#define ALARM_OFF
                         0
#define ALARM_ON
                         1
#define DISABLED
                         0
#define ENABLED
```

SOURCE FILE: datpg_xy.c

FUNCTION: None.

DESCRIPTION: This file is to be included with the

any file dealing with the DATA PAGE that needs to display something on the vector graphics display. It contains coordinates for plotting the various

items on the DATA PAGE.

DOCUMENTATION

FILES: None.

ARGUMENTS: None.

RETURN: None.

FUNCTIONS

CALLED: None.

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 01Feb87 Version 1.0

REVISIONS: None.

#define col2 1100

#define heading_XO col1

#define heading_YO 1980

#define heading_X1 col1 + (10 * 54)

#define heading_Y1 1980

#define airspeed_CAS_XO col1
#define airspeed_CAS_YO 1890

```
#define airspeed_CAS_X1 col1 + (10 * 54)
#define airspeed CAS Y1 1890
#define airspeed_TAS_XO col1
#define airspeed_TAS_YO 1800
#define airspeed_TAS_X1
#define airspeed_TAS Y1
#define gndspeed_XO col1
#define gndspeed_YO 1710
#define gndspeed_X1
#define gndspeed_Y1
#define assigned_XO col1
#define assigned_YO 1530
#define assigned_X1 col1 + (10 * 54)
#define assigned Y1 1530
#define altitude_XO col1
#define altitude_YO 1440
#define altitude_X1 col1 + (10 * 54)
#define altitude Y1 1440
#define mda_dh_XO col1
#define mda_dh_YO 1350
#define mda_dh_X1 col1 + (8 * 54)
#define mda_dh_Y1 1350
#define timer_XO col1
#define timer_YO 1170
#define timer_X1 col1 + (7 * 54)
#define timer Y1 1170
#define time_out_XO col1
#define time_out_YO 1080
#define time_out_X1 col1 + (9 * 54)
#define time out Y1 1080
#define com1_XO col1
#define com1_YO 900
#define com1 X1 col1 + (6 * 54)
#define com1_Y1 900
#define com2_XO col1
#define com2_YO 810
#define com2_X1 col1 + (6 * 54)
#define com2_Y1 810
#define nav1_X0 col1
#define nav1 YO 720
```

```
#define nav1 X1 col1 + (6 * 54)
#define nav1 Y1 720
#define nav2_X0 col1
#define nav2_Y0 630
#define nav2_X1 col1 + (6 * 54)
#define nav2_Y1 630
#define rnav_wpl_XO col1
#define rnav_wp1_YO 450
#define rnav_wp1_X1
#define rnav_wp1_Y1
#define rnav_wp2_X0 col1
#define rnav_wp2_Y0 360
#define rnav_wp2_X1
#define rnav_wp2_Y1
#define adf_XO col2
#define adf YO 450
#define adf_X1 co12 + (5 * 54)
#define adf_Y1 450
#define temp_XO col1
#define temp_YO 180
#define temp X1
#define temp_Y1
#define time_XO co12
#define time_YO 900
#define time_edt_XO time XO
#define time_edt_YO time_YO - 90
#define time_edt_X1 time_X0 + (10 * 52)
#define time_edt_Y1 time_Y0 - 90
#define time_zulu_XO time_XO
#define time_zulu_YO time_YO - 180
#define time_zulu_X1 time_X0 + (10 * 54)
#define time_zulu_Y1 time_Y0 - 180
#define time_since_lo_XO time_XO
#define time_since_lo_YO time_YO - 270
#define time_since_lo_X1 time_X0 + (10 * 54)
#define time_since_lo_Y1 time_Y0 - 270
#define barometer_XO col2
#define barometer_YO 180
#define barometer X1
#define barometer Y1
```

```
#define since_lo_XO col1
#define since_lo_YO 630
#define since_lo_X1
#define since lo Y1
/* coordinates for "CLIMB" */
#define climb arc XO col2 + 100
#define climb_arc_YO 1440 + 27
#define vert_speed_XO climb_arc_XO
#define vert_speed_YO climb arc YO + 270 - 27
#define C_XO climb_arc_XO - 100
#define C_YO climb_arc_YO + 180 - 27
#define L_XO climb_arc_XO - 100
#define L_YO climb_arc_YO + 90 - 27
#define I_XO climb_arc_XO - 100
#define I_YO climb_arc_YO - 27
#define M_XO climb_arc_XO - 100
#define M_YO climb_arc_YO - 90 - 27
#define B_XO climb_arc_XO - 100
#define B_YO climb_arc_YO - 180 - 27
#define climb_centerline1_X0 climb_arc_X0 - 47
#define climb_centerline1_YO climb_arc_YO
#define climb_centerlinel_X1 climb_arc_X0 + 47
#define climb_centerline1_Y1 climb arc Y0
#define climb_centerline2_XO climb_arc_XO - 47 + 190
#define climb_centerline2_YO climb arc_YO
#define climb_centerline2_X1 climb_arc_XO + 47 + 190
#define climb_centerline2_Y1 climb_arc_Y0
#define climb_centerline3_XO climb_arc_XO - 47 + 2 * 190
#define climb_centerline3_YO climb_arc_YO
#define climb_centerline3_X1 climb_arc_X0 + 47 + 2 * 190
#define climb_centerline3_Y1 climb_arc_Y0
#define climb_centerline4_X0 climb_arc_X0 - 47 + 3 * 190
#define climb_centerline4_YO climb_arc YO
#define climb_centerline4_X1 climb_arc_X0 + 47 + 3 * 190
#define climb_centerline4_Y1 climb_arc_Y0
#define climb_centerline5_XO climb_arc_XO - 47 + 4 * 190
#define climb_centerline5_YO climb_arc_YO
#define climb_centerline5_X1 climb_arc_X0 + 47 + 4 * 190
```

#define climb_centerline5_Yl climb_arc_YO
/* end of coordinates for "CLIMB" */
#define aidspeed_bar_XO 900
#define aidspeed_bar_YO 1800

#define binary_ip_XO col2
#define binary_ip_YO 300

#define command_line_XO 10
#define command_line_YO 160
#define command_line_X1 2038
#define command_line_Y1 160

#define command_XO 10
#define command_XO 10
#define command_YO 10

/*********************************

SOURCE FILE: navpg_xy.c

FUNCTION: None.

DESCRIPTION: This file is to be included with the

any file dealing with the NAV PAGE that needs to display something on the vector graphics display. It contains coordinates for plotting the various items on the NAV PAGE.

DOCUMENTATION

FILES: None.

ARGUMENTS: None.

RETURN: None.

FUNCTIONS

CALLED: None.

AUTHOR: CHUCK ROBERTSON

DATE CREATED: 05Apr87 Version 1.0

REVISIONS: None.

#define airspeed_XO 10 #define airspeed YO 1950

#define airspeed_X1 10 + (7 * 54)

#define airspeed_Y1 1950

#define gndspeed_XO 10 #define gndspeed_YO 1850

#define gndspeed_X1

#define gndspeed Y1

```
#define track XO 1600
 #define track_YO 1950
 #define track X1
 #define track Y1
 #define altitude_XO 1600
 #define altitude_YO 1850
 #define altitude_X1 1600 + (4 * 54)
 #define altitude_Y1 1850
 #define dme_XO 10
 #define dme_YO 1750
 #define dme_X1 10 + (5 * 54)
 #define dme_Y1 1750
#define plane_XO 1048
#define plane YO 855
#define compass_arc_XO
                                1048
#define compass_arc_YO
                               10
#define compass_arc_init_angle 45.0
#define compass_arc_end_angle 135.0
#define compass_arc_radius
                                1860
/* HEADING, HEADING BOX, AND POINTER */
#define heading_XO 967
#define heading_YO 1980
#define heading_box_XO heading_XO
#define heading_box_YO heading_YO
#define heading_box_no_char 3
#define heading_ptr_line1_XO heading_XO - 10
#define heading_ptr_line1_YO heading_YO - 10
#define heading_ptr_line1_X1 compass_arc_X0
#define heading_ptr_line1_Y1 compass_arc_Y0 + \
                             compass_arc_radius
#define heading_ptr_line2_XO (heading_XO +
                              (heading_box_no_char * 54) + \
#define heading_ptr_line2_YO heading_YO - 10
#define heading_ptr_line2_X1 compass_arc XO
#define heading_ptr_line2_Y1 compass_arc_Y0 + \
                             compass_arc_radius
#define command_line_XO 10
#define command_line_YO 160
#define command_line_X1 2038
```

#define command_line_Y1 160

#define command_X0 10
#define command_Y0 10

SOURCE FILE: 11spg_xy.c FUNCTION: None. DESCRIPTION: This file is to be included with the any file dealing with the ILS PAGE that needs to display something on the vector graphics display. It contains coordinates for plotting the various items on the ILS PAGE. DOCUMENTATION FILES: None. ARGUMENTS: None. RETURN: None. FUNCTIONS CALLED: None. AUTHOR: CHUCK ROBERTSON DATE CREATED: 05Apr87 Version 1.0 REVISIONS: None. #define airspeed X1 0 #define airspeed Y1 0 #define altitude_XO 10 #define altitude YO 1950
#define altitude X1 10 + (5 * 54) #define altitude_Y1 1950

#define mda_dh_XO 10
#define mda_dh_YO 1850

```
#define mda_dh_X1 10 + (4 \times 54)
#define mda_dh_Y1 1850
#define dme_XO 1600
#define dme_YO 1950
#define dme_X1 1600 + (5 * 54)
#define dme_Y1 1950
#define compass arc XO
                                1048
#define compass_arc_YO
                                10
#define compass_arc_init_angle 80.0
#define compass_arc_end_angle
                                100.0
#define compass_arc_radius
/* HEADING, HEADING BOX, AND POINTER */
#define heading XO 967
#define heading YO 1980
#define heading_box_XO heading_XO
#define heading_box_YO heading_YO
#define heading_box_no char 3
#define heading_ptr_line1_XO heading_XO - 10
#define heading_ptr_line1_YO heading YO - 10
#define heading_ptr_line1_X1 compass arc XO
#define heading_ptr_line1_Y1 compass_arc_Y0 + \
                             compass arc radius
#define heading_ptr_line2_XO heading XO + \
                             (heading_box_no_char * 54) + 10
#define heading_ptr_line2_YO heading_YO - 10
#define heading_ptr_line2_X1 compass_arc_X0
#define heading_ptr_line2_Y1 compass_arc_Y0 + \
                             compass_arc_radius
#define command_line_XO 10
#define command_line_YO 160
#define command_line_X1 2038
#define command_line_Y1 160
#define command_XO 10
#define command YO 10
```

GLOSSARY

- ADF Automatic Direction Finder. Used for finding the direction of a NDB which is transmitting the set ADF frequency. Usually shown as an angle relative to the planes heading.
- Bearing The horizontal direction from a point measured clockwise from some reference point, usually magnetic north, through 360 degrees.
- DACI Data Acquisition and Communications Interface. The interface unit, which ties the EHSI system together and communicates with the host computer.
- DH Decision Height. Height at which a decision must be made, during an ILS approach, to either continue the approach or to execute a missed approach.
- DME Distance Measuring Equipment. Equipment used to measure, in nautical miles, the slant range distance of an aircraft from the DME navigation aid.
- Glideslope Descent profile determined for vertical guidance during an approach.
- Groundspeed The speed of the airplane relative to the surface of the earth.
- Heading The horizontal direction to a point measured clockwise from some reference point, usually magnetic north, through 360 degrees.
- IFR Instrument Flight Rule. Rules which apply when flying an aircraft under instrument conditions.
- ILS Instrument Landing System. System used when landing an aircraft under IFR conditions. Contains a glideslope, localizer, and markers.
- Localizer The component of an ILS, which provides course guidance to the runway.
- MDA Minimum Descent Altitude. Lowest altitude to which descent is authorized on final approach in execution of a standard instrument approach procedure where no electronic glideslope is provided.

GLOSSARY

- NDB Non-directional Beacon. A radio beacon transmitting non-directional signals. Direction to or from found using ADF equipment.
- Track The actual flight path of an aircraft over the surface of the earth.
- VORTAC VHR Omnidirectional Range Tactical Air Navigation. A navigational aid providing VOR azimuth, TACAN azimuth, and TACAN DME equipment at one location.
- Waypoint A predetermined geographical position used for route progress that is defined relative to a VORTAC.

GRAPHICAL PAGE DEVELOPMENT OF AN ELECTRONIC HORIZONTAL SITUATION INDICATOR

ьу

CHARLES A. ROBERTSON

B.S., Kansas State University, 1985

AN ABSTRACT OF A MASTER'S THESIS
submitted in partial fulfillment of the
requirements for the degree

MASTER OF SCIENCE

Department of Electrical and Computer Engineering

KANSAS STATE UNIVERSITY

Manhattan, Kansas

1987

ABSTRACT

A brief introduction to an Electronic Horizontal Situation Indicator (EHSI) is given. Development of the graphical pages of the EHSI is discussed. An overview of the development system used is given. The controlling routine of the EHSI is discussed along with graphical routines used to generate the various pages of the EHSI. Suggestions are made for improving the EHSI Development System. Future recommendations for further development of the EHSI pages are given.